



U.S. Department
of Transportation
**Federal Aviation
Administration**

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

EXTEX Engineered Products Inc.
340 E Germann Road, Suite 113
Gilbert Arizona 85297 USA

PMA NO: PQ2203NM
SUPPLEMENT NO.: 1
DATE: May 9, 2016
REISSUED: September 8, 2016

ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Nut, Coupling- Power Turbine Outer	E23001801	23001801	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23001801 <u>Rev:</u> N/C <u>Date:</u> 11/09/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Key, First Stage Nozzle Assembly	E23001816	23001816	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23001816 <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Nozzle Assembly Pressure Oil, Power Turbine Support	E23001832	23001832	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23001832 <u>Rev:</u> N/C <u>Date:</u> 11/29/1995	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1
Spacer, Corrugated, First Section	E23003108	23003108	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23003108 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2

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Chamber, Primary Combustion, Second Section	E23003109	23003109	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23003109 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Spacer, Corrugated, Second Section	E23003110	23003110	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23003110 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Spacer Detail, Combustor	E23003111	23003111	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23003111 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Diaphragm, First Stage Nozzle	E23004208	23004208	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23004208 <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Shim, Tabbed	E23004212	23004212	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23004212 <u>Rev:</u> N/C <u>Date:</u> 02/20/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M

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Ferrule Subassembly, Swirler & Injector	E23004528	23004528	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23004528 <u>Rev:</u> N/C <u>Date:</u> 07/22/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S
Dome, Combustion Sub Assembly	E23004529	23004529	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23004529 <u>Rev:</u> A <u>Date:</u> 08/15/1994, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Nut, Hex- Nylon Cap .3750-24 (350° F)	E23004538	23004538	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23004538 <u>Rev:</u> N/C <u>Date:</u> 06/09/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Shield, Water, Combustion Liner	E23004560	23004560	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23004560 <u>Rev:</u> N/C <u>Date:</u> 07/22/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S
Chamber, Primary Combustion, First Section	E23004562	23004562	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23004562 <u>Rev:</u> N/C <u>Date:</u> 07/22/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S

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Ring, First Stage Nozzle Assembly	E23005218	23005218	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23005218 <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Race, Bearing	E23005267	23005267	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23005267 <u>Rev:</u> N/C <u>Date:</u> 10/23/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Adapter, Spline Compressor	EH23005271-1	23005271-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23005271 <u>Rev:</u> N/C <u>Date:</u> 03/04/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Adapter, Spline Compressor	EH23005271-2	23005271-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23005271 <u>Rev:</u> N/C <u>Date:</u> 03/04/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Adapter, Spline Compressor	EH23005271-3	23005271-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23005271 <u>Rev:</u> N/C <u>Date:</u> 10/16/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Ring, Piston, Split Seal	E23005274	23005274	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23005274 <u>Rev:</u> N/C <u>Date:</u> 07/15/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M

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Ball Bearing 10 x 26 x 8	E23005749	23005749	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23005749 <u>Rev:</u> A <u>Date:</u> 02/10/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4
Dilution Chamber Assembly, Combustion Liner	E23007007	23007007	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007007 <u>Rev:</u> N/C <u>Date:</u> 07/22/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S
Ball Bearing 25 x 42 x 9	E23007151	23007151	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007151 <u>Rev:</u> A <u>Date:</u> 02/10/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30R/1
Pin, Stepped	E23007190	23007190	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007190 <u>Rev:</u> N/C <u>Date:</u> 12/11/1996, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R

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Roller Bearing ID x 42 x 9	E23007202	23007202	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007202 <u>Rev:</u> N/C <u>Date:</u> 09/03/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30R/1
Outer Ring, First Stage Nozzle Assembly	E23007220	23007220	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007220 <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Cone, First Section, Combustion Liner	E23007530	23007530	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007530 <u>Rev:</u> A <u>Date:</u> 08/08/1994, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Ferrule, Igniter	E23007531	23007531	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007531 <u>Rev:</u> N/C <u>Date:</u> 07/22/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S

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Washer	E23008017	23008017	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008017 <u>Rev:</u> N/C <u>Date:</u> 07/06/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Shaft- Support, Idler Gear	E23008018	23008018	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008018 <u>Rev:</u> B <u>Date:</u> 09/16/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Bolt, Turbine Tie	E23008030	23008030	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008030 <u>Rev:</u> N/C <u>Date:</u> 11/02/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Nozzle Assembly, First Stage	E23008064	23008064	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008064 <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2

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Nozzle Assembly, First Stage	E23008064-738	23008064	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008064-738 <u>Rev:</u> N/C <u>Date:</u> 01/29/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2
Ball Bearing 17 x 30 x 7	E23008065	23008065	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008065 <u>Rev:</u> A <u>Date:</u> 02/10/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4
Gearshaft, Spur Adapter	E23008089	23008089	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008089 <u>Rev:</u> N/C <u>Date:</u> 10/11/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Gearshaft Assembly, Spur Adapter	E23008090	23008090	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008090 <u>Rev:</u> N/C <u>Date:</u> 11/16/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Ball Bearing 12 x 28 x 8	E23008469	23008469	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008469 <u>Rev:</u> A <u>Date:</u> 02/10/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30R/1

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Blanket, Insulation	E23008555	23008555	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008555 <u>Rev:</u> N/C <u>Date:</u> 12/20/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Combustion Liner Assembly, C28	E23008614	23008614	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23008614 <u>Rev:</u> N/C <u>Date:</u> 12/14/1998 , or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Retaining Ring, Internal	E23009342	23009342	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23009342 <u>Rev:</u> N/C <u>Date:</u> 11/01/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Plug, Positioning	E23009573	23009573	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23009573 <u>Rev:</u> A <u>Date:</u> 09/01/1995, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C30, C30M, C30P, C30S, C30G, C30G/2
Plug, Positioning - Turbine Nozzle Shield	E23009574	23009574	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23009574 <u>Rev:</u> N/C <u>Date:</u> 11/17/1994, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Splined Adapter, Turbine	E23030974	23030974	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23030974 <u>Rev:</u> N/C <u>Date:</u> 10/19/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2

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Seal, Serrated, Balance Piston	E23030978-1,-2	23030978-1,-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23030978-1,-2 <u>Rev:</u> N/C <u>Date:</u> 12/22/1995, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, C20, C20B, C20C, C20F, C20J, C20S, C30, C30M, C30P, C30S, C30G, C30G/2
Nut, Coupling- Power Turbine Inner (LH)	E23030981	23030981	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23030981 <u>Rev:</u> A <u>Date:</u> 11/04/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Nut, Coupling PT Inner	E23030982	23030982	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23030982 <u>Rev:</u> N/C <u>Date:</u> 01/17/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Shaft, Gear, Oil Pump, 0.435 X 2.330 In.	E23030988	23030988	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23030988 <u>Rev:</u> N/C <u>Date:</u> 11/06/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Plate, First Stage Nozzle Assembly	E23031291	23031291	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031291 <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2

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Dome, Shield, Turbine Nozzle First Stage	E23031415	23031415	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031415 <u>Rev:</u> N/C <u>Date:</u> 11/08/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2
Support, Nozzle, Turbine, First Stage	E23031419	23031419	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031419 <u>Rev:</u> N/C <u>Date:</u> 12/23/1992, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W
Seal, Lab, GP Turbine Aft	E23031494	23031494	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031494 <u>Rev:</u> N/C <u>Date:</u> 01/13/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Vane Segment Assembly, Turbine, Second Stage	E23031741	23031741	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031741 <u>Rev:</u> N/C <u>Date:</u> 05/18/1998, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Gearshaft, Spur Adapter	E23031921	23031921	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031921 <u>Rev:</u> A <u>Date:</u> 05/16/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4

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Gearshaft Assembly	E23031922	23031922	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031922 <u>Rev:</u> N/C <u>Date:</u> 05/16/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Nozzle Assembly, Turbine Second Stage	E23031938	23031938	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031938 <u>Rev:</u> D <u>Date:</u> 12/24/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Nozzle Assembly, Turbine, Second Stage	E23031938-UC	23031938	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23031938-UC <u>Rev:</u> N/C <u>Date:</u> 03/15/1999, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Ring, Energy Absorbing	E23032263	23032263	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032263 <u>Rev:</u> N/C <u>Date:</u> 11/16/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Cage, Pinion Bearing	E23032283	23032283	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032283 <u>Rev:</u> C <u>Date:</u> 02/14/1994, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Firewall Assembly, Vertical	E23032326	23032326	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032326 <u>Rev:</u> N/C <u>Date:</u> 06/11/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M

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Seal, Abradable No. 5, Labyrinth	E23032327	23032327	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032327 <u>Rev:</u> N/C <u>Date:</u> 01/23/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Seal, Rotating Labyrinth, No. 5	E23032328	23032328	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032328 <u>Rev:</u> N/C <u>Date:</u> 12/17/1996, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Nozzle Assembly, Second Stage	E23032335	23032335	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032335 <u>Rev:</u> N/C <u>Date:</u> 10/13/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Diaphragm, Second Stage Nozzle Assembly	E23032335-2	23031934	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032335-2 <u>Rev:</u> N/C <u>Date:</u> 02/18/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2
Ring, Second Stage Nozzle Assembly	E23032335-4	23050841	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032335-4 <u>Rev:</u> N/C <u>Date:</u> 02/18/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2
Nozzle Assembly, Turbine, Second Stage	E23032335-UC	23032335	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032335-UC <u>Rev:</u> N/C <u>Date:</u> 03/15/1999, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2

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Coupling, Turbine To Compressor	E23032345	23032345	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23032345 <u>Rev:</u> N/C <u>Date:</u> 10/20/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Cover, Assembly Oil Sump - PT Support	E23034608	23034608	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23034608 <u>Rev:</u> N/C <u>Date:</u> 11/12/1996, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R
Gearshaft Assembly	E23034784	23034784	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23034784 <u>Rev:</u> N/C <u>Date:</u> 10/30/1992, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Race	E23034786	23034786	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23034786 <u>Rev:</u> A <u>Date:</u> 09/15/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4
Fitting Assembly, Oil Pressure, GP Turbine Support	E23035275	23035275	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23035275 <u>Rev:</u> N/C <u>Date:</u> 12/06/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1
Shaft, Power Turbine Inner, LH Thread	E23037403	23037403	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23037403 <u>Rev:</u> D <u>Date:</u> 06/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4

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Shaft, PT Outer	E23037413	23037413	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23037413 <u>Rev:</u> C <u>Date:</u> 08/24/1993, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1
Seal, Labyrinth, PT Rotating	E23037446	23037446	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23037446 <u>Rev:</u> N/C <u>Date:</u> 12/10/1996, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Shaft, Power Turbine Outer	E23038136	23038136	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23038136 <u>Rev:</u> N/C <u>Date:</u> 05/03/1996, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Shaft, Power Turbine Inner	E23038137	23038137	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23038137 <u>Rev:</u> N/C <u>Date:</u> 12/20/1996, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1
Nozzle Assembly, Turbine, First Stage, C20R	E23038219-247	23038219	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23038219-247 <u>Rev:</u> N/C <u>Date:</u> 01/29/1999, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4
Nozzle Assembly, Turbine, First Stage, C20R	E23038219-738	23038219	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23038219-738 <u>Rev:</u> N/C <u>Date:</u> 08/04/1997, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4

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Nozzle Assembly, Turbine, First Stage, C20R	E23038219- 738-P	23038219	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23038219-738-P <u>Rev:</u> N/C <u>Date:</u> 03/15/1999, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4
Ring, Piston, Split Seal	E23039060	23039060	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23039060 <u>Rev:</u> N/C <u>Date:</u> 06/25/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Diaphragm, Second Stage Nozzle Assembly	E23050841	23050841	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23050841 <u>Rev:</u> N/C <u>Date:</u> 10/13/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Cover, Diffuser Manifold	E23051099	23051099	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051099 <u>Rev:</u> N/C <u>Date:</u> 01/22/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Diffuser Assembly, Compressor	E23051119	23051119	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051119 <u>Rev:</u> N/C <u>Date:</u> 01/28/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2
Swirler	E23051617	23051617	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051617 <u>Rev:</u> N/C <u>Date:</u> 07/22/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30R/1, C30G/2

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Diaphragm Assembly, Nozzle Assembly, Turbine, First Stage, Keyed	E23051626	23051626	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051626 <u>Rev:</u> B <u>Date:</u> 12/23/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Ring, Diaphragm - Nozzle Assembly, Turbine, First Stage, Keyed	E23051629	23051629	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051629 <u>Rev:</u> N/C <u>Date:</u> 12/23/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Key, Nozzle - Assembly, Turbine, First Stage, Keyed	E23051630	23051630	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051630 <u>Rev:</u> N/C <u>Date:</u> 12/23/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Diaphragm, Nozzle Assembly, Turbine, First Stage, Keyed	E23051631	23051631	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051631 <u>Rev:</u> A <u>Date:</u> 12/23/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Gasket, Lube Oil Pump	E23053259	23053259	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23053259 <u>Rev:</u> N/C <u>Date:</u> 06/25/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4

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Gasket, Turbine Thermo- couple	E23053263	6874718	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23053263 <u>Rev:</u> N/C <u>Date:</u> 08/04/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Gasket, 6 Mounting Hole, 2.52 I.D. X 0.020 in.	E23053995	23053995	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23053995 <u>Rev:</u> N/C <u>Date:</u> 06/25/1998, or later FAA approved revisions	Rolls-Royce	250-B17C, C18A, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C47M
Gasket, Filter Housing	E23053996	6840113	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23053996 <u>Rev:</u> N/C <u>Date:</u> 06/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M

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Gasket, Lube Oil Pump	E23053997	6840139	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23053997 <u>Rev:</u> N/C <u>Date:</u> 06/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Gasket, 3 Bolt Flange	E23054600	6859183	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23054600 <u>Rev:</u> N/C <u>Date:</u> 07/06/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Gasket, Bleed Control	E23054604	6852734	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23054604 <u>Rev:</u> N/C <u>Date:</u> 07/06/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Tube Assembly, PC Air Scroll	E23054628	23054628	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23054628 <u>Rev:</u> B <u>Date:</u> 03/27/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20W
Key, Second Stage Nozzle Assembly	E23055441	23055441	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23055441 <u>Rev:</u> N/C <u>Date:</u> 10/13/1995, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2, C40B, C47B, C47M
Liner Assembly, Combustor	E23056108	23056108	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23056108 <u>Rev:</u> A <u>Date:</u> 08/08/1994, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4

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Gearshaft, Spur-Gasifier Train Idler	E23056128	23056128	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23056128 <u>Rev:</u> N/C <u>Date:</u> 08/15/1994, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Wheel, Compressor First Stage	E23057111	23057111	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23057111 <u>Rev:</u> A <u>Date:</u> 11/01/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Wheel, Compressor Second & Third Stage	E23057112	23057112	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23057112 <u>Rev:</u> A <u>Date:</u> 11/01/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Wheel, Compressor Fourth Stage	E23057114	23057114	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23057114 <u>Rev:</u> A <u>Date:</u> 11/01/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Wheel, Compressor 5th Stage	E23057115	23057115	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23057115 <u>Rev:</u> A <u>Date:</u> 11/01/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Wheel, Compressor 6th Stage	E23057116	23057116	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23057116 <u>Rev:</u> A <u>Date:</u> 11/01/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W

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Tube Assembly, Governor PY to Fuel Control	E23057137	23057137	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23057137 <u>Rev:</u> N/C <u>Date:</u> 05/21/1997, or later FAA approved revisions	Rolls-Royce	250-C20B, C20C, C20F, C20J, C20S, C20W, C20R/2, C20R/4
Gasket, 3 Bolt Flange	E23061822	23061822	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23061822 <u>Rev:</u> N/C <u>Date:</u> 06/25/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Shield Assembly, Turbine Nozzle, First Stage	E23062750	23062750	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062750 <u>Rev:</u> N/C <u>Date:</u> 08/11/1997, or later FAA approved revisions	Rolls-Royce	250-B15G, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Shield Assembly, Turbine Nozzle - First Stage	E23062752	23062752	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062752 <u>Rev:</u> N/C <u>Date:</u> 02/18/1998, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Nozzle Assembly, Turbine, First Stage	E23062754	23062754	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062754 <u>Rev:</u> N/C <u>Date:</u> 02/25/1998, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4

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Dome	E23065575	23065575	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23065575 <u>Rev:</u> N/C <u>Date:</u> 02/18/1998, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Deflector	E23065580	23065580	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23065580 <u>Rev:</u> N/C <u>Date:</u> 02/18/1998, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Shaft, Power Turbine Inner	E23071309	23071309	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23071309 <u>Rev:</u> N/C <u>Date:</u> 12/10/1999, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Shaft, Power Turbine Inner	E23071313	23071313	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23071313 <u>Rev:</u> N/C <u>Date:</u> 12/10/1999, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1
Vane Ring, Second Stage Nozzle Assembly	E2335VR	23032335	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E2335VR <u>Rev:</u> N/C <u>Date:</u> 10/13/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2

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Ring, Retaining	E6726656	6726656	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6726656 <u>Rev:</u> N/C <u>Date:</u> 02/27/1997, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17F, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Seal –O Ring	E6784419	6784419	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6784419 <u>Rev:</u> N/C <u>Date:</u> 12/13/91, or later FAA approved revisions	Rolls-Royce	250-B51G-17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, G30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Nut-Spanner .84375-32 THD	E6808392	6808392	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6808392 <u>Rev:</u> N/C <u>Date:</u> 06/09/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4

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Spacer	E6808397	6808397	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6808397 <u>Rev:</u> A <u>Date:</u> 11/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C
Ring, Piston	E6810436	6810436	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6810436 <u>Rev:</u> N/C <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4
Plug, Oil Seal	E6810498	6810498	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6810498 <u>Rev:</u> N/C <u>Date:</u> 11/13/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30R/3, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M
Gasket, 6 Mounting Hole	E6820085	6820085	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6820085 <u>Rev:</u> N/C <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-C18A, C18C, C20, C20B, C20C, C20F, C20J, C20R/1, C20R/2, C30, C30P

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Shaft - Support, Idler Gear	E6820586	6820586	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6820586 <u>Rev:</u> N/C <u>Date:</u> 06/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C28B, C28C, C30, C30M, C30P, C30S, C30G
Locknut	E6823169	6823169	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6823169 <u>Rev:</u> A <u>Date:</u> 10/20/1993, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Bearing, Ball	E6824728	6824728	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6824728 <u>Rev:</u> N/C <u>Date:</u> 08/26/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S
Nut, Spanner- Spl Lock Flange	E6829641	6829641	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6829641 <u>Rev:</u> N/C <u>Date:</u> 05/27/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4

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Gasket, Filter Housing	E6840113	6840113	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6840113 <u>Rev:</u> N/C <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R/1, C20R/2, C30, C30M, C30P, C30S
Gasket, Lube Oil Pump	E6840139	6840139	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6840139 <u>Rev:</u> N/C <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R/1, C20R/2
Nut, Spanner- Turbine Tie, 5/16-32	E6840406	6840406	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6840406 <u>Rev:</u> D <u>Date:</u> 02/21/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M

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Nut, Spanner- 0.750-20 THD	E6840407	6840407	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6840407 <u>Rev:</u> N/C <u>Date:</u> 11/08/1995, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Gasket, Metallic Ring	E6841676	6841676	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6841676 <u>Rev:</u> A <u>Date:</u> 12/12/1991, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Washer, Spring Tension Wave 2.63 ID X 3.23 OD X .010 Thick, CRES	E6843609	6843609	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6843609 <u>Rev:</u> N/C <u>Date:</u> 11/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W
Packing, Preformed	E6843610	6843610	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6843610 <u>Rev:</u> B <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J

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Shaft, Oil Pump	E6844007	6844007	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6844007 <u>Rev:</u> N/C <u>Date:</u> 11/13/1992, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Damper Segment	E6844422	6844422	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6844422 <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Washer, Flat 2.160 ID X 2.430 OD	E6848996	6848996	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6848996 <u>Rev:</u> N/C <u>Date:</u> 11/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M

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Diaphragm, Nozzle, Turbine, Second Stage	E6851499	6851499	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6851499 <u>Rev:</u> N/C <u>Date:</u> 12/28/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Packing Preformed "O" Ring	E6851501	6851501	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6851501 <u>Rev:</u> B <u>Date:</u> 12/06/1991, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Seal, Labyrinth, Turbine, Fourth Stage	E6851530	6851530	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6851530 <u>Rev:</u> N/C <u>Date:</u> 01/05/1995, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Nozzle Assembly, Turbine, Fourth Stage	E6851531	6851531	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6851531 <u>Rev:</u> N/C <u>Date:</u> 01/05/1995, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W

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Shaft, Oil Pump	E6853446	6853446	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6853446 <u>Rev:</u> N/C <u>Date:</u> 11/13/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30M, C30P, C30S, C30G
Gearshaft Assembly, Oil Pump & Gasifier Tach	E6853448	6853448	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6853448 <u>Rev:</u> N/C <u>Date:</u> 11/13/1992, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4
Seal, Plain, Encased	E6854086	6854086	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6854086 <u>Rev:</u> A <u>Date:</u> 09/10/1992, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M

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Seal, Plain, Encased	E6854424	6854424	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6854424 <u>Rev:</u> A <u>Date:</u> 09/10/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Seal, Plain, Encased	E6854521	6854521	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6854521 <u>Rev:</u> A <u>Date:</u> 09/10/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Seal, Plain, Encased	E6856211	6856211	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6856211 <u>Rev:</u> A <u>Date:</u> 09/10/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4

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Ring, Retaining - Internal 1.765 OD X .042 Thick.	E6859073	6859073	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6859073 <u>Rev:</u> N/C <u>Date:</u> 11/24/1997, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1 B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Gasket, Rubber, 3 Bolt Flange	E6859183	6859183	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6859183 <u>Rev:</u> N/C <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R/1, C20R/2
Bearing, Ball, Annular 25 x 52 x 15	E6859433	6859433	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6859433 <u>Rev:</u> N/C <u>Date:</u> 09/03/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4

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Race, Bearing- Inner, 25mm ID X 11mm Wide	E6870031	6870031	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6870031 <u>Rev:</u> N/C <u>Date:</u> 09/15/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Tube Assembly, Air - Power Turbine Governor to Control	E6870035	6870035	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6870035 <u>Rev:</u> N/C <u>Date:</u> 05/19/1997, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1
Ring, Piston, Split, 2.230 Diameter	E6870736	6870736	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6870736 <u>Rev:</u> A <u>Date:</u> 08/01/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W

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Ring, Piston, Split, 2.970 Diameter	E6870737	6870737	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6870737 <u>Rev:</u> A <u>Date:</u> 08/01/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Coupling, Turbine Shaft To Pinion Gear	E6870832	6870832	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6870832 <u>Rev:</u> E <u>Date:</u> 10/30/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Shaft, Gear, Oil Pump, 0.435 X 1.380 In.	E6870927	6870927	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6870927 <u>Rev:</u> N/C <u>Date:</u> 11/06/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20S, C20W, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M

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Shim, Tabbed	E6871162	6871162	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6871162 <u>Rev:</u> N/C <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Bolt, Tie, Compressor Rotor	E6871259	6871259	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6871259 <u>Rev:</u> E <u>Date:</u> 06/09/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Tube Assembly, Air - Fuel Control to Regulator Pressure	E6871461	6871461	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6871461 <u>Rev:</u> N/C <u>Date:</u> 05/19/1997, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20W, C20R, C20R/1

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Seal, Mating Ring-Front Compressor	E6871902	6871902	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6871902 <u>Rev:</u> B <u>Date:</u> 07/14/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2, C30R/1, C03R/3, C40B, C47B, C47M
Valve, Compressor Bleed	E6874816	6874816	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6874816 <u>Rev:</u> N/C <u>Date:</u> 10/19/1994, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R/1
Roller, Bearing 25 x 62 x 17	E6875035	6875035	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6875035 <u>Rev:</u> N/C <u>Date:</u> 09/03/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4

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Gasket, Metallic - U Ring	E6875350	6875350	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6875350 <u>Rev:</u> A <u>Date:</u> 12/16/1991, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Gasket, Metallic - U Ring	E6875351	6875351	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6875351 <u>Rev:</u> A <u>Date:</u> 12/16/1991, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C
Seal, Mating Ring- Rotating	E6875491	6875491	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6875491 <u>Rev:</u> A <u>Date:</u> 08/18/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Liner Assembly, Partial Combustor, Aft Section	E6875628	6875628	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6875628 <u>Rev:</u> A <u>Date:</u> 08/15/1995, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4

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Ring, Piston Split Seal	E6876016	6876016	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6876016 <u>Rev:</u> A <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J
Tube Assembly, Air – Bleed Valve	E6876294	6876294	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6876294 <u>Rev:</u> N/C <u>Date:</u> 05/23/1997, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J
Tube Assembly, PC Filter to PT Governor Tee	E6876542	6876542	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6876542 <u>Rev:</u> A <u>Date:</u> 03/27/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Ring, Piston- Seal	E6876637	6876637	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6876637 <u>Rev:</u> N/C <u>Date:</u> 01/02/1992, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4
Stud	E6876741	6876741	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6876741 <u>Rev:</u> N/C <u>Date:</u> 01/09/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C22B, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C40B, C47B, C47M

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Nut, Spanner 1.125-32 Thread Spl Lock (LH)	E6876878	6876878	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6876878 <u>Rev:</u> A <u>Date:</u> 08/31/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W
Tube Assembly, Fuel Control to Accumulator Pressure	E6877277	6877277	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6877277 <u>Rev:</u> N/C <u>Date:</u> 05/19/1997, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20R, C20R/1
Seal, Labyrinth Stator, GPT	E6877736	6877736	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6877736 <u>Rev:</u> B <u>Date:</u> 08/10/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Nozzle Assembly, Turbine, Second Stage	E6878426	6878426	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6878426 <u>Rev:</u> N/C <u>Date:</u> 08/04/1997, or later FAA approved revisions	Rolls-Royce	250-B17, C18, C18A, C18B, C18C, C20
Hose Assembly, TFE - Coded Configuration	E6886430	6886430	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6886430 <u>Rev:</u> N/C <u>Date:</u> 05/27/1997, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20W, C20R, C20R/1, C28B

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Seal, Plain Encased- 0.625 DIA X 0.531 Wide	E6887619	6887619	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6887619 <u>Rev:</u> A <u>Date:</u> 08/17/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Thermo- couple Harness Assembly	E6887761	6887761	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6887761 <u>Rev:</u> N/C <u>Date:</u> 06/03/1993, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28, C28B, C28C
Roller Bearing	E6887772	6887772	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6887772 <u>Rev:</u> N/C <u>Date:</u> 09/03/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4
Gearshaft, Starter	E6888529	6888529	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6888529 <u>Rev:</u> A <u>Date:</u> 05/04/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W

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Cover Assembly, Oil Sump - Power Turbine Support	E6888547	6888547	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6888547 <u>Rev:</u> C <u>Date:</u> 10/20/1993, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Cover Assembly, Oil Sump - Power Turbine Support	E6888547-UC	6888547	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6888547-UC <u>Rev:</u> C <u>Date:</u> 03/15/1999, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Coupling, Turbine Shaft to Pinion Gear	E6889069	6889069	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6889069 <u>Rev:</u> N/C <u>Date:</u> 06/07/1996, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Ball Bearing 30 x 55 x 13/16	E6889093	6889093	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6889093 <u>Rev:</u> N/C <u>Date:</u> 08/26/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/4
Nut, Spanner, 1.5625-18	E6889318	6889318	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6889318 <u>Rev:</u> N/C <u>Date:</u> 01/17/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1
Ring, Piston - Seal	E6889347	6889347	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6889347 <u>Rev:</u> N/C <u>Date:</u> 02/13/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M

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Gear, Helical - Power Train Drive	E6889700	6889700	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6889700 <u>Rev:</u> N/C <u>Date:</u> 05/04/1994, or later FAA approved revisions	Rolls-Royce	250-B17, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Diaphragm, Fourth Stage Nozzle Assembly	E6890034	6890034	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6890034 <u>Rev:</u> N/C <u>Date:</u> 01/05/1995, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Ring, Fourth Stage Nozzle Assembly	E6890035	6890035	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6890035 <u>Rev:</u> N/C <u>Date:</u> 01/05/1995, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Shield Assembly, Turbine Nozzle - First Stage	E6890040	6890040	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6890040 <u>Rev:</u> A <u>Date:</u> 06/29/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20R, C20R/1, C20R/2, C20R/4
Ring, Retaining, Internal	E6892073	6892073	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6892073 <u>Rev:</u> N/C <u>Date:</u> 02/11/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30R/1, C30R/3, C47B, C47M
Ring, Nozzle, Turbine, Second Stage	E6892823	6892823	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6892823 <u>Rev:</u> N/C <u>Date:</u> 12/28/1992, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4

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Shield Assembly, Turbine Nozzle, First Stage	E6892828	6892828	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6892828 <u>Rev:</u> N/C <u>Date:</u> 11/09/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Ring, Retaining	E6892838	6892838	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6892838 <u>Rev:</u> N/C <u>Date:</u> 02/27/1997, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30R/1, C30R/3, C40B, C47B, C47M
Seal, Labyrinth, Fourth Stage Nozzle Assembly	E6893606	6893606	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893606 <u>Rev:</u> N/C <u>Date:</u> 01/05/1995, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Seal, Labyrinth, Fourth Stage Nozzle Assembly	E6893607	6893607	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893607 <u>Rev:</u> N/C <u>Date:</u> 01/05/1995, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Nut, Spanner	E6893614	6893614	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893614 <u>Rev:</u> N/C <u>Date:</u> 12/16/1996, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1
Gear, Helical Powertrain Pinion	E6893672	6893672	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893672 <u>Rev:</u> N/C <u>Date:</u> 12/02/1996, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C47B, C47M
Gearshaft Assembly, Helical Torquemeter	E6893673	6893673	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893673 <u>Rev:</u> N/C <u>Date:</u> 11/01/1996, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30R/1, C30R/3, C47B, C47M

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Deflector, Shield, Turbine Nozzle First Stage	E6893676	6893676	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893676 <u>Rev:</u> N/C <u>Date:</u> 08/19/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2
Pin	E6894196	6894196	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6894196 <u>Rev:</u> N/C <u>Date:</u> 11/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C
Plate, Retaining - Gas Producer Thrust Bearing	E6895005	6895005	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6895005 <u>Rev:</u> B <u>Date:</u> 11/20/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W

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Seal, Labyrinth GP Turbine, Forward	E6896098	6896098	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6896098 <u>Rev:</u> N/C <u>Date:</u> 10/24/1995, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Nut, Spline Lock	E6896408	6896408	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6896408 <u>Rev:</u> N/C <u>Date:</u> 11/07/1995, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G/2
Support, Torque Bearing and Shaft	E6898558	6898558	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898558 <u>Rev:</u> N/C <u>Date:</u> 01/16/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Coupling, Adapter Compressor	E6898561	6898561	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898561 <u>Rev:</u> B <u>Date:</u> 10/30/1992, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W

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Seal Assembly, Oil Bellows	E6898764	6898764	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898764 <u>Rev:</u> B <u>Date:</u> 08/20/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30G, C30M, C30P, C30S
Adapter, Splined, Second Stage Turbine	E6898784	6898784	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898784 <u>Rev:</u> A <u>Date:</u> 10/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Gearshaft, Assembly Helical Power Take-Off	E6898785	6898785	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898785 <u>Rev:</u> N/C <u>Date:</u> 11/25/1996, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30R/1, C30R/3, C47B, C47M
Cover, Assembly, Oil GP Turbine Bearing	E6898945	6898945	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898945 <u>Rev:</u> N/C <u>Date:</u> 12/20/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M

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Coupling, Turbine to Compressor	E6898977	6898977	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898977 <u>Rev:</u> C <u>Date:</u> 10/30/1992, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Gearshaft, Spur, Adapter	E6899239	6899239	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6899239 <u>Rev:</u> D <u>Date:</u> 12/29/1992, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Vane and Band Assembly – First Stage	E6899391	6899391	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6899391 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Vane and Band Assembly – Second Stage	E6899392	6899392	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6899392 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Vane and Band Assembly – Third Stage	E6899393	6899393	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6899393 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Vane and Band Assembly – Fourth Stage	E6899394	6899394	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6899394 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W

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Vane and Band Assembly – 5 th Stage	E6899395	6899395	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6899395 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Vane and Band Assembly – 6 th Stage	E6899396	6899396	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6899396 <u>Rev:</u> N/C <u>Date:</u> 05/12/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Machined Vane Ring, First Stage Nozzle	E8064VR	23008064	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E8064VR <u>Rev:</u> N/C <u>Date:</u> 09/27/1995, or later FAA approved revisions	Rolls-Royce	250-C30, C30M, C30P, C30S, C30G, C30G/2
Adapter, Turbine Spline	EH23030974	23030974	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23030974 <u>Rev:</u> N/C <u>Date:</u> 03/04/1997, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Shaft, Power Turbine Inner, LH Thread	EH23037403	23037403	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23037403 <u>Rev:</u> N/C <u>Date:</u> 02/05/1997, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Shaft, PT Outer	EH23037413	23037413	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23037413 <u>Rev:</u> N/C <u>Date:</u> 02/05/1997, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1

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ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Coupling, Adapter Compressor	EH6898561	6898561	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH6898561 <u>Rev:</u> N/C <u>Date:</u> 06/17/1994, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Adapter, Splined, Second Stage Turbine	EH6898784	6898784	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH6898784 <u>Rev:</u> N/C <u>Date:</u> 06/17/1994, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Pin	E3103067-4	3103067-4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3103067-4 <u>Rev:</u> N/C <u>Date:</u> 04/15/1998, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 series as installed on: 757-200, 200PF, 200CB series; 767-200, -300, - 300F series
				Airbus	Honeywell GTCP331-250 series as installed on: A300 Model(s) B4-600, B4-600R, F4-600R series; A310 Models 200, 300, 325 series

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Shaft, Oil	E3600604-2	3600604-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3600604-2 <u>Rev:</u> N/C <u>Date:</u> 08/07/1998, or later FAA approved revisions	Honeywell TSCP700-4B TSCP700-5 series as installed on: Boeing Airbus	DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F (KC- 10A) -40, -40F A300 Model B2-1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203
Blade, HP Turbine Rotor	E3604912-6	3604912-6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3604912-6 <u>Rev:</u> A <u>Date:</u> 12/15/1999, or later FAA approved revisions	Honeywell TSCP700-4B TSCP700-4E TSCP700-5 as installed on: Boeing Airbus	DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F (KC-10A) MD-11, MD-11F A300 Model B2-1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203; A300 Model B4-600 series, B4-600R series, F4-600R; A310 model 200 series, 300 series

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Turbine Shroud	E3611904-1	3611904-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3611904-1 <u>Rev:</u> N/C <u>Date:</u> 01/07/1998, or later FAA approved revisions	Boeing	Honeywell GTCP85 series as installed on: 727 Series; 727-200, 200F Series; 737-200, 300, 400, 500 Series DC-8-71, 71F, 72, 72F, 73, 73F; DC-9-11, 12, 13, 14, 15, 15F, 21, 31, 32(VC-9C), 32F, 32F(C- 9A,C-9B), 33F, 34, 34F, 41, 51, 81,-82, 83, 87; MD-88 British Aerospace 1-11-200 Series 1-11-400 Series Lockheed Martin 382, 382B, 382E, 382F, 382G
Seal, Stationary	E3615081-1,-2	3615081-1,-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3615081-1, -2 <u>Rev:</u> N/C <u>Date:</u> 04/15/1998, or later FAA approved revisions	Boeing Airbus	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series; 767-200, 300, 300F series A300 Model(s) B4-600,B4-600R, F4-600R series; A310 Models 200, 300, 325 series

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Sector Gear - Spur IGV	E3810119-2	3810119-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3810119-2 <u>Rev:</u> N/C <u>Date:</u> 07/06/1998, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series; 767-200, 300, 300F series
				Airbus	A300 Model(s) B4-600, B4-600R, F4-600R series; A310 Models 200, 300, 325 series
Vane Inlet	E3826099-1, -3, -4, -5	3826099-1, -3, -4, -5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3826099-1, -3, -4, -5 <u>Rev:</u> N/C <u>Date:</u> 08/13/1998, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series; 767-200, 300, 300F series
				Airbus	A300 Model(s) B4-600, B4-600R, F4-600R series A310 Models 200, 300, 325 series

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Carrier - Seal	E3826127-1	3826127-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3826127-1 <u>Rev:</u> N/C <u>Date:</u> 07/28/1998, or later FAA approved revisions		Honeywell GTCP331-200 GTCP331-250 series as installed on:
				Boeing	757-200, 200PF, 200CB series
					767-200, 300, 300F series
				Airbus	A300 Model(s) B4-600, B4-600R, F4-600R series
					A310 Models 200, 300, 325 series
Blade, First Stage Turbine Rotor	E3840004-5	3840004-5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3840004-5 <u>Rev:</u> A <u>Date:</u> 07/01/1998, or later FAA approved revisions		Honeywell GTCP331-200 GTCP331-250 series as installed on:
				Boeing	757-200, 200PF, 200CB series
					767-200, 300, 300F series
				Airbus	A300 Model(s) B4-600, B4-600R, F4-600R series;
					A310 Models 200, 300, 325 series

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Blade, Second Stage Turbine Rotor	E3840008-2	3840008-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3840008-2 <u>Rev:</u> A <u>Date:</u> 07/01/1998, or later FAA approved revisions		Honeywell GTCP331-200 GTCP331-250 series as installed on: Boeing 757-200, 200PF, 200CB series 767-200, 300, 300F series Airbus A300 Model(s) B4-600, B4-600R, F4-600R series; A310 Models 200, 300, 325 series

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Nozzle, Turbine	E3846484-3, -4, -7, -8	3846484-3, -4, -7, -8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3846484 <u>Rev:</u> N/C <u>Date:</u> 02/05/1998, or later FAA approved revisions	Boeing	Honeywell GTCP85 series as installed on: 727-200, 200F Series 737-300, 400, 500 Series DC-8-71, 71F, 72, 72F, 73F; DC9-11, 12, 13, 14, 15, 15F, 21, 31, 32(VC-9C), 32F, 32F(C-9A, C-9B), 33F, 34, 34F, 41, 51, 81, 82, 83, 87; MD-88 British Aerospace 1-11 200 Series 1-11-400 Series Lockheed Martin 382, 382B, 382E, 382F, 382G Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series 767-200, 300, 300F series Airbus A300 Model(s) B4- 600, B4-600R, F4-600R series A310 Models 200, 300, 325 series
Bushing - Sleeve	E3862152-1, -2, -3, -4, -5, -6	3862152-1, -2, -3, -4, -5, -6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3862152 <u>Rev:</u> N/C <u>Date:</u> 06/15/1998, or later FAA approved revisions	Boeing	757-200, 200PF, 200CB series 767-200, 300, 300F series Airbus A300 Model(s) B4- 600, B4-600R, F4-600R series A310 Models 200, 300, 325 series

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Bushing - Sleeve	E3862154-1, -2, -3, -4, -5, -6, -7	3862154-1, -2, -3, -4, -5, -6, -7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3862154 <u>Rev:</u> N/C <u>Date:</u> 06/30/1998, or later FAA approved revisions		Honeywell GTCP331-200 GTCP331-250 series as installed on: Boeing 757-200, 200PF, 200CB series 767-200, 300, 300F series Airbus A300 Model(s) B4- 600, B4-600R, F4-600R series; A310 Models 200, 300, 325 series

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Body, Valve	E663017-4	663017-4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E663017-4 <u>Rev:</u> N/C <u>Date:</u> 12/17/1997, or later FAA approved revisions	Boeing	Honeywell GTCP85 GTCP36 series as installed on: 727-100, 100C Series; 727-200, 200F Series; 737-100, 200, 200C, 300, 400, 500 Series DC-8-71, 71F, 72, 72F, 73, 73F; DC-9-11, 12, 13, 14, 15, 15F, 21, 31, 32(VC-9C), 32F, 32F(C-9A,C-9B), 33F, 34, 34F, 41, 51, 81,-82, 83, 87; MD-88 British Aerospace 1-11-200 Series 1-11-400 Series Dassault Aviation Mercure 100C Airbus A320 Model-111, 211, 231

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		APPROVED REPLACEMENT	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
ARTICLE NAME	ARTICLE NUMBER	FOR ARTICLE NUMBER			
Bushing	E693472-1, -2, - 3, -4, -5, -8, -11, -25	693472-1, -2, -3, - 4, -5, -8 -11, -25	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E693472 <u>Rev:</u> N/C <u>Date:</u> 07/29/1998, or later FAA approved revisions	Boeing	Honeywell GTCP85 series as installed on: 727, 727C Series; 727-100, 100C, 200, 200F Series 737-100, 200, 200C, 300, 400, 500 Series DC8-71, -71F, 72F, 73, 73F, DC-9-11, 12, 13, 14, 15, 15F, 21, 31, 32(VC-9C), 32F, 32F(C-9A,C-9B), 33F, 34, 34F, 41, 51, 81, 82, 83, 87; MD-88 British Aerospace 1-11-200 Series; 1-11-400 Series Lockheed Martin 188A, 188C, 382, 382B, 382E, 382F, 382G Dassault Aviation Mercure 100C SUD Aviation Caravelle SE 210 Model I, III, VIR

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Bushing	E693473-1, -2, -3, -4, -5	693473-1, -2, -3, -4, -5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E693473 <u>Rev:</u> N/C <u>Date:</u> 07/29/1998, or later FAA approved revisions	Boeing	Honeywell GTCP85 series as installed on: 727, 727C Series; 727-100, 100C, 200, 200F Series 737-100, 200, 200C, 300, 400, 500 Series DC-8-71, 71F, 72, 72F, 73, 73F DC-9-11, 12, 13, 14, 15, 15F, 21, 31, 32(VC-9C), 32F, 32F(C-9A,C-9B), 33F, 34, 34F, 41, 51, 81, 82, 83, 87; MD-88 British Aerospace 1-11-200 Series; 1-11-400 Series Lockheed Martin 188A, 188C, 382, 382B, 382E, 382F, 382G Dassault Aviation Mercure 100C SUD Aviation Caravelle SE 210 Model I, III, VIR

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Seal Assembly, Rotating	E868186-1, -2, - 3, -4	868186-1, -2, -3, - 4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E868186 <u>Rev:</u> N/C <u>Date:</u> 03/26/1998, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series 767-200, 300, 300F series
				Airbus	A300 Model(s) B4- 600, B4-600R, F4-600R series A310 Models 200, 300, 325 series
Vane Inlet	EN3826099 -1, -3, -4, -5	3826099 -1, -3, -4, -5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EN3826099 <u>Rev:</u> N/C <u>Date:</u> 08/13/1998, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series 767-200, 300, 300F series
				Airbus	A300 Model(s) B4- 600, B4-600R, F4-600R series A310 Models 200, 300, 325 series

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Bushing - Sleeve	EZ3862152 -1,-2,-3,-4,-5, -6	3862152- 1, -2, -3, -4,-5, -6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EZ3862152 <u>Rev:</u> N/C <u>Date:</u> 11/06/1998, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series 767-200, 300, 300F series
				Airbus	A300 Model(s) B4- 600, B4-600R, F4-600R series A310 Models 200, 300, 325 series
Bushing - Sleeve	EZ3862154 -1,-2,-3,-4,-5,- 6,-7	3862154 -1,-2,-3,-4,-5, -6,-7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EZ3862154 <u>Rev:</u> N/C <u>Date:</u> 11/06/1998, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB series 767-200, 300, 300F series
				Airbus	A300 Model(s) B4- 600, B4-600R, F4-600R series A310 Models 200, 300, 325 series

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Blade, LPC 1 st Stage	E969650-1	969650-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E969650-1 <u>Rev:</u> B <u>Date:</u> 09/14/2000, or later FAA approved revisions		Honeywell TSCP700-4B TSCP700-4E TSCP700-5 as installed on:
				Boeing	DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F(KC- 10A)
					MD-11, 11F
				Airbus	A300 models B2- 1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203, B4-600 series B4-600R series, F4- 600R series
					A310-200 series, 300 series
Blade, LPC 1 st Stage	E3614183-2	3614183-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3614183-2 <u>Rev:</u> B <u>Date:</u> 09/14/2000, or later FAA approved revisions		Honeywell TSCP700-4B TSCP700-4E TSCP700-5 as installed on:
				Boeing	DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F(KC- 10A)
					MD-11, MD-11F
				Airbus	A300 models B2-1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203, B4-600 series, B4-600R series, F4-600R series
					A310-200 series, 300 series

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Blade, LPC 3 rd Stage	E969651-1	969651-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E969651-1 <u>Rev:</u> B <u>Date:</u> 09/20/2000, or later FAA approved revisions		Honeywell TSCP700-4B TSCP700-4E TSCP700-5 as installed on: Boeing DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F(KC- 10A) MD-11, MD-11F Airbus A300 models B2-1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203, B4-600 series, B4-600R series, F4-600R series A310-200 series, 300 series

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Blade, LPC 1st Stage	E969603-3	969603-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E969603-3 <u>Rev:</u> A <u>Date:</u> 08/24/2000, or later FAA approved revisions		Honeywell TSCP700-4B TSCP700-4E TSCP700-5 as installed on: Boeing DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F(KC- 10A) MD-11, MD-11F Airbus A300 models B2-1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203, B4-600 series, B4-600R series, F4-600R series A310-200 series, 300 series

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Vane, LPC 2nd Stage	E969604-5	969604-5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E969604-5 <u>Rev:</u> A <u>Date:</u> 08/24/2000, or later FAA approved revisions		Honeywell TSCP700-4B TSCP700-4E TSCP700-5 as installed on: Boeing DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F(KC- 10A) MD-11, MD-11F Airbus A300 models B2-1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203, B4-600 series, B4-600R series, F4-600R series A310-200 series, 300 series

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Vane, LPC 3rd Stage	E3604714-1	3604714-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3604714-1 <u>Rev:</u> B <u>Date:</u> 09/14/2000, or later FAA approved revisions		Honeywell TSCP700-4B TSCP700-4E TSCP700-5 as installed on: Boeing DC-10-10, 10F, 15, 30, 40, 40F; DC-10-30F(KC- 10A) MD-11, MD-11F Airbus A300 models B2-1A, B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203, B4-600 series, B4-600R series, F4-600R series A310-200 series, 300 series
Nozzle Assembly Turbine, 1 st Stage, C20R	E23038219-247- P	23038219	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23038219-247-P <u>Rev:</u> N/C <u>Date:</u> 08/01/2000, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4
Gearshaft, Spur Adapter	E23056633	23056633	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23056633 <u>Rev:</u> N/C <u>Date:</u> 11/06/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30S, C40B, C47B, C47M
Gearshaft Assembly Spur Adapter	E23056634	23056634	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23056634 <u>Rev:</u> N/C <u>Date:</u> 11/06/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30S, C40B, C47B, C47M

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Ring Retaining, 7.250 in.	E23071792	23071792	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23071792 <u>Rev:</u> N/C <u>Date:</u> 07/24/2000, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Ring, Retaining, Internal 5.604 OD Balanced	E23071793	23071793	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> E23071793 <u>Rev:</u> N/C <u>Date:</u> 07/24/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Gearshaft, Spur-Fuel Control & Oil Pump	E6896437	6896437	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6896437 <u>Rev:</u> N/C <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S
Gear - IGV Ring	E3826102-1	3826102-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3826102-1 <u>Rev:</u> A <u>Date:</u> 01/24/2001, or later FAA approved revisions		Honeywell GTCP331-200 GTCP331-250 series as installed on:
				Boeing	757-200 series, 200PF series, 200CB series;
					767-200 series, 300, 300F series
				Airbus	A300 Models B4-601, B4-603, B4-620, B4-605R, B4-622R, B4-605R
					A310 Models 221, 222, 203, 322, 324, 304, 325

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Guide Vane, Matched	E3826825-3	3826825-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3826825-3 <u>Rev:</u> A <u>Date:</u> 06/12/2001, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Piston Ring, Split Seal 2.800 Inch OD	E6889096	6889096	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6889096 <u>Rev:</u> N/C <u>Date:</u> 07/05/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Seal, Ring, Compressor Discharge Tube	E6898657	6898657	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898657 <u>Rev:</u> N/C <u>Date:</u> 07/05/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Thermo-couple Assembly TOT Gas Producer	E23034926	23034926	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23034926 <u>Rev:</u> N/C <u>Date:</u> 06/25/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28, C28B, C28C
Nozzle Assembly, Turbine, 4 th Stage	E23061922	23061922	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23061922 <u>Rev:</u> N/C <u>Date:</u> 06/25/2001, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Gearshaft Assembly Spur Adapter	E23073520	23073520	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23073520 <u>Rev:</u> N/C <u>Date:</u> 06/21/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30S, C40B, C47B, C47M
Gearshaft, Spur Adapter	E23073521	23073521	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> E23073521 <u>Rev:</u> N/C <u>Date:</u> 06/21/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30S, C40B, C47B, C47M

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Nozzle Assembly, Turbine, Second Stage	E23053330	23053330	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23053330 <u>Rev:</u> N/C <u>Date:</u> 12/20/2000, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30U, C40B, C47B, C47M
Ring, Energy Absorbing	E23054648	23054648	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23054648 <u>Rev:</u> N/C <u>Date:</u> 01/26/2001, or later FAA approved revisions	Rolls-Royce	250-C30R/3, C40B, C47B, C47M
Nozzle Assembly, Turbine, 2 nd Stage	E23064622	23064622	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23064622 <u>Rev:</u> N/C <u>Date:</u> 12/20/2000, or later FAA approved revisions	Rolls-Royce	250-C30R/3, C30U, C40B, C47B, C47M
Shield Assembly Turbine Nozzle - 1 st Stage	E23073533	23073533	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23073533 <u>Rev:</u> N/C <u>Date:</u> 03/05/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Shield Assembly Turbine Nozzle - 1 st Stage	E23073566	E23073566	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> E23073566 <u>Rev:</u> N/C <u>Date:</u> 03/05/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Seal Labyrinth - PT Rotating	E23037444	23037444	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23037444 <u>Rev:</u> N/C <u>Date:</u> 07/27/2001, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Blade, Turbine	3013102E	3013102	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3013102E <u>Rev:</u> N/C <u>Date:</u> 08/13/2001, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 110, 114, 114A
Gearshaft Assembly, Helical- Torquemeter	E23035299	23035299	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23035299 <u>Rev:</u> C <u>Date:</u> 10/16/2001, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20R, C20R/1, C20R/2, C20S
Sump Assembly Scavenge Oil C20	E23053320	23053320	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23053320 <u>Rev:</u> N/C <u>Date:</u> 01/15/2002, or later FAA approved revisions	Rolls-Royce	250- B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20S, C20W, C28B, C28C
Sump Assembly Scavenge Oil C30	E6893029	6893029	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893029 <u>Rev:</u> N/C <u>Date:</u> 01/15/2002, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Bracket Assembly, Fireshield	E6853366	6853366	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6853366 <u>Rev:</u> A <u>Date:</u> 01/30/2002, or later FAA approved revisions	Rolls-Royce	250-C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/4, C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30R/2, C30S, C30U, C40B, C47B, C47M

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Ring Gear Assembly	E3612887-1	3612887-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3612887-1 <u>Rev:</u> A <u>Date:</u> 09/13/2001, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200 series, 200PF series, 200CB series 767-200 series, 300 series, 300F series
				Airbus	A300 Model B4-600 series, B4-600R series, F4-600R, C4-600R series A310 Model 200 series, 300 series
Bearing, Assembly Gear Drive Journal	E3612702-1	3612702-1	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> E3612702-1 <u>Rev:</u> C <u>Date:</u> 05/15/2002, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200 series, 200PF series, 200CB series; 767-200 series, 300 series, 300F series
				Airbus	A300 Model B4-600 Series, B4-600R series, F4-600R series, C4-600R series A310 Models 200 series, 300 series
Bolt, Slab Head 0.190-32 X 0.500	E23033399	23033399	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23033399 <u>Rev:</u> N/C <u>Date:</u> 02/28/2002, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30 G/2, C30M, C30P, C30S

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Ring, Energy Absorbing	E23035175	23035175	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23035175 <u>Rev:</u> N/C <u>Date:</u> 02/13/2002, or later FAA approved revisions	Rolls-Royce	250-B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Pin, Stepped	E23064575	23064575	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23064575 <u>Rev:</u> N/C <u>Date:</u> 01/4/2002, or later FAA approved revisions	Rolls-Royce	250-C40B, C47B, C47M
Diffuser Assembly, Compressor	E23064637	23064637	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23064637 <u>Rev:</u> N/C <u>Date:</u> 11/20/2001, or later FAA approved revisions	Rolls-Royce	250-C40B, C47B, C47M
Blade, Turbine	3030102E	3030102	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3030102E <u>Rev:</u> N/C <u>Date:</u> 04/08/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6B-36, 36A, 36B; PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B
Tube Assembly, F/C to Fire Shield	E6890580	6890580	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6890580 <u>Rev:</u> N/C <u>Date:</u> 03/20/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30S, C30U
Tube Assembly, F/C to Accumulator	E6890581	6890581	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6890581 <u>Rev:</u> N/C <u>Date:</u> 03/20/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30P

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Tube Assembly F/C to Governor	E6890583	6890583	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6890583 <u>Rev:</u> N/C <u>Date:</u> 03/20/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S
Tube Assembly, Accumulator to Governor	E6890584	6890584	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6890584 <u>Rev:</u> N/C <u>Date:</u> 03/20/2002, or later FAA approved revisions	Rolls-Royce	250-C28C, C30, C30P, C30S
Tube Assembly, PC Filter to Governor	E23001986	23001986	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23001986 <u>Rev:</u> N/C <u>Date:</u> 03/20/2002, or later FAA approved revisions	Rolls-Royce	250-C28C, C30M, C30P
Tube Assembly, F/C to PC Tee	E6893073	6893073	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6893073 <u>Rev:</u> N/C <u>Date:</u> 03/20/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30S
Tube Assembly, PY Air, F/C to Governor	E23039785	23039785	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23039785 <u>Rev:</u> N/C <u>Date:</u> 03/20/2002, or later FAA approved revisions	Rolls-Royce	250-C30
Nozzle Assembly 1 st Stage C40/C47	E23062061	23062061	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062061 <u>Rev:</u> N/C <u>Date:</u> 04/04/2002, or later FAA approved revisions	Rolls-Royce	250-C30R/3, C40B, C47B, C47M

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Rotor Seal	E3822057-3	3822057-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3822057 <u>Rev:</u> N/C <u>Date:</u> 03/19/2002, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200 series, 200PF series, 200CB series 767-200 series, 300 series, 300F series
				Airbus	A300 Model B4-600 Series, B4-600R series, F4-600R series, C4-600R series A310 Models 200 series, 300 series
Inlet Guide Vane	3827076-3E	3827076-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3827076-3E <u>Rev:</u> N/C <u>Date:</u> 04/04/2002, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: B-777-200
Inlet Guide Vane	3827179-3E	3827179-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3827179-3E <u>Rev:</u> N/C <u>Date:</u> 04/04/2002, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: B-777-200

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Blade, Turbine, 1st Stage	3840223-4E	3840223-4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3840223-4E <u>Rev:</u> N/C <u>Date:</u> 04/24/2002, or later FAA approved revisions		Honeywell GTCP131-9[A] GTCP131-9[B] GTCP131-9[D] as installed on:
				Airbus	Model A319-111; 112, 113; 114; 115; 131, 132;-133 Model A320-111, 211, 212, 214, 231, 232, 233 Model A321-111, 112, 131, 211, 231
				Boeing	737-600, 700, 800, 900 MD-90
Gearshaft, Starter - C30	E6892285	6892285	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6892285 <u>Rev:</u> N/C <u>Date:</u> 02/28/2002, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Gearshaft, Centrifugal Breather - C30	E6898642	6898642	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898642 <u>Rev:</u> A <u>Date:</u> 05/08/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Gearshaft, Generator Idler - C30	E6898981	6898981	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6898981 <u>Rev:</u> N/C <u>Date:</u> 02/27/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M

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Separator, Oil Pump Body Pressure & Scavenge	E23030986	23030986	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23030986 <u>Rev:</u> A <u>Date:</u> 04/18/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30R, C30S
Gearshaft, 77 Tooth, Idler - C30	E23007246	23007246	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23007246 <u>Rev:</u> N/C <u>Date:</u> 02/26/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Gearshaft, 26 Tooth Idler - C30	E23007247	23007247	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23007247 <u>Rev:</u> N/C <u>Date:</u> 02/27/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Gearshaft, F/C Oil Pump -C30	E23007248	23007248	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23007248 <u>Rev:</u> N/C <u>Date:</u> 02/28/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M

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Stator 1 st Stage GTCP 331-200	E3844038-4	3844038-4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3844038-4 <u>Rev:</u> C <u>Date:</u> 07/25/2002, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200 series, 200PF series, 200CB series 767-200 series, 300 series, 300F series
				Airbus	A300 Model B4-600 series, B4-600R series, F4-600R series, C4-600R series A310 Models 200 series, 300 series
Nozzle Assembly, Turbine C20 2 nd Stage	E23072654	23072654	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23072654 <u>Rev:</u> B <u>Date:</u> 01/28/2003, or later FAA approved revisions	Rolls-Royce	250-C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4
Nozzle Assembly, Turbine C20 2 nd Stage	E23072654-UC	23072654, 23074472	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23072654-UC <u>Rev:</u> B <u>Date:</u> 01/28/2003, or later FAA approved revisions	Rolls-Royce	250-C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4
Rotor Seal	E3827086-2	3827086-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3827086-2 <u>Rev:</u> N/C <u>Date:</u> 07/03/2002, or later FAA approved revisions	Honeywell	GTCP36-150[RJ] GTCP36-150[A], - 150[AA] as installed on
				Bombardier	CL-600-2B19
				Embraer S.A.	EMB-120, 120RT, 120ER, 120FC

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Secondary Fuel Manifold	E3830156-2	3830156-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3830156-2 <u>Rev:</u> N/C <u>Date:</u> 08/02/2002, or later FAA approved revisions		Honeywell GTCP36-150[RJ] GTCP36-150[A] GTCP36-150[AA] GTCP36-150[F] GTCP36-150[CX] GTCP36-150[M] GTCP36-150[W] as installed on: Bombardier CL-600-2B19 Embraer S.A. EMB-120, 120RT, 120ER, 120FC Dassault Aviation Mystere-Falcon 900, 900EX BAE Systems (Operations) Limited BAe 146-200A; Avro 146-RJ70A, RJ85A, RJ100A SAAB Aircraft AB SAAB SF340A; SAAB 340B Bombardier DHC-8-100 series, 200 series, 300 series, 400 series Cessna Aircraft Company 650 (Citation III) 750 (Citation X) Beechcraft Corporation BAe.125 Series 800A, 800B, 1000A, 1000B Gulfstream Aerospace LP Astra 1125 Dassault Aviation Mystere-Falcon 20- C5, D5, E5, F5 Fokker Services F 28Mark 0100 B.V.

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Outer Pilot Ring	E3842347-1	3842347-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3842347-1 <u>Rev:</u> A <u>Date:</u> 10/01/2002, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Inner Pilot Ring	E3842348-1	3842348-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3842348-1 <u>Rev:</u> A <u>Date:</u> 10/01/2002, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19

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Planet Gear Shaft	E3860005-2	3860005-2	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> E3860005-2 <u>Rev:</u> N/C <u>Date:</u> 07/03/2002, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19 Honeywell GTCP36-150[A], GTCP36-150[AA] as installed on:
				Embraer S.A.	EMB-120, 120RT, 120ER, 120FC Honeywell GTCP36-150[F] as installed on:
				Dassault Aviation	Mystere-Falcon 900, 900EX Honeywell GTCP36-150[CX] as installed on:
				Cessna Aircraft Company	750 (Citation X) Honeywell GTCP36-150[M] as installed on:
				BAE Systems (Operations) Limited	BAe 146-200A, 300A; Avro 146-RJ70A, RJ85A, RJ100A Honeywell GTCP36-150[W] as installed on:
				SAAB Aircraft AB	SAAB SF340A SAAB 340B
				Bombardier	DHC-8-100 series, 200 series, 300 series, 400 series
				Cessna Aircraft Company	650 (Citation III) Page 80 of 289
				Beechcraft	BAe 125 series -

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Nut	E3860675-1	3860675-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3860675-1 <u>Rev:</u> A <u>Date:</u> 10/01/2002, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Gearset Matched, Spur	E3860339-1	3860339-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3860339-1 <u>Rev:</u> A <u>Date:</u> 11/13/2002, or later FAA approved revisions	Bombardier Embraer S.A.	Honeywell GTCP36-150[RJ] GTCP36-150[A], GTCP36-150[AA] as installed on: CL-600-2B19 EMB-120, 120RT, 120ER, 120FC
Gearshaft, Generator Idler	E3860404-1	3860404-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3860404-1 <u>Rev:</u> A <u>Date:</u> 11/13/2002, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19

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Airflow Deflector	E3846206-3	3846206-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3846206-3 <u>Rev:</u> N/C <u>Date:</u> 08/15/2002, or later FAA approved revisions		Honeywell GTCP36-150[RJ] GTCP36-150[A], GTCP36-150[AA] GTCP36-150[F] GTCP36-150[CX] Pre GTE 1267 GTCP36-150[M] GTCP36-150[W] as installed on: Bombardier CL-600-2B19 Embraer S.A. EMB-120, 120RT, 120ER, 120FC Dassault Aviation Mystere-Falcon 900, 900EX Cessna Aircraft Company 650 (Citation III) 750 (Citation X) BAE Systems (Operations) Limited BAe 146-200A, 300A Avro 146-RJ70A, RJ85A, RJ100A SAAB Aircraft AB SAAB SF340A SAAB 340B Bombardier DHC-8-100 series, 200 series, 300 series, 400 series
Rotor Seal	E3616104-1	3616104-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3616104-1 <u>Rev:</u> N/C <u>Date:</u> 10/01/2002, or later FAA approved revisions		Honeywell GTCP36-150[RJ] as installed on: Bombardier CL-600-2B19

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Rotor Seal	E3822493-1	3822493-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3822493-1 <u>Rev:</u> N/C <u>Date:</u> 03/25/2002, or later FAA approved revisions	Airbus	Honeywell GTCP36-300[A] as installed on: Model A319-111; 112, 113; 114; 131, 132 Model A320-111, 211, 212, 214, 231, 232, 233 Model A321-111, 112, 131, 212, 231
Deswirl – Compressor	E3826695-1	3826695-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3826695-1 <u>Rev:</u> N/C <u>Date:</u> 09/30/2002, or later FAA approved revisions	Bombardier Fokker Services B.V.	Honeywell GTCP36-150[RJ] GTCP36-150[RR] as installed on: CL-600-2B19 F.28 Mark 0070 F.28 Mark 0100
Housing, Aft Diffuser	E3826779-1	3826779-1	Test and Computations per 14 CFR §21.303 <u>DWG:</u> E3826779-1 <u>Rev:</u> N/C <u>Date:</u> 09/03/2002, or later FAA approved revisions	Bombardier Fokker Services B.V.	Honeywell GTCP36-150[RJ] GTCP36-150[RR] as installed on: CL-600-2B19 F.28 Mark 0070; F.28 Mark 0100

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Primary Fuel Manifold	E3830155-2	3830155-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3830155-2 <u>Rev:</u> N/C <u>Date:</u> 10/01/2002, or later FAA approved revisions		Honeywell GTCP36-150[RJ] GTCP36-150[A], GTCP36-150[AA] GTCP36-150[F] GTCP36-150[CX] GTCP36-150[M] GTCP36-150[W] as installed on: Bombardier CL-600-2B19 Embraer S.A. EMB-120, 120RT, 120ER, 120FC Dassault Aviation Mystere-Falcon 900, 900EX Cessna Aircraft Company 750 Citation X 650 Citation III BAE Systems (Operations) Limited BAe 146-200A; Avro 146-RJ70A, RJ85A, RJ100A SAAB Aircraft AB SAAB SF340A SAAB 340B Bombardier DHC-8-100 series, 200 series, 300 series, 400 series Beechcraft Corporation BAe.125 series 800A, 800B, 1000A, 1000B Gulfstream Aerospace LP Astra 1125 Dassault Aviation Mystere-Falcon 200-C5, D5, E5, F5

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Turbine Nozzle	E3613101-8	3613101-8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3613101-8 <u>Rev:</u> A <u>Date:</u> 05/05/2003, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Bearing	3101985-01E	3101985-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3101985E <u>Rev:</u> N/C <u>Date:</u> 08/21/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 6/C20, 20, 20A, 20B, 21, 25, 25A, 27, 28, 34, 34AG, 34B, 36, 38, 41AG, 41, 42, 62, 135, 135A
Bearing	3101985-02E	3101985-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3101985E <u>Rev:</u> N/C <u>Date:</u> 08/21/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 6/C20, 20, 20A, 20B, 21, 25, 25A, 27, 28, 34, 34AG, 34B, 36, 38, 41AG, 41, 42, 62, 135, 135A
Tube Assembly PC Filter to Governor Tee	E23033885	23033885	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23033885 <u>Rev:</u> N/C <u>Date:</u> 07/31/2002, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30S
Gearshaft Assembly Helical Power Take-Off	E23068295	E23068295	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23068295 <u>Rev:</u> A <u>Date:</u> 06/16/2003, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30S, C30P, C30M, C30R, C30R/1, C30U, C30L, C30R/3, C47B, C47M
Shroud Segment, Turbine Compressor	3031670CL4E	3031670CL4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL5E	3031670CL5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6

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Shroud Segment, Turbine Compressor	3031670CL6E	3031670CL6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL7E	3031670CL7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL8E	3031670CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL9E	3031670CL9	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL10E	3031670CL10	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL11E	3031670CL11	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL12E	3031670CL12	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6

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Shroud Segment, Turbine Compressor	3031670CL13E	3031670CL13	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL14E	3031670CL14	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL15E	3031670CL15	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3031670CL16E	3031670CL16	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3031670CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6
Shroud Segment, Turbine Compressor	3018503CL8E	3018503CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL9E	3018503CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6

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ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Shroud Segment, Turbine Compressor	3018503CL10E	3018503CL10	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL11E	3018503CL11	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL12E	3018503CL12	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL13E	3018503CL13	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL14E	3018503CL14	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL15E	3018503CL15	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6

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Shroud Segment, Turbine Compressor	3018503CL16E	3018503CL16	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL17E	3018503CL17	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL18E	3018503CL18	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL19E	3018503CL19	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Shroud Segment, Turbine Compressor	3018503CL20E	3018503CL20	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3018503CLE <u>Rev:</u> N/C <u>Date:</u> 10/11/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3D, 3DE, 3DF, 6
Port Plate	E3881123-1	3881123-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3881123-1 <u>Rev:</u> N/C <u>Date:</u> 01/31/2003, or later FAA approved revisions	Bombardier.	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19

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Port Plate	E3881124-1	3881124-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3881124-1 <u>Rev:</u> N/C <u>Date:</u> 01/22/2003, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Port Plate	E3881125-1	3881125-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3881125-1 <u>Rev:</u> N/C <u>Date:</u> 12/13/2002, or later FAA approved revisions	Bombardier.	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Port Plate	E3881126-1	3881126-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3881126-1 <u>Rev:</u> N/C <u>Date:</u> 12/13/2002, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Port Plate	E3881127-1	3881127-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3881127-1 <u>Rev:</u> N/C <u>Date:</u> 12/13/2002, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Shroud Segment, High Pressure Turbine	E976866-4	976866-4 EE826891-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> E976866-4 <u>Rev:</u> N/C <u>Date:</u> 08/26/2003, or later FAA approved revisions	Boeing Airbus	Honeywell TSCP700-4E TSCP700-4B TSCP700-5 as installed on: DC-10-10, 15, 30, 40, 10F, 30F MD-11, 11F A300 B2-1A B2-1C, B4-2C, B2K-3C, B4-103, B2-203, B4-203

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Cover Assembly	E3881122-1	3881122-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3881122-1 <u>Rev:</u> N/C <u>Date:</u> 07/07/2003, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-1A11, 2A12, 2B16, 2B19, 2C10
Diffuser Assembly Compressor, C-47	E23064637-2P	23064637	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23064637-2P <u>Rev:</u> N/C <u>Date:</u> 02/22/2002, or later FAA approved revisions	Rolls-Royce	250-C40B, C47B, C47M install per EXTEX Service Bulletin T-068
Seal, Omega- Ring 6.395 ID X 6.610 OD X .091 Inch	E23060164	23060164	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23060164 <u>Rev:</u> N/C <u>Date:</u> 05/13/2004, or later FAA approved revisions	Rolls-Royce	250-C30R/3, C40B, C47B, C47M
Ring, Split	E3844006-1	3844006-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3844006-1 <u>Rev:</u> N/C <u>Date:</u> 04/23/2004, or later FAA- approved revisions.	Boeing	Honeywell GTCP331-200 series as installed on: 757-200, 200PF, 200CB Series; 767-200, 300, 300F Series
				Airbus	A300 Model B4-601, B4-603, B4-620, B4-650R, B4-622R, F4-605R, F4-622R; A310 Model 221, 222, 203, 322, 324, 304, 325

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Retainer, Piston Seal	E3844058-1	3844058-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3844058-1 <u>Rev:</u> N/C <u>Date:</u> 04/23/2004, or later FAA- approved revisions.	Boeing	Honeywell GTCP331-200 series as installed on: 757-200, 200PF, 200CB Series; 767-200, 300, 300F Series
				Airbus	A300 Model B4-601, B4-603, B4-620, B4-650R, B4-622R, F4-605R, F4-622R; A310 Model 221, 222, 203, 322, 324, 304, 325
Shroud Segment 1 st Stage Turbine	3844481-2E	3844481-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3844481-2E <u>Rev:</u> A <u>Date:</u> 04/07/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: 777-200, 300, 300ER
Gear Sector, Spur	3827041-1E	3827041-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3827041-1E <u>Rev:</u> A <u>Date:</u> 09/30/2004, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: 777-200,-300, 300ER series
Vane Assembly	3827140-2E	3827140-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3827140-2E <u>Rev:</u> N/C <u>Date:</u> 01/27/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: 777-200,-300, 300ER series

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Deswirl	3827141-2E	3827141-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3827141-2E <u>Rev:</u> A <u>Date:</u> 03/17/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: 777-200,-300, 300ER series
Turbine Seal	3844514-1E	3844514-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3844514-1E <u>Rev:</u> A <u>Date:</u> 05/05/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: 777-200,-300, 300ER series
Insulation Blanket	E3826185-4	3826185-4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3826185-4 <u>Rev:</u> NC <u>Date:</u> 03/22/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-200[A], 200[AC], 200[ER] series as installed on: 757-200, 200PF, 200CB, 300 Series 767-200, 300, 300F Series
Blade, Turbine	E3013102	3013102	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3013102 <u>Rev:</u> N/C <u>Date:</u> 07/27/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 110, 114, 114A
Blade, Turbine	E3030102	3030102	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3030102 <u>Rev:</u> N/C <u>Date:</u> 07/27/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6B-36, 36A, 36B PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B

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Seal, Air, Compressor Rotor	E3017967	3017967	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3017967 <u>Rev:</u> N/C <u>Date:</u> 05/23/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 65B, 65R PT6B-36, 36A, 36B, 37A PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF
Seal Assembly, Air, Power Turbine Stator	E3031377	3031377	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031377 <u>Rev:</u> N/C <u>Date:</u> 05/23/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 62, 65B, 65R, 65AR, 65AG, 60A, 60AG, 61
Air Seal	E3100013-01	3100013-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031377 <u>Rev:</u> N/C <u>Date:</u> 05/23/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 62, 65B, 65R, 65AR, 65AG, 60A, 60AG, 61
Input Drive Gearshaft	E3011406	3011406	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3011406 <u>Rev:</u> B <u>Date:</u> 04/08/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 35, 36, 110, 112, 114, 114A, 121, 135, 135A

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Compressor Stubshaft	E3102336-01	3102336-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3102336-01 <u>Rev:</u> A <u>Date:</u> 04/08/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 112, 114, 114A, 121, 135, 135A; PT6B-36, 36A; PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF
Input Drive Coupling Shaft, Accessory Gearbox	E3113770-01	3113770-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3113770-01 <u>Rev:</u> B <u>Date:</u> 04/08/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 65B, 65R, 65AR, 65AG
Static Ring Seal	3844467-1E	3844467-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3844467-1E <u>Rev:</u> B <u>Date:</u> 04/12/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: 777-200, 300, 300ER

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Oil Pump Spur Gear Assembly	E3860165-2	3860165-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3860165-2 <u>Rev:</u> D <u>Date:</u> 09/09/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB Series; 767-200, 300, 300F Series
				Airbus	A300 Models: B4-601, B4-603, B4-620, B4-605R, B4-622, B4-622R, C4-605R variant F, F4-605R, F4-622R A310 Models: 203, 204, 221, 222, 304, 322, 324, 325
Compressor Rotor Spline Adapter	E23051117-1	23051117-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051117 <u>Rev:</u> A <u>Date:</u> 08/29/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C; 250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S; 250-C40B, C47B, C47M
Compressor Rotor Spline Adapter	E23051117-2	23051117-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051117 <u>Rev:</u> A <u>Date:</u> 08/29/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C; 250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S; 250-C40B, C47B, C47M

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Compressor Rotor Spline Adapter	E23051117-3	23051117-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23051117 <u>Rev:</u> A <u>Date:</u> 08/29/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C; 250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S; 250-C40B, C47B, C47M
Compressor Rotor Spline Adapter	EH23051117-1	23051117-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23051117 <u>Rev:</u> A <u>Date:</u> 08/29/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C; 250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S; 250-C40B, C47B, C47M
Compressor Rotor Spline Adapter	EH23051117-2	23051117-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23051117 <u>Rev:</u> A <u>Date:</u> 08/29/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C; 250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S; 250-C40B, C47B, C47M
Compressor Rotor Spline Adapter	EH23051117-3	23051117-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> EH23051117 <u>Rev:</u> A <u>Date:</u> 08/29/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C; 250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S; 250-C40B, C47B, C47M

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Blade, CT	E3123131-02	3123131-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3123131-02 <u>Rev:</u> A <u>Date:</u> 11/21/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-41, 42, 42A, 45A, 45B, 45R, 50
Blade, CT	E3123071-01	3123071-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3123071-01 <u>Rev:</u> N/C <u>Date:</u> 08/25/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-35, 135, 135A; PT6B-36A; PT6T-3, 3B, 3BE, 3BF, 3BG, 6, 6B
Splined Adapter, Turbine	E23076529	23076529	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23076529 <u>Rev:</u> B <u>Date:</u> 10/25/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C 250-C30, C30G, C30G/2, C30M, C30P, C30S, C30R, C30R/1, C30U; 250-C40B, C47B, C47M
Nut, Spline Lock	E23076531	23076531	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23076531 <u>Rev:</u> B <u>Date:</u> 10/13/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C 250-C30, C30G, C30G/2, C30M, C30P, C30S, C30R, C30R/1, C30U; 250-C40B, C47B, C47M

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Gear-Starter	E3860168-2	3860168-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3860168-2 <u>Rev:</u> A <u>Date:</u> 08/30/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 as installed on: 757-200 series, 200PF series, 200CB series; 767-200 series, 300 series, 300F series Honeywell GTCP331-250 series as installed on:
				Airbus	A300 Models: B4-600 series, B4-600R series, F4-600R series, C4-600R series A310 Models: 200 series, 300 series
Gear-Spur Fan	E3860169-5	3860169-5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3860169-5 <u>Rev:</u> A <u>Date:</u> 08/30/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-200 GTCP331-250 series as installed on: 757-200, 200PF, 200CB Series 767-200, 300, 300F Series
				Airbus	A300 Models: B4-600 series, B4-600R series, F4-600R series, C4-600R series A310 Models: 200 series, 300 series

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SUPPLEMENT NO.: 1
DATE: May 9, 2016
REISSUED: September 8, 2016

ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Blade Assembly- Cooled	E3108125-1	3108125-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3108125-1 <u>Rev:</u> A <u>Date:</u> 04/14/2006, or later FAA approved revisions	Honeywell	TPE331-10, 10GP, 10GT, 10N, 10P, 10R, 10T, 10U, 10UA, 10UF, 10UG, 10UGR, 10UR, 11U
Gearshaft, Spur, Oil Pump Drive	E3008127	3008127	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3008127 <u>Rev:</u> A <u>Date:</u> 02/27/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B, 11, 11AG, 15AG, 110, 112, 121, 25, 25A, 25C, 38, 41, 42, 42A, 21, 27, 28, 34, 34AG, 34B, 36, 45A, 45B, 45R, 50, 60A, 60AG, 61, 64, 65B, 65R, 65AR, 65AG, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D, 114, 114A, 135, 135A
Coupling, Power Turbine Shaft	E3109766-01	3109766-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3109766 <u>Rev:</u> A <u>Date:</u> 02/27/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 65B, 65R, 65AR, 65AG
Seal, Air, Turbine, Interstage Rotor	E3023683	3023683	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3023683 <u>Rev:</u> A <u>Date:</u> 11/04/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 64, 65B, 65R, 66, 65AG, 65AR, 67, 67A, 67B, 67D, 67R, 67AF, 67AG, 67T

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

EXTEX Engineered Products Inc.
340 E Germann Road, Suite 113
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Seal Assembly	E3036777	3036777	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3036777 <u>Rev:</u> N/C <u>Date:</u> 12/12/2005, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 10, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 35, 36, 38, 41, 41AG, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 62, 65B, 65R, 65AR, 65AG, 67, 67A, 67R, 67T, 67AF, 67AG, 110, 112, 114, 114A, 121, 135, 135A
Shroud Segment, Turbine Compressor	E3031670CL4	3031670CL4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL5	3031670CL5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL6	3031670CL6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL7	3031670CL7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B

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Shroud Segment, Turbine Compressor	E3031670CL8	3031670CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL9	3031670CL9	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL10	3031670CL10	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL11	3031670CL11	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL12	3031670CL12	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL13	3031670CL13	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B

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Shroud Segment, Turbine Compressor	E3031670CL14	3031670CL14	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL15	3031670CL15	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3031670CL16	3031670CL16	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3031670CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 6, 6B, 3D, 3DE, 3DF; PT6B-36B
Shroud Segment, Turbine Compressor	E3018503CL4	3018503CL4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-50
Shroud Segment, Turbine Compressor	E3018503CL6	3018503CL6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-50
Shroud Segment, Turbine Compressor	E3018503CL8	3018503CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B

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Shroud Segment, Turbine Compressor	E3018503CL9	3018503CL9	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL10	3018503CL10	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL11	3018503CL11	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL12	3018503CL12	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B

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Shroud Segment, Turbine Compressor	E3018503CL13	3018503CL13	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL14	3018503CL14	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL15	3018503CL15	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 50, 45A, 45B, 45R, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL16	3018503CL16	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B

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Shroud Segment, Turbine Compressor	E3018503CL17	3018503CL17	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL18	3018503CL18	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL19	3018503CL19	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL20	3018503CL20	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 50, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B

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Bearing	E3101985-01	3101985-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3101985 <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 6/C20, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 27, 28, 34, 34AG, 34B, 36, 38, 40, 41AG, 41, 42, 42A, 61, 61A, 62, 112
Bearing	E3101985-02	3101985-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3101985 <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 6/C20, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 27, 28, 34, 34AG, 34B, 36, 38, 40, 41AG, 41, 42, 42A, 61, 61A, 62, 112
Ring, Sealing, Power Output Section	E3100236-01	3100236-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3100236-01 <u>Rev:</u> N/C <u>Date:</u> 07/05/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A, 60A, 61, 62, 65B, 65R, 65AR, 65AG
Stub Shaft, Compressor front	E3102336-02	3102336-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3102336-02 <u>Rev:</u> N/C <u>Date:</u> 02/15/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110
Seal, Air, Power Turbine Rotor	E3109767-01	3109767-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3109767 <u>Rev:</u> N/C <u>Date:</u> 05/19/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-45A, 45B, 45R, 50, 60A, 60AG, 65B, 65R, 65AR, 65AG

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Seal, Air, Power Turbine Rotor	E3109767-02	3109767-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3109767 <u>Rev:</u> N/C <u>Date:</u> 05/19/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 61
Shroud Segment, Turbine Compressor	E3018503CL4	3018503CL4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 114, 114A, 135, 135A PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL5	3018503CL5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 114, 114A, 135, 135A; PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL6	3018503CL6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 114, 114A, 135, 135A; PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Shroud Segment, Turbine Compressor	E3018503CL7	3018503CL7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018503CL <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 35, 36, 38, 41, 42, 45A, 45B, 45R, 114, 114A, 135, 135A; PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B

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Parts Kit, Second-Stage Planet Gear Bearing	E3035030	3035030	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3101985 <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 27, 28, 34, 34AG, 34B, 36, 38, 40, 41, 41AG, 42, 42A, 6/C20, 61, 61A, 62, 112
Parts Kit, First-Stage Planet Gear Bearing	E3035031	3035031	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3101985 <u>Rev:</u> N/C <u>Date:</u> 02/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 27, 28, 34, 34AG, 34B, 36, 38, 40, 41, 41AG, 42, 42A, 6/C20, 61, 61A, 62, 112
Blade, Compressor	E3036701	3036701	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3036701 <u>Rev:</u> E <u>Date:</u> 10/27/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 114, 114A, 135, 135A; PT6B-36, 36A; PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF
Gearshaft Assembly, Helical- Torquemeter	E23075442	23035299	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23075442 <u>Rev:</u> NC <u>Date:</u> 11/22/2006, or later FAA approved revisions	Rolls-Royce	250-B17; B17B; B17C; B17D; B17E; B17F; B17F/1; B17F/2; C20; C20B; C20C; C20F; C20R; C20R/1; C20R/2; C20S
Ring, Sealing, Power Output Section	E3014660	3014660	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3014660 <u>Rev:</u> N/C <u>Date:</u> 11/01/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 36, 38, 41, 42, 42A, 45A, 45B, 45R , 50, 135; PT6T-3BE, 3B, 3, 6, 6B

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Ring, Sealing, Power Output Section	E3024992	3024992	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3024992 <u>Rev:</u> N/C <u>Date:</u> 11/01/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 36, 114, 114A, 135, 135A PT6T-3BE, 3B, 3DF, 3D, 3, 3DE, 6, 6B
Ring, Retaining, Vane Ring	E3120116-01	3120116-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3120116-01 <u>Rev:</u> NC <u>Date:</u> 08/28/2006, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 62, 65AG, 65AR, 65B, 65R
Cover, Deswirl	E3827135-1	3827135-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3827135-1 <u>Revisions:</u> N/C <u>Date:</u> 12/16/2005, or later FAA approved revisions	Boeing	Honeywell GTCP331-500[B] series as installed on: 777-200, 300, 300ER Series
Nozzle - Turbine	E3846779-1	3846779-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3846779-1 <u>Rev:</u> N/C <u>Date:</u> 09/11/2006, or later FAA approved revisions	Bombardier	Honeywell GTCP36-150[RJ] as installed on: CL-600-2B19
Shroud Segments, Compressor Turbine	E3037347CL12	3037347CL12	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL13	3037347CL13	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG

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Shroud Segments, Compressor Turbine	E3037347CL14	3037347CL14	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL15	3037347CL15	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL16	3037347CL16	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL17	3037347CL17	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL18	3037347CL18	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL19	3037347CL19	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL20	3037347CL20	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG

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Shroud Segments, Compressor Turbine	E3037347CL21	3037347CL21	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL22	3037347CL22	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3037347CL23	3037347CL23	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037347CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 62, 65B, 65R, 65AR, 65AG
Shroud Segments, Compressor Turbine	E3039465CL11	3039465CL11	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL12	3039465CL12	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL13	3039465CL13	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL14	3039465CL14	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D

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Shroud Segments, Compressor Turbine	E3039465CL15	3039465CL15	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL16	3039465CL16	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL17	3039465CL17	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL18	3039465CL18	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL19	3039465CL19	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL20	3039465CL20	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL21	3039465CL21	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D

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Shroud Segments, Compressor Turbine	E3039465CL22	3039465CL22	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL23	3039465CL23	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Shroud Segments, Compressor Turbine	E3039465CL24	3039465CL24	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3039465CL <u>Rev:</u> NC <u>Date:</u> 02/05/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66D, 68, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Blade, Turbine	E3840282-3	3840282-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3840282-3 <u>Rev:</u> N/C <u>Date:</u> 05/02/2007, or later FAA approved revisions		Honeywell GTCP131-9[A] GTCP131-9[B] GTCP131-9[D] as installed on:
				Airbus	Model A318-111, 112, 121, 122 Model A319-111, 112, 113, 114, 115, 121, 122, 131, 132, 133 Model A320- 100/200 series 111, 211, 212, 214, 231, 232, 233 Model A321- 100/200 series 111, 112, 131, 211, 212, 213, 231, 232
				Boeing	737-600, 700, 800, 900 MD-90

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Shaft Assembly	3500058-1R	3500058-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3500058R <u>Rev:</u> C <u>Date:</u> 01/17/2007, or later FAA approved revisions	Lockheed Martin	Honeywell ATS100-242 series starter as installed on: 188A, 188C 382, 382B, 382E, 382F, 382G, 382J
Shaft Assembly	3500058-5R	3500058-5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3500058R <u>Rev:</u> C <u>Date:</u> 01/17/2007, or later FAA approved revisions	Lockheed Martin	Honeywell ATS100-242 series starter as installed on: 188A, 188C 382, 382B, 382E, 382F, 382G, 382J
Shroud Segment, Compressor Turbine	E3019438CL1	3019438CL1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121
Shroud Segment, Compressor Turbine	E3019438CL2	3019438CL2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121
Shroud Segment, Compressor Turbine	E3019438CL3	3019438CL3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121

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Shroud Segment, Compressor Turbine	E3019438CL4	3019438CL4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121
Shroud Segment, Compressor Turbine	E3019438CL5	3019438CL5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121
Shroud Segment, Compressor Turbine	E3019438CL6	3019438CL6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121
Shroud Segment, Compressor Turbine	E3019438CL7	3019438CL7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121
Shroud Segment, Compressor Turbine	E3019438CL8	3019438CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121
Shroud Segment, Compressor Turbine	E3019438CL9	3019438CL9	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019438CL <u>Rev:</u> NC <u>Date:</u> 08/10/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 110, 112, 121

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Ring, Sealing, Power Turbine Housing	E3057818-01	3057818-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3057818-01 <u>Rev:</u> N/C <u>Date:</u> 01/02/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 66, 66A, 66B, 67, 67A, 67AF, 67AG, 67B, 67D, 67R, 67T
Ring, Retaining, Compressor Turbine, Shroud	E3110741-02	3110741-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3110741-02 <u>Rev:</u> N/C <u>Date:</u> 03/11/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-25C, 34, 34AG, 34B, 36, 38, 41, 42, 42A, 45A, 45B, 45R, 50, 114, 114A, 135, 135A
Ring, Retaining, Compressor Turbine, Shroud	E3112397-01	3112397-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3112397-01 <u>Rev:</u> N/C <u>Date:</u> 03/11/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61, 60AG, 65B, 65R, 65AR, 65AG
Blade, Turbine, Compressor	E3045741-01	3045741-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3045741-01 <u>Rev:</u> N/C <u>Date:</u> 02/20/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11; 11AG, 15AG, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 38, 41, 110, 112, 114, 114A, 121; PT6D-114A
Seal, Air, Turbine Interstage Stator	E3023682	3023682	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3023682 <u>Rev:</u> N/C <u>Date:</u> 03/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 64, 65AG, 65AR, 65B, 65R, 66, 66B, 67, 67A, 67AF, 67AG, 67B, 67D, 67R, 67T, 68

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Gear Sector, Spur	E3810794-2	3810794-2	Test and Computations per 14 CFR §21.303 <u>DWG. No:</u> E3810794-2 <u>Rev:</u> N/C <u>Date:</u> 02/29/2008, or later FAA approved revisions		Honeywell 131-9[A] as installed on:
				Airbus	A319 series A320 series A321series
					Honeywell 131-9[B] as installed on
				Boeing	737-600, 700, 800, 900 series
					Honeywell 131-9[D] as installed on:
				Boeing	MD-90-30
					Honeywell GTCP36-280[B] as installed on:
				Boeing	B737-300, 400, 500 series
					Honeywell GTCP 36-280[D] as installed on:
				Boeing	DC-9-81, 82, 83, 87 MD-80 MD-88
					Honeywell GTCP 36-300[A] as installed on:
				Airbus	A320 series

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Fuel Nozzle Sheath	E3119920-01	3119920-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3119920-01 <u>Rev:</u> N/C <u>Date:</u> 04/16/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 40, 41, 41AG, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 64, 65AG, 65AR, 65B, 65R, 66, 66A, 66B, 66D, 67, 67A, 67AF, 67AG, 67B, 67D, 67R
Runner, Propeller Shaft Oil Seal	E3111231-01	3111231-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3111231-01 <u>Rev:</u> N/C <u>Date:</u> 02/22/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-65AR, 67, 67A, 67AF, 67AG, 67B, 67D, 67R, 67T
Runner, Propeller Shaft Oil Seal	E3114662-01	3114662-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3114662-01 <u>Rev:</u> N/C <u>Date:</u> 02/22/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34B, 34AG, 35, 36, 38, 41, 42, 42A, 61, 64, 66, 66B, 110, 112, 114, 114A, 121, 135, 135A
Runner, Propeller Shaft Oil Seal	E3114664-01	3114664-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3114664-01 <u>Rev:</u> N/C <u>Date:</u> 02/04/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-45A, 45B, 45R, 60A, 60AG, 65B, 65R, 65AR, 65AG
Pin	TAISGT-537-1	SGT537-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT-537 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Liner, Inboard Bearing	TAI-4011-129-1	204-011-129-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-129 <u>Rev:</u> I/R <u>Date:</u> 01/09/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212

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Liner, Outboard Bearing	TAI-4011-130-1	204-011-130-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-130 <u>Rev:</u> I/R <u>Date:</u> 01/09/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212
Bushing	TAI-4010-413- 27	204-010-413-27	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-413 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212
Bushing	TAI-4010-413- 29	204-010-413-29	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-413 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212
Pin, Lower, Latch	TAI-5030-517- 001	205-030-517-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5030-517 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212, 412
Barrel-Control Link, Pitch Change, M/R	TAI-4011-137-1	204-011-137-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-137 <u>Rev:</u> I/R <u>Date:</u> 01/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A 205A-1, 212
Static Stop M/R	TAI-2010-102- 001	212-010-102-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2010-102 <u>Rev:</u> I/R <u>Date:</u> 01/10/2006, or later FAA approved revisions	Bell Helicopter Textron	212
Shield, Shield Assembly, M/R	TAI-4012-118-1	204-012-118-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4012-118 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Sleeve, Seal, Strap Pitting Outboard M/R	TAI-4012-107-1	204-012-107-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4012-107 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Liner, Bearing Scissors Pivot	TAI-4010-454- 003	204-010-454-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-454 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212
Liner, Collective Sleeve	TAI-4011-443- 003	204-011-443-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-443 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Clevis Assembly Ram Ground Handling	TAI-4050-170- 001	204-050-170-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4050-170 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 214B, 412
Glass, Sight Gage, Oil Gearbox	TAI-4040-514- 001	204-040-514-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-514 <u>Rev:</u> <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 206A, 206B, 206L, 206L-1, 206L-3, 212, 214B, 412
Liner, Outer Ring	TAI-4011-425-1	204-011-425-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-425 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 212
Bearing & Liner, Assembly	TAI-4010-433-1	204-010-433-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-433 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Shim	TAI-4010-434-1	204-010-434-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-434 <u>Rev:</u> I/R <u>Date:</u> 01/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Bearing, & Liner, Assembly	TAI-4011-443- 001	204-011-443-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-443 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412

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Nut, Drag Brace	TAI-4011-144- 001	204-011-144-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-144 <u>Rev:</u> I/R <u>Date:</u> 01/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A
Adapter, Front & Back	TAI-5070-786- 001	205-070-786-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5070-786 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 412
Exhaust Duct, Assembly of	TAI-2061-202-9	212-061-202-9	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2061-202 <u>Rev:</u> I/R <u>Date:</u> 03/09/2006, or later FAA approved revisions	Bell Helicopter Textron	212 412
Exhaust Duct, Assembly of	TAI-2061-202- 10	212-061-202-10	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2061-202 <u>Rev:</u> I/R <u>Date:</u> 03/09/2006, or later FAA approved revisions	Bell Helicopter Textron	212 412
Bushing	TAI-4010-413- 165	204-010-413-165	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-413 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B
Plate	TAI-4010-779-9	204-010-779-9	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-779 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B
Plate, Swashplate	TAI-4011-458-1	204-011-458-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-458 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Plate, Swashplate	TAI-4011-458-3	204-011-458-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-458 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Plate, Swashplate	TAI-4011-458-5	204-011-458-5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-458 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Plate, Swashplate	TAI-4011-458-7	204-011-458-7	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-458 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Spacer, Inner Swashplate Assembly	TAI-4011-460-1	204-011-460-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-460 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-1	204-011-464-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-3	204-011-464-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-105	204-011-464-105	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-107	204-011-464-107	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-109	204-011-464-109	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Bushing	TAI-4011-464-111	204-011-464-111	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-113	204-011-464-113	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-115	204-011-464-115	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing	TAI-4011-464-117	204-011-464-117	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-464 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Support-Roller Cargo Door	TAI-5031-433-005	205-031-433-005	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-433 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Support-Roller Cargo Door	TAI-5031-433-006	205-031-433-006	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-433 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Transparent Window	TAISGT528-1	SGT528-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT528 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing Flange	TAI-6010-425-003	206-010-425-003	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6010-425 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B, 206BIII, 206L, 206L-1, 206L-3, 206L-4

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Sleeve, Inboard Seal M/R	TAI-6010-245- 101	206-010-245-101	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6010-245 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B
Liner, Bearing Rotating Controls	TAI-4010-422- 021	204-010-422-021	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-422 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Race-Controls Tail Rotor	TAI-9011-717- 001	209-011-717-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-9011-717 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214, 412
Bushing- Support Assembly	TAI-4011-497- 101	204-011-497-101	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-497 <u>Rev:</u> I/R <u>Date:</u> 02/03/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 412
Bushing- Support Assembly	TAI-4011-497- 103	204-011-497-103	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-497 <u>Rev:</u> I/R <u>Date:</u> 02/03/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 412
Pad, Grip- Main Rotor Blade	TAI-6010-233- 005	206-010-233-005	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6010-233 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B, 206L, 206L-1
Boot-Tail Rotor	TAI-6011-736- 001	206-011-736-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6011-736 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B, 206L, 206L-1, 206L-3
Clevis- Baggage Door Upper	TAI-6031-325- 001	206-031-325-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6031-325 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B, 206L, 206L-1, 206L-3

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Bushing, Hinge Door, Crew	TAI-4030-052- 001	204-030-052-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-052 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B
Bushing, Hinge Door, Crew	TAI-4030-052- 003	204-030-052-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-052 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B
Hinge Half Lower Cabin Assembly of	TAI-4031-467- 005	204-031-467-005	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-467 <u>Rev:</u> I/R <u>Date:</u> 01/31/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B
Hinge Half Lower Cabin Assembly of	TAI-4031-467- 006	204-031-467-006	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-467 <u>Rev:</u> I/R <u>Date:</u> 01/31/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B
Hinge Half Lower Cabin Assembly of	TAI-4031-467- 011	204-031-467-011	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-467 <u>Rev:</u> I/R <u>Date:</u> 01/31/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Hinge Half Lower Cabin Assembly of	TAI-4031-467- 012	204-031-467-012	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-467 <u>Rev:</u> I/R <u>Date:</u> 01/31/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Hinge Half Upper Cabin Assembly of	TAI-4031-837- 005	204-031-837-005	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-837 <u>Rev:</u> I/R <u>Date:</u> 01/25/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Hinge Half Upper Cabin Assembly of	TAI-4031-837- 006	204-031-837-006	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-837 <u>Rev:</u> I/R <u>Date:</u> 01/25/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Hinge Half Upper Cabin Assembly of	TAI-4031-837-011	204-031-837-011	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-837 <u>Rev:</u> I/R <u>Date:</u> 01/25/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Hinge Half Upper Cabin Assembly of	TAI-4031-837-012	204-031-837-012	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-837 <u>Rev:</u> I/R <u>Date:</u> 01/25/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Seal - Cargo Door Assembly of	TAI-5030-550-001	205-030-550-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5030-550 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Seal - Cargo Door Assembly of	TAI-5030-550-002	205-030-550-002	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5030-550 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Seal - Cargo Door Assembly of	TAI-5030-551-001	205-030-551-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5030-551 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Seal - Cargo Door Assembly of	TAI-5030-551-002	205-030-551-002	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5030-551 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Ring - Safety Belt Attach	TAI-5070-725-001	205-070-725-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5070-725 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 412
Ring - Safety Belt Attach	TAI-5070-725-003	205-070-725-003	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5070-725 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 412

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ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Ring - Safety Belt Attach	TAI-5070-725- 009	205-070-725-009	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5070-725 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 412
Ring - Safety Belt Attach	TAI-5070-725- 011	205-070-725-011	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5070-725 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 412
Rod End, Latch Crew Door	TAI-4030-478- 001	204-030-478-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-478 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Rod Extension Latch, Crew Door	TAI-4030-479- 001	204-030-479-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-479 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Push Rod, Crew Door Latch Assembly of	TAI-4030-481- 001	204-030-481-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-481 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Push Rod, Crew Door Latch Assembly of	TAI-4030-481- 101	204-030-481-101	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-481 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Rod, Crew Door Latch, Assembly of	TAI-4030-483- 001	204-030-483-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-483 <u>Rev:</u> I/R <u>Date:</u> 02/27/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Rod, Crew Door Latch, Assembly of	TAI-4030-483- 003	204-030-483-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-483 <u>Rev:</u> I/R <u>Date:</u> 02/27/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412

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APPROVED REPLACEMENT			APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
ARTICLE NAME	ARTICLE NUMBER	FOR ARTICLE NUMBER			
Clevis, Latch Crew Door	TAI-4030-484- 001	204-030-484-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-484 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Roller, Assembly Latch, Crew Door	TAI-4030-499- 005	204-030-499-005	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-499 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Roller, Latch Crew Door	TAI-4030-501- 001	204-030-501-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-501 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Reservoir, Main Rotor Trunnion	TAI-4010-191- 001	204-010-191-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-191 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Inner Race, Drive Link Bearing	TAI-4010-441- 003	204-010-441-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-441 <u>Rev:</u> I/R <u>Date:</u> 12/03/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Housing Assembly Drive Link Thrust Washer	TAI-4011-421- 001	204-011-421-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-421 <u>Rev:</u> I/R <u>Date:</u> 02/03/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Retainer- Grease Spherical Coupling Main D/S	TAI-5040-175- 001	205-040-175-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5040-175 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Cap, Outer Race Bearing Retainer	TAI-4011-426-1	204-011-426-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-426 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Cap, Inner Race Bearing Retainer	TAI-4011-427-1	204-011-427-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-427 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Cover, Swashplate Bearing	TAI-4011-428-3	204-011-428-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-428 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Shim, Inner Ring Swashplate	TAI-4011-431-1	204-011-431-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-431 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Decal, Barrel Drag Brace Assembly	TAI-4011-146- 003	204-011-146-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-146 <u>Rev:</u> I/R <u>Date:</u> 01/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B
Tube, Tail Rotor Controls	TAI-9011-709-3	209-011-709-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-9011-709 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212, 412
Clevis, Control Tube, Tail Rotor	TAI-9011-718-1	209-011-718-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-9011-718 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212, 412
Sleeve, Bearing Inner Race S/P Inner Ring	TAI-4011-423-1	204-011-423-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-423 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Liner Assembly, Trunnion Bearing, Main Rotor	TAI-4011-122- 001	204-011-122-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-122 <u>Rev:</u> I/R <u>Date:</u> 01/06/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212

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Bushing- Controls Tail Rotor	TAI-9011-716- 001	209-011-716-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-9011-716 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	212, 412
Insulator Slip Ring Cargo Suspension	TAI-4075-155- 001	204-075-155-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4075-155 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 212, 412
Insulator Slip Ring Cargo Suspension	TAI-4075-155- 003	204-075-155-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4075-155 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 212, 412
Inner Ring, Slip Cargo Suspension, Assembly of	TAI-4075-156- 001	204-075-156-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4075-156 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 212, 412
Outer Ring, Slip Cargo Suspension, Assembly of	TAI-4075-156- 003	204-075-156-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4075-156 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 212, 412
Ground Ring, Slip Cargo Suspension, Assembly of	TAI-4075-156- 005	204-075-156-005	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4075-156 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 212, 412
Ring, Slip Cargo Suspension, Assembly of	TAI-4075-157- 001	204-075-157-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4075-157 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 212, 412
Cap, Landing Gear Fitting, Aft Assembly	TAI-5030-292- 011	205-030-292-011	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5030-292 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A

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Cap, Landing Gear Fitting, Aft Assembly	TAI-5030-292- 012	205-030-292-012	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5030-292 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A
Support-Roller Cargo Door, Upper Fwd	TAI-5031-435- 001	205-031-435-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-435 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Support-Roller Cargo Door, Upper Fwd	TAI-5031-435- 002	205-031-435-002	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-435 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Plate Supt- Roller Cargo Door, Upper Fwd	TAI-5031-435- 003	205-031-435-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-435 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Support-Roller Cargo Door, Upper Aft	TAI-5031-436- 001	205-031-436-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-436 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Plate Support - Roller Cargo Door, Upper Aft	TAI-5031-436- 003	205-031-436-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-436 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Hinge Half Lower, Crew Door, Assembly of	TAI-4031-468- 101	204-031-468-101	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-468 <u>Rev:</u> I/R <u>Date:</u> 01/27/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Hinge Half Lower, Crew Door, Assembly of	TAI-4031-468- 102	204-031-468-102	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-468 <u>Rev:</u> I/R <u>Date:</u> 01/27/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412

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Hinge Half Upper, Crew Door, Assembly of	TAI-4031-658- 101	204-031-658-101	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-658 <u>Rev:</u> I/R <u>Date:</u> 01/23/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Hinge Half Upper, Crew Door, Assembly of	TAI-4031-658- 102	204-031-658-102	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-658 <u>Rev:</u> I/R <u>Date:</u> 01/23/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Spring, Hinge Half, Assembly of	TAI-4031-762- 001	204-031-762-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-762 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Spring, Hinge Half, Assembly of	TAI-4031-762- 002	204-031-762-002	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-762 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Block, Spring Holder, Crew Door	TAI-4031-766- 001	204-031-766-1 Rev. F	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-766 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212, 214B, 412
Pin, Pivot	TAI-6010-408- 001	206-010-408-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6010-408 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B, 206L, 206L-1, 206L-3
Screw Assembly, Adjusting	TAI-6010-111- 105	206-010-111-105	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6010-111 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B, 206L, 206L-1, 206L-3
Radius Ring Trunnion M/R	TAI-4011-148- 001	204-011-148-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-148 <u>Rev:</u> I/R <u>Date:</u> 01/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Plate Supt- Roller Cargo Door, WL5439	TAI-5031-433- 003	205-031-433-3	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5031-433 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 205A-1, 212, 214B, 412
Universal Control Link, M/R Assembly	TAI-4011-128-1	204-011-128-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-128 <u>Rev:</u> I/R <u>Date:</u> 01/09/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Sleeve, Seal Oil Transfer	1-030-214- 01AL	1-030-214-01	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-214-01AL <u>Rev:</u> A <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Bearing, Ball	358348AL	358348	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 358348AL <u>Rev:</u> I/R <u>Date:</u> 10/28/2003, or later FAA approved revisions	Honeywell	TPE331-1, 2, 2UA, 5, 5A, 5AB, 5B, 6, 6A, 8, 10, 10GP, 10GT, 10N, 10P, 10T, 10UA, 10UF, 10UG, 10UGR, 10UR, 11U, 12, 12UA, 12UAR, 2UHR
Wingshaft	TAISGT-519- 002	SGT519-2	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT-519 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	212
Retainer	TAISGT-543- 001	SGT-543-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT-543 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	212
Retainer	TAISGT-527- 001	SGT-527-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT-527 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1

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Wingshaft	TAISGT-522-2	SGT-522-2	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT-522 <u>Rev:</u> A <u>Date:</u> 02/02/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1
Wingshaft	TAISGT-522-1	SGT-522-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT-522 <u>Rev:</u> A <u>Date:</u> 02/02/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1
Bolt, Isolation Mount-Pylon	TAI-2030-096- 001	212-030-096-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2030-096- <u>Rev:</u> I/R <u>Date:</u> 01/09/2006, or later FAA approved revisions	Bell Helicopter Textron	212, 412
Shaft, Tail Rotor 90° Gear Box	TAI-2040-462- 001	212-040-462-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2040-462 <u>Rev:</u> I/R <u>Date:</u> 01/23/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212, 412
Bolt, Drag Brace	TAI-4011-166- 009	204-011-166-9	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-166 <u>Rev:</u> I/R <u>Date:</u> 01/10/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212
Bolt, Drag Brace	TAI-4011-166- 007	204-011-166-7	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-166 <u>Rev:</u> I/R <u>Date:</u> 01/10/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212
Nut, Blade Main Rotor	TAI-4011-153- 001	204-011-153-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-153 <u>Rev:</u> I/R <u>Date:</u> 01/09/2006 or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Barrel, Rotary Damper	TAISGT-521- 001	SGT-521-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAISGT-521 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Bolt, Lift Link	TAI-4040-815-001	204-040-815-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-815 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A
Bolt, Lift Link	TAI-2040-117-001	212-040-117-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2040-117 <u>Rev:</u> I/R <u>Date:</u> 12/29/2005, or later FAA approved revisions	Bell Helicopter Textron	205A 212
Bolt, Lift Link	TAI-2040-117-103	212-040-117-103	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2040-117 <u>Rev:</u> I/R <u>Date:</u> 12/29/2005, or later FAA approved revisions	Bell Helicopter Textron	212 412
Bushing-Set	TAI-4076-162-001	204-076-162-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4076-162 <u>Rev:</u> I/R <u>Date:</u> 04/17/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A
Plate Assembly-Seal Drive Shaft Coupling, T/R Drive	TAI-4040-610-001	204-040-610-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-610 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Cone Set, Main Rotor Trunion	TAI-4010-127-001	204-010-127-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-127 <u>Rev:</u> I/R <u>Date:</u> 12/29/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Ring, Backup Inboard Tension Strap Retainer	TAI-4011-123-001	204-011-123-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-123 <u>Rev:</u> I/R <u>Date:</u> 01/06/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Bushing, Link Transmission Lift, Assembly of	TAI-4030-909-011	204-030-909-011	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-909 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212

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Bolt, 5th Mount Assembly of	TAI-4031-244- 005	204-031-244-5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4031-244 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Plate, Grip Main Rotor	TAI-4011-125- 005	204-011-125-5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-125 <u>Rev:</u> I/R <u>Date:</u> 03/09/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 205A-1, 212
Spool	TAI-C42641-31	C42641-31	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-C42641 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B
Fitting Assembly-Aft Pylon Support	TAI-6031-592- 001	206-031-592-1,	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6031-592 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B
Fitting Assembly-Aft Pylon Support	TAI-6031-592- 002	206-031-592-2	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6031-592 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A 206B
Fitting Assembly- Forward Pylon Support	TAI-6031-593- 001	206-031-593-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6031-593 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A 206B
Fitting Assembly- Forward Pylon Support	TAI-6031-593- 002	206-031-593-2	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6031-593 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	206A 206B
Spool	TAI-C42641-33	C42641-33	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-C42641-33 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B

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Spool, Bypass	TAI-C42642-38	C42642-38	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-C42642-38 <u>Rev:</u> I/R <u>Date:</u> 01/23/2006, or later FAA approved revisions	Bell Helicopter Textron	206L 206L-1 206L-3
Spool, Servo	TAI-C42642-40	C42642-40	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-C42642-40 <u>Rev:</u> I/R <u>Date:</u> 04/06/2006, or later FAA approved revisions	Bell Helicopter Textron	206L 206L-1 206L-3
Bearing – Ball	383305TA	383305	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 383305TA <u>Rev:</u> I/R <u>Date:</u> 02/13/2007, or later FAA approved revisions	Pratt & Whitney	JT8D-209, 217, 217A, 217C, 219
Bearing - Roller, Assembly	70R782TA	70R782	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 70R782TA <u>Rev:</u> I/R <u>Date:</u> 02/13/2007, or later FAA approved revisions	Pratt & Whitney	PW4052, 4056, 4060, 4060C, 4074, 4074D, 4077, 4077D, 4084D, 4090, 4090D, 4090-3, 4098, 4164, 4168, 4168A
Glass, Sight Gage, Oil Gearbox	TAI-4040-514- 001	204-040-514-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-514 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 206A, 206B, 206L, 206L-1, 206L-3, 212, 214B, 412
Spacer, TR Drive Quill	TAI-4040-313- 001	204-040-313-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-313 <u>Rev:</u> I/R <u>Date:</u> 02/03/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A
Bolt, Spherical Bearing Shaft, Lower Planet, Transmission	TAI-4040-787- 001	204-040-787-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-787 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212

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Case-Support, Bevel Gear, Transmission, Assembly of	TAI-4040-386- 001	204-040-386-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-386 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Spacer, Bolt Lift Link, Transmission	TAI-2040-118- 001	212-040-118-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2040-118 <u>Rev:</u> I/R <u>Date:</u> 12/29/2005, or later FAA approved revisions	Bell Helicopter Textron	205A, 212
Spacer, Bolt Lift Link, Transmission	TAI-2040-118- 101	212-040-118-101	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2040-118 <u>Rev:</u> I/R <u>Date:</u> 12/29/2005, or later FAA approved revisions	Bell Helicopter Textron	212, 412
Liner-Bearing, Upper, Gearshaft, Transmission	TAI-4040-336- 001	204-040-336-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-336 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Cap Assembly Filler, Oil, Gear Box	TAI-4040-502- 009	204-040-502-9	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-502 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 214B, 412
Cap-Oil Filler, Oil Gear Box	TAI-4040-502- 011	204-040-502-011	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-502 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A
Cap-Oil Filler, Gear Box	TAI-4040-503- 003	204-040-503-3	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-503 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 214B, 412
Plug, Oil Filler Gear Box	TAI-4040-505- 005	204-040-505-5	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-505 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 212, 214B, 212

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Plug, Oil Filler Gear Box	TAI-4040-505- 007	204-040-505-7	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-505 <u>Rev:</u> I/R <u>Date:</u> 04/05/2006, or later FAA approved revisions	Bell Helicopter Textron	205A
Case-Top, Transmission Assembly	TAI-4040-359-1	204-040-359-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-359 <u>Rev:</u> I/R <u>Date:</u> 01/25/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212
Support - 5th Mount Transmission Assembly	TAI-4030-061- 001	204-030-061-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4030-061 <u>Rev:</u> I/R <u>Date:</u> 03/08/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A
Shaft-Bearing, Planetary Pinion, Transmission	TAI-4040-130- 005	204-040-130-5	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-130 <u>Rev:</u> I/R <u>Date:</u> 03/08/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Race-Bearing, Planetary Pinion, Transmission	TAI-4040-132- 001	204-040-132-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-132 <u>Rev:</u> I/R <u>Date:</u> 03/09/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212, 412
Nut, Sprocket Retainer Tail Rotor	TAI-4010-767- 001	204-010-767-001	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-767 <u>Rev:</u> I/R <u>Date:</u> 10/03/2007, or later FAA approved revisions	Bell Helicopter Textron	204B
Sleeve Assembly, Spiral Bevel Gear, Hydraulic, Pump & Tachometer Drive, Transmission	TAI-4040-306- 011	204-040-306-011	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-306 <u>Rev:</u> I/R <u>Date:</u> 05/03/2007, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212

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Sleeve Assembly, Spiral Bevel Gear, Hydraulic, Pump & Tachometer Drive, Transmission Housing, Jet, Planetary, Trans, Assembly of	TAI-4040-306- 101	204-040-306-101	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-306 <u>Rev:</u> I/R <u>Date:</u> 05/03/2007, or later FAA approved revisions	Bell Helicopter Textron	205A-1, 212
Bearing, Ball	358894AL-1	358894-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 358894AL-1 <u>Rev:</u> I/R <u>Date:</u> 02/13/2007, or later FAA approved revisions	Honeywell	TPE331-1, 2, 2UA, 3U, 3UW, 5, 5A, 5AB, 5B, 6, 6A, 8, 10, 10F, 10AV, 10GP, 10GT, 10N, 10P, 10R, 10T, 10U, 10UA, 10UF, 10UG, 10UGR, 10UR, 11U, 12, 12B, 12JR, 12UA, 12UAR, 12UHR
Bolt, Machine- Slab Head, 1/4-28 UNJF CRES	6840383TA 6840383TA-012 thru 6840383TA-32	6840383	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6840383TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Bolt, Machine- Drilled 12 Pt Head, .250-28, AMS 5643, Full Shank	6871523TA 6871523TA-011 6871523TA-018 6871523TA-026	6871523	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6871523TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt, Machine- .250-28 UNJF-3A, 12 Point Head, CRES	6841662TA 6841662TA-007 thru 6841662TA-022	6841662	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6841662TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22A, D22C, D22G

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Bolt 5/16-24 X 3/4 Long -12 Point Head CRES	6784788TA	6784788	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6784788TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt-Machine, 12 Point Head, 250-28 X 1.000 Long	6820217TA	6820217	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6820217TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C,
Bushing- Flanged, .500- 20 X .688-16 X .390 Long	6820438TA	6820438	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6820438TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Pin-Hinge .080 X 14.80 Long	6787480TA	6787480	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6787480TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22C
Nut-Spanner 2.3600-16 UNS-3B Thread - Left Hand	6737045TA	6737045	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6737045TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22
Liner Assembly, Combustion	6894068TA	6894068	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6894068TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Saddle, Turbine - 1st Stage	23031744TA	23031744	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031744TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Retainer, Vane Assembly - 2nd Stage	6844619TA	6844619	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6844619TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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Strut Assembly - Red, Gear To Power Section	6825230TA 6825230TA-001 6825230TA-002 6825230TA-003 6825230TA-201 6825230TA-202	6825230	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6825230TA <u>Rev:</u> I/R <u>Date:</u> 10/20/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller, Cylindrical 40mm X 68mm X 15mm	23003798TA	23003798	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23003798TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seal, Labyrinth Rotating Compressor Rear-Primary	23053857TA	23053857	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23053857TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22A, D22C, D22G
Seal, Labyrinth Stator-Rear Compressor, Secondary	23057658TA	23057658	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23057658TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Nut - Spanner .469-32 NS-3 Thread	6721454TA	6721454	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6721454TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt-Prop, Control Shaft Pivot	6819694TA	6819694	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6819694TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Bearing, Roller, Cylindrical 55mm X 90mm X 26mm Wide	6824762TA	6824762	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6824762TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Shaft Assembly, Hydraulic Pump Drive	6825237TA	6825237	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6825237TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Pin Slotted- .406 OD X .579 Long & Oversizes	6828696TA 6828696TA-010	6828696	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6828696TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22
Bushing, Threaded- .625 -18 OD X .375-24 ID, Flanged	6828998TA	6828998	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6828998TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Lock, Pressure Adjusting Valve	6828999TA	6828999	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6828999TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Retainer Assembly, Bearing-Rear Turbine	6870706TA	6870706	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6870706TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006y, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Bearing, Roller-55mm X 90mm X 26 (29)mm	6871201TA	6871201	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6871201TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Body Assembly, Scavenge Oil Pump	6871880TA	6871880	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6871880TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Plate Assembly, Scavenge Oil Pump	6871881TA	6871881	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6871881TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller, Cylindrical- 160mm X 220mm X 28mm Wide	6873232TA	6873232	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6873232TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Bearing, Roller, Cylindrical- 160mm X 220mm X 28mm Wide	6873233TA	6873233	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6873233TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Cage Assembly, Bearing- Turbine Front	6873603TA 6873603TA-010	6873603	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6873603TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D22
Bearing, Roller, Cylindrical- 160mm X 220mm X 28mm Wide	6890542TA	6890542	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6890542TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Lock, Spline- Prop Shaft Thrust Nut	23003503TA	23003503	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23003503TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cuplock, Retention- Prop Shaft Thrust Nut Lock	23003504TA	23003504	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23003504TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Plate, Nose Bearing - Thrust	23034572TA	23034572	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23034572TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft Assembly - Starter	23005956TA	23005956	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23005956TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C
Shaft, Starter	23005957TA	23005957	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23005957TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C
Gear And Bearing Assembly, Planet	23055550TA	23055550	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23055550TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Separator (Cage) Planet Bearing	23056998TA	23056998	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23056998TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Roller Sets, Spherical Planet Bearing	23056999TA 23056999TA- 001 thru 23056999TA- 006	23056999	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23056999TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Gear - Oil Pump Drive	6781679TA	6781679	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6781679TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft- Compressor Extension	6783872TA	6783872	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6783872TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A

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Shaft Gear - Access Drive Side	6787344TA	6787344	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6787344TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft - Compressor Extension	6793016TA	6793016	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6793016TA <u>Rev:</u> I/R <u>Date:</u> 11/03/2006, or later FAA approved revisions	Rolls-Royce	501-D22C, D22G
Seal, Turbine Front Labyrinth - Rear	6844614TA	6844614	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6844614TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22A,D22C, D22G
Cage, Bearing- Accessory Drive Shafts (and oversizes)	23004641TA- 23004641TA- 005 23004641TA- 010 23004641TA- 015 23004641TA- 020	23004641	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004641TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D22C
Cage, Bearing- Main Idler Gear Front (and oversizes)	23004642TA- 23004642TA- 005 23004642TA- 010 23004642TA- 015 23004642TA- 020	23004642	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004642TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D22C
Cage, Bearing- Main Idler Gear Rear (and oversizes)	23004643TA- 23004643TA- 005 23004643TA- 010 23004643TA- 015 23004643TA- 020 23004643TA- 025 23004643TA- 030	23004643	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004643TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D22C, D22G

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Cage, Bearing - EDC Idler Gear Front (and oversizes)	23004644TA 23004644TA- 005 23004644TA- 010 23004644TA- 015 23004644TA- 020 23004644TA- 025 23004644TA- 030	23004644	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004644TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cage, Bearing -EDC Idler Gear Rear (and oversizes)	23004645TA 23004645TA- 005 23004645TA- 010 23004645TA- 015 23004645TA- 020 23004645TA- 025 23004645TA- 030	23004645	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004645TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D22C, D22G
Cage-Bearing 1.3779 ID X .577 Long	6731153TA 6731153TA-001 through 6731153TA-004	6731153	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6731153TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cone, Inner Prop Brake	6792040TA	6792040	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6792040TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear, Spur - Hydraulic Pump Idler Drive	6829392TA	6829392	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6829392TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A

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Gear, Spur - Main Idler	6846363TA	6846363	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6846363TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Ring, Air Seal - 2nd Stage Vane	6892269TA	6892269	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6892269TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C
Roller Bearing	23058579TA	23058579	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058579TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing Roller 160mm X 220mm X 28mm	23058585TA	23058585	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058585TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Bearing Roller 30mm x 55mm x 13mm	23058589TA	23058589	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058589TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller 40mm X 68mm X 15mm	23062205TA	23062205	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23062205TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller 65mm X 106mm X 23 (27)mm	23058593TA	23058593	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058593TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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Bearing, Ball 125mm X 215mm X 36mm	23058582TA	23058582	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058582TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Roller Bearing 160mm X 230mm X 28mm	23058584TA	23058584	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058584TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Roller Bearing 55mm X 105mm X 22mm	23055591TA	23055591	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23055591TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller, 17mm X 35mm X 10mm	23062341TA	23062341	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23062341TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller 17mm X 35mm X 10mm	23062342TA	23062342	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23062342TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller 60mm X 110mm X 22mm	23058594TA	23058594	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058594TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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Bearing, Roller 23mm X 55mm X 16mm	23062238TA	23062238	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23062238TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22
Bearing, Ball, 25mm X 45mm X 12mm	23030560TA	23030560	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23030560TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Ball 60mm X 110mm X 22mm	23057662TA	23057662	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23057662TA <u>Rev:</u> A <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Ball 60mm X 95mm X 18mm	6871077TA	6871077	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6871077TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Ball, 20mm X 47mm X 28mm	23030559TA	23030559	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23030559TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Ball, Annular 30mm X 55mm X 13mm	23030563TA	23030563	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23030563TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Sleeve-Valve Speed Gov	542208TA	542208	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 542208TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Hamilton Sundstrand	54H60-91, 117

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Ring- Retaining	540347TA	540347	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 540347TA <u>Rev:</u> I/R <u>Date:</u> 12/21/2004, or later FAA approved revisions	Hamilton Sundstrand	54H60-91, 117
Ring-Seal Groove, Speed Gov	540338TA	540338	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 540338TA <u>Rev:</u> I/R <u>Date:</u> 10/27/2006, or later FAA approved revisions	Hamilton Sundstrand	54H60-91, 117
Blade, Compressor, 1st Stage	6809081TA	6809081	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809081TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor, 2nd Stage	6809082TA	6809082	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809082TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor, 3rd Stage	6809083TA	6809083	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809083TA <u>Rev:</u> I/R <u>Date:</u> 08/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor, 4th Stage	6809084TA	6809084	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809084TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 5th Stage	6809085TA	6809085	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809085TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Blade, Compressor 6th Stage	6809086TA	6809086	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809086TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 7th Stage	6809087TA	6809087	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809087TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 8th Stage	6809088TA	6809088	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809088TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 9th Stage	6809089TA	6809089	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809089TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 10th Stage	6809090TA	6809090	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809090TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 11th Stage	6809091TA	6809091	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809091TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Blade, Compressor 12th Stage	6809092TA	6809092	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809092TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 13th Stage	6809093TA	6809093	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809093TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Compressor 14th Stage	6809094TA	6809094	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6809094TA <u>Rev:</u> I/R <u>Date:</u> 08/31/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller, Cylindrical	23071567TA	23071567	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23071567TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller, Cylindrical	23058581TA	23058581	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058581TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing, Roller, Cylindrical	23064888TA	23064888	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23064888TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Nut, 3.1875-24 External Thread	6724756TA	6724756	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6724756TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Flange - Compressor., Rear, Bearing Seal	6792257TA	6792257	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6792257TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing, 1.240 ID X 1.509 OD X .552 Long - Flanged	6780888TA	6780888	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6780888TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Key, .125 X 1.500 Dia Special	6813006TA	6813006	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6813006TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shim - 4.223 ID X 5.559 OD X .005 Thick	6821948TA	6821948	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6821948TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Washer, Flat - Countersunk, 1200°F	6825450TA 6825450TA-001 thru 6825450TA-009	6825450	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6825450TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Nut, Spanner - End Slots, Steel Black Oxide	6825540TA 6825540TA-002 thru 6825540TA-041	6825540	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6825540TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cage, Compressor Ext Shaft Bearing	6841960TA 6841960TA-001	6841960	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6841960TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Pin, Anti- Rotation - Front Turbine Bearing	6844648TA 6844648TA-005 6844648TA-010	6844648	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6844648TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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Damper, Turbine Front Bearing Vibration	6844649TA	6844649	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6844649TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Seal Ring, Split - 5.310 OD X .168	6895818TA	6895818	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6895818TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Tube Assembly, Turbine Rear Bearing Oil - .500-20 Thread	6897481TA	6897481	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6897481TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Coupling, Clamp-V Type, Comb Chamber Crossover Tube	6898223TA	6898223	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6898223TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22A, D22C, D22G
Spring-Red, Gear Oil Pressure Relief Valve	6708521TA	6708521	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6708521TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Stud-3/8-24 X 7/16-14 Thread X 1- 15/16 Long (& oversizes)	6709216TA 6709216TA-001 6709216TA-003 6709216TA-006 6709216TA-009 6709216TA-012	6709216	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6709216TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Lock-Oil Pressure Relief Valve Plug	6736213TA	6736213	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6736213TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Tube Assembly - Pinion Gear Oil Jet	6739488TA 6739488TA-001 6739488TA-002 6739488TA-003	6739488	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6739488TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Lock-Planet Nut, Splined	6788398TA	6788398	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6788398TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Lock, Propellor Shaft Thrust Nut	6788713TA	6788713	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6788713TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear, Spur- Nose Scavenge Pump Drive	6829388TA	6829388	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6829388TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear, Spur- Scavenge Pump Drive	6829398TA	6829398	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6829398TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Gear, Sun	6841222TA	6841222	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6841222TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Flange, Starter Shaft Bearing	6887148TA	6887148	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6887148TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing-Oil Pump Idler Gear	41642TA 41642TA-006	41642	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 41642TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Support Assembly Turbine-Rear Bearing	23005861TA	23005861	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23005861TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Elbow, Tube- 3 Way Flared Tube & Bulkhead	23005930TA	23005930	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23005930TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Support Assembly Turbine-Rear Bearing	23005961TA 23005961TA- 001 thru 23005961TA- 007 23005961TA- 009 thru 23005961TA- 013 23005961TA- 201 thru 23005961TA- 208	23005961	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23005961TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22
Facing, Propeller - Brake	23031797TA 23031797TA- 001 23031797TA- 002	23031797	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031797TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Ring, Repair - Starter Pad ID Bore	23032867TA	23032867	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23032867TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22
Facing, Propeller - Brake	23039657TA 23039657TA- 001 23039657TA- 002	23039657	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23039657TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Seal Assembly Housing Accessory	23052098TA 23052098TA- 001 23052098TA- 002 23052098TA- 003	23052098	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23052098TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Insert - Drain Plug (and oversizes)	23056964TA 23056964TA- 003 23056964TA- 006	23056964	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23056964TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear - Variable Resistor	6824269TA 6824269TA-001 6824269TA-002	6824269	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6824269TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22
Arm - Propellor Control, Rear	6826935TA	6826935	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6826935TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Tube Assembly, Sun Gear Lube	6828461TA 6828461TA-001 6828461TA-002 6828461TA-003	6828461	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6828461TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Seal, Labyrinth - Turbine Front Bearing	6841158TA	6841158	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6841158TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D22
Gear, Spur – Starter	6845748TA 6845748TA-043 6845748TA-044	6845748	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6845748TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Seal Assembly, Labyrinth - Front Compressor	6846208TA 6846208TA-001 thru 6846208TA-004	6846208	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6846208TA <u>Rev:</u> I/R <u>Date:</u> 12/29/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft Gear - Accessory Main Drive	6731183TA	6731183	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6731183TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Support Assembly, Turbine- Scavenge Oil Pump	6785234TA	6785234	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6785234TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Support Assembly - Turbine Scavenge Oil Pump	6783985TA 6783985TA-001 6783985TA-002 6783985TA-003 6783985TA-004 6783985TA-005 6783985TA-006	6783985	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6783985TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Tube - Bearing Lock	6792068TA 6792068TA-001 6792068TA-002 6792068TA-003	6792068	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6792068TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cone Assembly - Inner Front Exhaust	6824478TA 6824478TA-001 6824478TA-002 6824478TA-003 6824478TA-004 6824478TA-005	6824478	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6824478TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Stud, Plain- 5/16-24 X 5/16-24 X 1.031	6825525TA 6825525TA-005 6825525TA-030 6825525TA-045 6825525TA-060 6825525TA-075	6825525	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6825525TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Key, Locking - Rear Turbine Bearing	6850367TA	6850367	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6850367TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Retainer, Bearing-Rear Turbine	6855657TA	6855657	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6855657TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Nozzle, Oil - .070 Dia. Orifice	6870701TA	6870701	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6870701TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Gear Assembly - Aux Pressure Oil Pump Idler	6716336TA	6716336	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6716336TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear - Aux Pressure Oil Pump Idler	6716337TA	6716337	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6716337TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing - Aux Pressure Oil Pump Idler Gear	6716338TA	6716338	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6716338TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft-Oil Pump Idler	6785512TA	6785512	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6785512TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft Gear - Oil Pump Drive	6789695TA	6789695	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6789695TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Cage - Nose Bearing (& oversizes)	6809074TA 6809074TA-001 6809074TA-002 6809074TA-004 6809074TA-006 6809074TA-010 6809074TA-015 6809074TA-020	6809074	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6809074TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Pin - Straight, Headless .094 Dia X .250 Long and oversizes	6823374TA 6823374TA-001 6823374TA-002 6823374TA-003 6823374TA-004	6823374	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6823374TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Housing - Coupling	6823502TA	6823502	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6823502TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt Assembly, Turbine To Compressor – Tie	6844778TA 6844778TA-001	6844778	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6844778TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Bolt, Tie - Compressor	6845945TA	6845945	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6845945TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft, Drive - Oil Pump	6846749TA	6846749	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6846749TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear, Spur Oil Pump Idler	6850824TA	6850824	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6850824TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Gear, Spur Oil Pump Drive	6850825TA	6850825	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6850825TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft, Idler Gear – Oil Pump	6850826TA	6850826	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6850826TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cage, Bearing - Front Pinion	6871866TA 6871866-001 6871866-002 6871866-003 6871866-004 6871866-005 6871866-006 6871866-007	6871866	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6871866TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cage, Bearing - 230mm X 28mm, Wide Flanged	6873728TA 6873728TA-001 6873728TA-002 6873728TA-003 6873728TA-004 6873728TA-005	6873728	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6873728TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft Gear, Spur - Pump Drive	6876478TA	6876478	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6876478TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Nut, Thrust Bearing Prop Shaft	23033549TA	23033549	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23033549TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear - Scavenge Pump Drive	6723179TA	6723179	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6723179TA <u>Rev:</u> I/R <u>Date:</u> 12/04/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Bushing, .556 ID X .6895 OD X .375 Long	6723227TA	6723227	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6723227TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft, Nose Pump Idler	6787878TA	6787878	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6787878TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing, .366 ID X .502 OD X .365 Long	6871659TA	6871659	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6871659TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Grommet Assembly - Thermo- Couple	6877738TA 6877738TA-001 6877738TA-002 6877738TA-003	6877738	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6877738TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C
Insert-Drain Plug (& Oversizes)	6723206TA 6723206TA-003 6723206TA-006	6723206	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6723206TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing - Delivery Flange	6782914TA	6782914	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6782914TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing - .491 ID X .752 OD X .625 Long	6784642TA	6784642	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6784642TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Nut - 2 1/8-16 R.H. Splined	6784692TA	6784692	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6784692TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Bushing, .5625-18 X .750-16 X .594 Long - Flanged	6784729TA	6784729	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6784729TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Insert-Screw Thread 3/4-16 X 15/16 -20 X .750	6784960TA	6784960	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6784960TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Rod-Reverse Torque Switch Actuator	6785177TA	6785177	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6785177TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Block-Carrier Spacer	6786580TA	6786580	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6786580TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear-Scav Pump Drive (Pinion)	6788225TA	6788225	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6788225TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Nut-Prop Shaft Thrust Bearing	6788469TA	6788469	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6788469TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Screen Assembly - Scavenge Oil	6789772TA 6789772TA-001 6789772TA-002 6789772TA-003	6789772	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6789772TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Plug- Extension Shaft Coupling	6792069TA	6792069	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6792069TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Nozzle Assembly - Oil Jet	6793640TA 6793640TA-001	6793640	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6793640TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Link Rod Assembly - Prop Control, Inter	6794188TA 6794188TA-001 6794188TA-002 6794188TA-003	6794188	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6794188TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Shaft - Coupling Outer Member	6794399TA 6794399TA-001 6794399TA-002 6794399TA-003	6794399	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6794399TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Ring- Retaining (Internal)	6813098TA 6813098TA-050 6813098TA-051 6813098TA-056 6813098TA-062 6813098TA-068 6813098TA-075 6813098TA-077 6813098TA-081 6813098TA-086 6813098TA-087 6813098TA-090 6813098TA-093 6813098TA-100 6813098TA-102 6813098TA-106 6813098TA-112 6813098TA-118 6813098TA-125 6813098TA-131 6813098TA-137 6813098TA-143 6813098TA-145 6813098TA-150 6813098TA-156 6813098TA-162 6813098TA-165 6813098TA-168 6813098TA-175 6813098TA-181 6813098TA-185	6813098	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6813098TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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	6813098TA-187				
	6813098TA-193				
	6813098TA-200				
	6813098TA-206				
	6813098TA-212				
	6813098TA-218				
	6813098TA-225				
	6813098TA-231				
	6813098TA-237				
	6813098TA-244				
	6813098TA-250				
	6813098TA-253				
	6813098TA-256				
	6813098TA-262				
	6813098TA-268				
	6813098TA-275				
	6813098TA-281				
	6813098TA-283				
	6813098TA-287				
	6813098TA-300				
	6813098TA-306				
	6813098TA-312				
	6813098TA-315				
	6813098TA-325				
	6813098TA-334				
	6813098TA-347				
	6813098TA-350				
	6813098TA-356				
	6813098TA-362				
	6813098TA-375				
	6813098TA-387				
	6813098TA-393				
	6813098TA-400				
	6813098TA-412				
	6813098TA-425				
	6813098TA-433				
	6813098TA-450				
	6813098TA-462				
	6813098TA-475				
	6813098TA-500				
	6813098TA-525				
	6813098TA-537				
	6813098TA-550				
	6813098TA-575				
	6813098TA-600				
	6813098TA-625				
	6813098TA-650				
	6813098TA-662				

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	6813098TA-675				
	6813098TA-700				
	6813098TA-725				
	6813098TA-750				
	6813098TA-775				
	6813098TA-800				
	6813098TA-825				
	6813098TA-850				
	6813098TA-875				
	6813098TA-900				
	6813098TA-925				
	6813098TA-950				
	6813098TA-975				
	6813098TA-1000				
	6813098TA-1050				
Gasket-Spark Igniter Mounting	6816047TA	6816047	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6816047TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt And Bearing Assembly, Prop Control Shaft Pivot	6819699TA	6819699	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6819699TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Nut-Spanner, 2 1/8-16 Thread Special	6823308TA	6823308	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6823308TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Stud-Stepped 1/4-28 Thread X 5/16-18 Thread X 1.120 Long	6823309TA 6823309TA-001 6823309TA-003 6823309TA-006 6823309TA-009 6823309TA-012	6823309	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6823309TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A

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Nut-Thrust Bearing Prop Shaft	6847254TA	6847254	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6847254TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Cover Assembly, Oil Pump	6850820TA 6850820TA-001	6850820	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6850820TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Spring, Helical Compr .434 OD X .916 Long, 4.860 Rate	6858282TA	6858282	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6858282TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing, Sleeve - .491 ID X .637 OD X .375 Long	6871625TA	6871625	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6871625TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing, Sleeve - Stepped .49 ID X .70 OD X .67 Long	6851626TA	6851626	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6851626TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft, Valve Reduction Gear Main Oil Pump Pressure Adjusting	23005923TA	23005923	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23005923TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cover Assembly, Pump- Reduction Gearbox Main Oil	23006662TA	23006662	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23006662TA <u>Rev:</u> I/R <u>Date:</u> 01/12/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Vane Assembly, Compressor, Air Inlet, Anti- Icing	6873624TA 6873624TA-001 6873624TA-002 6873624TA-003 6873624TA-004 6873624TA-005	6873624	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6873624TA <u>Rev:</u> I/R <u>Date:</u> 09/12/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane, Compressor, Air Inlet, Anti- Icing	6873625TA 6873625TA-001 6873625TA-002	6873625	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6873625TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Blade, Turbine, 1st Stage	6898061TA	6898061	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6898061TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Blade, Turbine, 2nd Stage	6898062TA	6898062	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6898062TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Blade, Turbine, 3rd Stage	6875243TA	6875243	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6875243TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Blade, Turbine, 4th Stage	6852412TA	6852412	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6852412TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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Vane Segment, Turbine, 2nd Stage	23031741TA	23031741	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23031741TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Vane Segment, Turbine, 1st Stage	23031745TA 23031745TA-001 23031745TA-002 23031745TA-003 23031745TA-004	23031745	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23031745TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Vane Segment, Turbine, 1st Stage	23031746TA 23031746TA-001 23031746TA-002 23031746TA-003 23031746TA-004	23031746	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23031746TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Vane Segment, Turbine, 1st Stage	23031747TA 23031747TA-001 23031747TA-002 23031747TA-003 23031747TA-004	23031747	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23031747TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Vane Segment, Turbine, 1st Stage	23031748TA 23031748TA-001 23031748TA-002 23031748TA-003 23031748TA-004	23031748	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23031748TA <u>Rev:</u> I/R <u>Date:</u> 09/13/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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Vane Segment, Turbine, 3rd Stage	6846994TA	6846994	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6846994TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Vane Segment, Turbine, 4th Stage	6846995TA	6846995	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6846995TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Plug, Oil, Seal, 3/8 Dia X 5/16 Long (and Oversizes)	18326TA 18326TA-005 18326TA-010	18326	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 18326TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing, Screw, .375-24 X .5625-18 (and oversizes)	6702703TA 6702703TA-003 6702703TA-006	6702703	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6702703TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Plug, 1.314 OD X .312 Long (and oversizes)	6705924TA 6705924TA-005 6705924TA-010 6705924TA-015 6705924TA-020	6705924	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6705924TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cage, Rear Compressor Bearing, (and oversizes)	6723753TA 6723753TA-002 6723753TA-004 6723753TA-006	6723753	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6723753TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C
Ring, Retaining, Internal	6726656TA 6726656TA-050 6726656TA-051 6726656TA-053 6726656TA-056 6726656TA-059 6726656TA-062 6726656TA-065 6726656TA-068 6726656TA-071 6726656TA-075	6726656	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6726656TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
	6726656TA-077				
	6726656TA-078				
	6726656TA-081				
	6726656TA-084				
	6726656TA-086				
	6726656TA-087				
	6726656TA-090				
	6726656TA-093				
	6726656TA-096				
	6726656TA-098				
	6726656TA-100				
	6726656TA-102				
	6726656TA-103				
	6726656TA-106				
	6726656TA-109				
	6726656TA-112				
	6726656TA-115				
	6726656TA-118				
	6726656TA-121				
	6726656TA-125				
	6726656TA-128				
	6726656TA-131				
	6726656TA-134				
	6726656TA-137				
	6726656TA-143				
	6726656TA-145				
	6726656TA-146				
	6726656TA-150				
	6726656TA-156				
	6726656TA-162				
	6726656TA-165				
	6726656TA-168				
	6726656TA-175				
	6726656TA-181				
	6726656TA-185				
	6726656TA-187				
	6726656TA-193				
	6726656TA-200				
	6726656TA-204				
	6726656TA-206				
	6726656TA-212				
	6726656TA-216				
	6726656TA-218				
	6726656TA-225				
	6726656TA-231				
	6726656TA-237				
	6726656TA-243				
	6726656TA-244				

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	6726656TA-250				
	6726656TA-253				
	6726656TA-256				
	6726656TA-262				
	6726656TA-267				
	6726656TA-268				
	6726656TA-275				
	6726656TA-281				
	6726656TA-283				
	6726656TA-287				
	6726656TA-293				
	6726656TA-295				
	6726656TA-300				
	6726656TA-306				
	6726656TA-312				
	6726656TA-314				
	6726656TA-318				
	6726656TA-325				
	6726656TA-331				
	6726656TA-334				
	6726656TA-337				
	6726656TA-343				
	6726656TA-350				
	6726656TA-354				
	6726656TA-356				
	6726656TA-362				
	6726656TA-368				
	6726656TA-374				
	6726656TA-375				
	6726656TA-381				
	6726656TA-387				
	6726656TA-393				
	6726656TA-400				
	6726656TA-406				
	6726656TA-412				
	6726656TA-418				
	6726656TA-425				
	6726656TA-431				
	6726656TA-433				
	6726656TA-437				
	6726656TA-443				
	6726656TA-450				
	6726656TA-452				
	6726656TA-456				
	6726656TA-462				
	6726656TA-468				
	6726656TA-472				
	6726656TA-475				

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	6726656TA-481				
	6726656TA-487				
	6726656TA-492				
	6726656TA-493				
	6726656TA-500				
	6726656TA-511				
	6726656TA-512				
	6726656TA-525				
	6726656TA-537				
	6726656TA-550				
	6726656TA-551				
	6726656TA-575				
	6726656TA-587				
	6726656TA-590				
	6726656TA-600				
	6726656TA-612				
	6726656TA-625				
	6726656TA-629				
	6726656TA-637				
	6726656TA-650				
	6726656TA-662				
	6726656TA-669				
	6726656TA-675				
	6726656TA-687				
	6726656TA-700				
	6726656TA-708				
	6726656TA-712				
	6726656TA-725				
	6726656TA-737				
	6726656TA-750				
	6726656TA-762				
	6726656TA-775				
	6726656TA-787				
	6726656TA-800				
	6726656TA-825				
	6726656TA-826				
	6726656TA-846				
	6726656TA-850				
	6726656TA-875				
	6726656TA-885				
	6726656TA-900				
	6726656TA-905				
	6726656TA-925				
	6726656TA-944				
	6726656TA-950				
	6726656TA-975				
	6726656TA- 1000				

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	6726656TA- 1025 6726656TA- 1050 6726656TA- 1075 6726656TA- 1100				
Ring, Snap, 2- 7/16 OD X 1/16 Thick, External	6738053TA	6738053	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6738053TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Spring, Safety, Coupling	6738054TA	6738054	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6738054TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Ring, Internal, Snap, 3-3/4 Bore Dia.	6738077TA	6738077	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6738077TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Ring, Safety Coupling, Spherical, External	6738451TA	6738451	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6738451TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cage, Front Compressor Bearing	6787535TA 6787535TA-002 6787535TA-004 6787535TA-006	6787535	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6787535TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seal, Labyrinth, Rotating, Compressor, Rear, Primary	6793048TA	6793048	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6793048TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A

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Seal, Labyrinth, Rotating, Compressor, Rear, Primary	6794722TA	6794722	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6794722TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Casing Turbine Vane	23009372TA	23009372	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23009372TA <u>Rev:</u> I/R <u>Date:</u> 11/17/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Sleeve - Diffuser Comb, Inner Casing	6783834TA 6783834TA-010 6783834TA-020	6783834	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6783834TA <u>Rev:</u> I/R <u>Date:</u> 09/25/2006, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Fitting - Spray Shield	6785425TA	6785425	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6785425TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing - Plain, Self Aligning .375 ID X .812 OD X .406 Wide	6815460TA	6815460	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6815460TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Nut - Self Locking Hex Reverse Entry Thin 450°F	6816370TA 6816370TA-428 6816370TA-524 6816370TA-624	6816370	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6816370TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Nut - Self Locking Hex Reverse Entry Thin 450°F, L.H. Thread	6816376TA 6816376TA-428 6816376TA-524 6816376TA-624	6816376	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6816376TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Clevis - Bell Crank, Prop Control, Intermediate	6819698TA 6819698TA-001 6819698TA-002	6819698	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6819698TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A

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Plug, Magnetic - Machine Thread	6822092TA	6822092	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6822092TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Plug, Magnetic - Machine Thread	6822222TA	6822222	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6822222TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Ring - Retaining, Internal Spl	6823108TA	6823108	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6823108TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Nut - Hex 5/16-24 Spl	6823125TA	6823125	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6823125TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Nut - Hex Head Drilled, 1/4-28 Thd	6823306TA	6823306	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6823306TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Pin - Stepped, .187 OD X .380 Long, Drilled	6823334TA	6823334	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6823334TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Washer, Key .250 ID X .031 Thick, Tab Locking	6855504TA	6855504	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6855504TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt, Machine - Hex Head, .312-24 AMS 5643, Full Shank	6871529TA 6871529TA-012 6871529TA-014 6871529TA-016	6871529	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6871529TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Pump Assembly, Scavenge Oil - Main	6871879TA	6871879	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6871879TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Liner Assembly, Transition	6876865TA	6876865	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6876865TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C
Tube, Turbine Rear Bearing Oil	6891552TA	6891552	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6891552TA <u>Rev:</u> I/R <u>Date:</u> 01/17/2007, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Flange - Tube, Rear Turbine Bearing Oil	6892292TA	6892292	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6892292TA <u>Rev:</u> I/R <u>Date:</u> 10/05/2006, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Member, Safety Coupling – Intermediate	T6842119	6842119	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6842119 <u>Rev:</u> I/R <u>Date:</u> 08/17/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Member, Safety Coupling - Inner	T6842120	6842120	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6842120 <u>Rev:</u> I/R <u>Date:</u> 08/17/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gearshaft, Bevel Turbine Scavenge Oil Pump Drive, Rear	T6822347	6822347	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6822347 <u>Rev:</u> I/R <u>Date:</u> 08/10/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear-Spur, Tachometer & Oil Pump Drive	T6824043	6824043	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6824043 <u>Rev:</u> I/R <u>Date:</u> 08/10/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Shaft Assembly, Drive Tachometer	T6846747	6846747	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6846747 <u>Rev:</u> I/R <u>Date:</u> 08/10/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Ferrule, Fuel Nozzle	T6846518	6846518	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6846518 <u>Rev:</u> I/R <u>Date:</u> 08/10/2000, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Tube, Crossover	T6857509	6857509	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6857509 <u>Rev:</u> I/R <u>Date:</u> 08/11/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Mount Assembly, Igniter Ferrule	T6829556	6829556	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829556 <u>Rev:</u> I/R <u>Date:</u> 08/10/2000, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Seal Ring, Split - 4.450 OD X .135 Thick	T6895817	6895817	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6895817 <u>Rev:</u> I/R <u>Date:</u> 09/22/2000, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Ring, Air Seal - 3rd Stage Vane	T6892263	6892263	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6892263 <u>Rev:</u> I/R <u>Date:</u> 09/22/2000, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Stud-Stepped, 2 Diameter Engagement, 3/8-16 NJS X 5/16-24 UNJF	T6841729	6841729	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6841729 <u>Rev:</u> I/R <u>Date:</u> 09/20/2000, or later FAA approved revisions	Rolls-Royce	501-D22C, D22G
Washer, Check Valve	T6812646	6812646	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6812646 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Spacer, Combustion Chamber To Crossover Tube	T6898224	6898224	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6898224 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Packing, Preformed O'Ring - Viton	T6877242	6877242	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6877242 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Pin, Shouldered, Headless- .311 Dia X .332 Long CRES	T6847913	6847913	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6847913 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Washer-Thrust Teflon .835 ID X 1.135 OD X .015 Thick	T6827183	6827183	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6827183 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Nut, Turbine Rear Bearing Clamp	T6785084	6785084	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6785084 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Ferrule, Crossover	T6899778	6899778	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6899778 <u>Rev:</u> I/R <u>Date:</u> 09/22/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seal, Labyrinth- Turbine Rear	T6897646	6897646	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6897646 <u>Rev:</u> I/R <u>Date:</u> 08/10/2000, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Seal, Labyrinth- Turbine Front Bearing, Front	T6847330	6847330	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6847330 <u>Rev:</u> I/R <u>Date:</u> 09/20/2000, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G

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Coupling, Compressor Shaft	T6843873	6843873	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6843873 <u>Rev:</u> I/R <u>Date:</u> 09/21/2000, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Gear, Spur- Internal Spline	T23001425	23001425	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T23001425 <u>Rev:</u> I/R <u>Date:</u> 09/20/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Support Liner- Two Piece Igniter	T6843949	6843949	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6843949 <u>Rev:</u> I/R <u>Date:</u> 09/22/2000, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Support Liner- Two Piece Igniter	T6824805	6824805	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6824805 <u>Rev:</u> I/R <u>Date:</u> 09/22/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22
Gear-Spur, Pump Drive Idler	T6824044	6824044	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6824044 <u>Rev:</u> I/R <u>Date:</u> 09/22/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 2nd Thru 5th Stage Compressor Master	T6714792	6714792	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714792 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 6th Thru 11th Stage Compressor Master	T6714793	6714793	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714793 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 12th & 13th Stage Compressor Master	T6714794	6714794	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714794 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Vane 2nd Stage Compressor	T6714802	6714802	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714802 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 3rd Stage Compressor	T6714803	6714803	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714803 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 4th Stage Compressor	T6714804	6714804	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714804 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 5th Stage Compressor	T6714805	6714805	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714805 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 6th Stage Compressor	T6714806	6714806	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714806 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 7th Stage Compressor	T6714807	6714807	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714807 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 8th Stage Compressor	T6714808	6714808	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714808 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 9th Stage Compressor	T6714809	6714809	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714809 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Vane 10th Stage Compressor	T6714810	6714810	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714810 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 11th Stage Compressor	T6714811	6714811	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714811 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 12th Stage Compressor	T6714812	6714812	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714812 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 13th Stage Compressor	T6714813	6714813	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714813 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane 14th Stage Compressor	T6714814	6714814	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714814 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Compressor Outlet	T6714815	6714815	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6714815 <u>Rev:</u> I/R <u>Date:</u> 09/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor - 1st Stage	T6859601	6859601	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6859601 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Ring Retaining (Internal)	T6808649	6808649	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6808649 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Rod-Control 5/16-24 X 3.163 Long	T6816326	6816326	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6816326 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Eyebolt, Reduction Gear	T6824774	6824774	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6824774 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing Ball, Rod End, .375-24 X .250 Eye	T6828147	6828147	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6828147 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Member, Safety Coupling- Outer	T6829670	6829670	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829670 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Coupling Assembly, Safety	T6841049	6841049	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6841049 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Seat, Spring- Safety Coupling	T6841117	6841117	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6841117 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Ring, 3.055 ID X 3.625 OD X .219 Wide	T6848143	6848143	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6848143 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Stud, Shouldered, Straight- .250- 28 X .250-28 X 2.906 Long	T6854917	6854917	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6854917 <u>Rev:</u> I/R <u>Date:</u> 09/22/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Seat, Spring Safety Coupling	T6876196	6876196	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6876196 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Spacer, Safety Coupling Seat	T6876198	6876198	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6876198 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seal - 2.000 ID X .076 X .168	T6877243	6877243	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6877243 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Coupling Assembly, Safety	T6889382	6889382	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6889382 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Seal, Propeller Shaft	T23034555	23034555	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T23034555 <u>Rev:</u> I/R <u>Date:</u> 05/15/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Roller Planet Gear & Bearing Spherical	T6789777	6789777	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6789777 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Mount Assembly, Igniter Ferrule	T6794335	6794335	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6794335 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13
Gasket- Turbine Inlet Thermocouple	T6805748	6805748	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6805748 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Plug - Straight 7/16-20 Thread	T6823415	6823415	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6823415 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Gear, Spur - Main Accessory Drive	T6829396	6829396	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829396 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear, Planet	T6841219	6841219	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6841219 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Journal, Planet Gear And Bearing	T6843135	6843135	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6843135 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Seal Assembly, Propeller Shaft	T6875798	6875798	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6875798 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear & Bearing Assembly, Planet	T6894129	6894129	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6894129 <u>Rev:</u> I/R <u>Date:</u> 07/06/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Vane Assembly, Compressor 2nd Stage	T6859602	6859602	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859602 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 3rd Stage	T6859603	6859603	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859603 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Vane Assembly, Compressor 4th Stage	T6859604	6859604	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859604 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 5th Stage	T6859605	6859605	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859605 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 6th Stage	T6859606	6859606	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859606 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 7th Stage	T6859607	6859607	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859607 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 8th Stage	T6859608	6859608	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859608 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 9th Stage	T6859609	6859609	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859609 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 10th Stage	T6859610	6859610	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859610 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 11th Stage	T6859611	6859611	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859611 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Vane Assembly, Compressor 12th Stage	T6859612	6859612	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859612 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 13th Stage	T6859613	6859613	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859613 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Vane Assembly, Compressor 14th Stage Comp & Outlet	T6789474	6789474	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6789474 <u>Rev:</u> I/R <u>Date:</u> 09/18/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cupwasher - Spanner Nut Lock, .375 Dia.	T6850984	6850984	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6850984 <u>Rev:</u> I/R <u>Date:</u> 01/09/2001, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Washer, Cup - 2.060 OD X .430 Long	T6844819	6844819	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6844819 <u>Rev:</u> I/R <u>Date:</u> 01/09/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seat, Check Valve	T6859371	6859371	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6859371 <u>Rev:</u> I/R <u>Date:</u> 01/09/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt, Machine - Hex, .250-28 X .469 Long, AMS 5643	T6871530	6871530	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6871530 <u>Rev:</u> I/R <u>Date:</u> 11/08/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seal, Labyrinth - Rotor, Inner	T6845277	6845277	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6845277 <u>Rev:</u> I/R <u>Date:</u> 01/04/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Ring - Retaining, External	T6735310	6735310	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6735310 <u>Rev:</u> I/R <u>Date:</u> 10/17/2000, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Tube, Turbine Vent	T6846968	6846968	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6846968 <u>Rev:</u> I/R <u>Date:</u> 01/04/2001, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Nut - Spanner, Slotted Periphery, 2 7/8-12 Thread	T6784981	6784981	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6784981 <u>Rev:</u> I/R <u>Date:</u> 01/09/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seal Assembly, Rear Turbine Bearing Oil	T23004669	23004669	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T23004669 <u>Rev:</u> I/R <u>Date:</u> 01/04/2001, or later FAA approved revisions	Rolls-Royce	501-D22A, D22C, D22G
Seal, Labyrinth - Rotor, Outer, Compressor Front Bearing	T6844056	6844056	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6844056 <u>Rev:</u> I/R <u>Date:</u> 01/04/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bearing Ball, Thrust Swashplate	204-011-430-1T	204-011-430-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 204-011-430-1T <u>Rev:</u> I/R <u>Date:</u> 04/21/2003, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 205B 212 412 412EP 412CF
Fitting 3 Way .8750-14 X .8750-14 X .7500-16	T23009984	23009984	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T23009984 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A

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Bearing, Roller 75mm X 130mm X 27mm	T23055590	23055590	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> T23055590 <u>Rev:</u> I/R <u>Date:</u> 11/05/2002, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22C, D22G
Hinge- Structural Extruded	T6783463	6783463	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6783463 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Hinge- Structural Extruded	T6783465	6783465	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6783465 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Hinge- Structural Extruded	T6783466	6783466	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6783466 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Hinge- Structural Extruded	T6783467	6783467	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6783467 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Coupling Turbine Rear Scavenge Pump Drive	T6789877	6789877	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6789877 <u>Rev:</u> I/R <u>Date:</u> 07/30/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Stud, Shouldered & Stepped .190- 32 X .250-28 X .909 Long	T6843305	6843305	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6843305 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Piston, Pressure Relief-Oil	T6843442	6843442	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6843442 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Plug Assembly, Vent	T6843533	6843533	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6843533 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cone Assembly, Inner Rear Exhaust	T6845753	6845753	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6845753 <u>Rev:</u> I/R <u>Date:</u> 07/28/2003, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Key, Turbine Casing Piloting-Front	T6859014	6859014	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6859014 <u>Rev:</u> I/R <u>Date:</u> 06/19/2003, or later FAA approved revisions	Rolls-Royce	501D22, D22A
Cam - Control, Beta Backup Valve	T541859	541859	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T541859 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Hamilton Sundstrand	54H60-19, -117
Pin - Shoulder, Headless	T546440	546440	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T546440 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Hamilton Sundstrand	54H60-19, -117
Cam And Pin - Cont	T546442	546442	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T546442 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Hamilton Sundstrand	54H60-19, -117
Bushing .250 ID X .375 OD X .281 Long	T6716838	6716838	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6716838 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear Assembly - Oil Pump Idler	T6781680	6781680	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> T6781680 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Facing - Propeller Brake	T6789589	6789589	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6789589 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft - Prop Control, Forward	T6827282	6827282	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6827282 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Gear, Spur - Scavenge Pump Drive	T6829397	6829397	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829397 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D22, D22A
Nut, Thrust Bearing - Prop Shaft	T6841119	6841119	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6841119 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A
Shaft, Pressure Adjusting Valve	T6841983	6841983	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6841983 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Body, Oil Pump - Scavenge. External	T6846568	6846568	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6846568 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Insulation Blanket, Thermal Front Turbine Bearing Support	T6846933	6846933	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6846933 <u>Rev:</u> I/R <u>Date:</u> 01/14/2002, or later FAA approved revisions	Rolls-Royce	501-D22, D22A, D22G
Cage, Bearing- Hydraulic Pump Idler Gear (Front And oversize)	T23004646	23004646	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T23004646 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Ring-Snap 1 1/2 OD X .054 Thick	T6721451	6721451	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6721451 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing- Screw-.3125 X 500-20 (& oversizes)	T6721919	6721919	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6721919 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing- Carrier Delivery Tube	T6723248	6723248	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6723248 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bolt-Mach, Hex Head 5/16-24 X .744 Lg (SPL)	T6823140	6823140	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6823140 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Shaft Assembly- Alternator Drive	T6825238	6825238	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6825238 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Coupling Assembly, Intermediate	T6828457	6828457	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6828457 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Cage, Bearing- Hydraulic Pump Drive	T6829092	6829092	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829092 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Gear, Spur- Main Idler	T6829385	6829385	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829385 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G

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Plug, Planet Journal	T6829473	6829473	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829473 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Pin, Reduction Gear, Eye Bolt	T6829896	6829896	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6829896 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing, NTS Actuator Rod	T6843526	6843526	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6843526 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Seal, Propeller Shaft	T6851491	6851491	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6851491 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Insert, Oil Transfer	T6873358	6873358	Identity per 14 CFR §21.303 <u>Dwg No.:</u> T6873358 <u>Rev:</u> I/R <u>Date:</u> 07/20/2001, or later FAA approved revisions	Rolls-Royce	501-D13, D22, D22A, D22C, D22G
Bushing	ASI-206-010- 338-1	206-010-338-1	Identity per 14 CFR §21.303 <u>Dwg No:</u> ASI-6010-338 <u>Rev:</u> N/C <u>Date:</u> 11/05/1991, or later FAA approved revisions	Bell Helicopter Textron	206A 206B 206L-1 206L-3
Bushing	ASI-206-010- 338-3	206-010-338-3	Identity per 14 CFR §21.303 <u>Dwg No:</u> ASI-6010-338 <u>Rev:</u> N/C <u>Date:</u> 11/05/1991, or later FAA approved revisions	Bell Helicopter Textron	206A 206B 206L-1 206L-3
Ring Assembly Swashplate Gimbal	ASI-4010-404-1	204-010-404-1	Identity per 14 CFR §21.303 <u>Dwg No:</u> ASI-4010-404 <u>Rev:</u> I/R <u>Date:</u> 07/13/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 212 412

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Nut Retention Strap, Main Rotor	ASI-4011-116-1	204-011-116-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-116 <u>Rev:</u> I/R <u>Date:</u> 10/10/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A, 205A-1 212
Pitch Horn Assembly, Main Rotor	ASI-4011-120-5	204-011-120-5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-120 <u>Rev:</u> I/R <u>Date:</u> 05/14/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212
Grip Assembly, Main Rotor	ASI-4011-121-5	204-011-121-5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-121 <u>Rev:</u> A <u>Date:</u> 08/01/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1
Main Rotor Grip	ASI-4011-121-9	204-011-121-9	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-121 <u>Rev:</u> I/R <u>Date:</u> 03/24/1992, or later FAA approved revisions	Bell Helicopter Textron	212
Main Rotor Grip	ASI-4011-121-113	204-011-121-113	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-121 <u>Rev:</u> A <u>Date:</u> 03/24/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1
Bushing, Pillow Block	ASI-4011-135-003	204-011-135-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-135 <u>Rev:</u> N/C <u>Date:</u> 07/09/1993, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212
Clevis, Pitch Change, Main Rotor	ASI-4011-136-103	204-011-136-103	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-136 <u>Rev:</u> A <u>Date:</u> 02/05/1993, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212

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Bolt – Blade, Main Rotor Assembly	ASI-4011-151-1	204-011-151-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-151 <u>Rev:</u> A <u>Date:</u> 10/22/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212
Bolt, Pillow Block	ASI-4011-171-3	204-011-171-3	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-171 <u>Rev:</u> I/R <u>Date:</u> 01/06/1993, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212
Fitting, Drag Brace Assembly	ASI-4011-180-1	204-011-180-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-180 <u>Rev:</u> I/R <u>Date:</u> 05/14/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1
Ring, Inner Swashplate	ASI-4011-402- 127	204-011-402-127	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-402 <u>Rev:</u> I/R <u>Date:</u> 07/13/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212
Ring Assembly, ASI-4011-403-1 Outer Swashplate	ASI-4011-403-1	204-011-403-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-403 <u>Rev:</u> I/R <u>Date:</u> 07/13/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212
Support Assembly, Swashplate	ASI-4011-404- 125	204-011-404-125	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-404 <u>Rev:</u> I/R <u>Date:</u> 07/10/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 212
Sleeve, Collective	ASI-4011-408- 105	204-011-408-105	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-408 <u>Rev:</u> I/R <u>Date:</u> 05/14/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212 412

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Fitting, Retention Strap, Main Rotor Assembly	ASI-4012-102-009	204-012-102-9	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4012-102 <u>Rev:</u> I/R <u>Date:</u> 10/10/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1
Fitting, Retention Strap, Outboard, Main Rotor	ASI-4012-103-001	204-012-103-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4012-103 <u>Rev:</u> I/R <u>Date:</u> 01/25/1993, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1
Fitting, Retention Strap, Outboard, Main Rotor	ASI-4012-103-005	204-012-103-5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4012-103 <u>Rev:</u> I/R <u>Date:</u> 01/25/1993, or later FAA approved revisions	Bell Helicopter Textron	212
Spacer, Tail Rotor Drive Quill	ASI-4040-313-001	204-040-313-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4040-313 <u>Rev:</u> I/R <u>Date:</u> 06/25/1993, or later FAA approved revisions	Bell Helicopter Textron	204B 205A
Clamp Assembly, Curvic Coupling, Tail Rotor Driveshaft	ASI-4040-811-1	204-040-811-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4040-811 <u>Rev:</u> I/R <u>Date:</u> 12/05/1992, or later FAA approved revisions	Bell Helicopter Textron	204B 205A 205A-1 212 412
Grip Assembly	ASI-5011-711-101	205-011-711-101	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-5011-711 <u>Rev:</u> I/R <u>Date:</u> 02/14/1992, or later FAA approved revisions	Bell Helicopter Textron	204B
Sleeve Assembly, Pivot, Swashplate	ASI-6010-454-109	206-010-454-109	Identity per 14 CFR §21.303 <u>Dwg No.:</u> ASI-6010-454 <u>Rev:</u> I/R <u>Date:</u> 04/03/1992, or later FAA approved revisions	Bell Helicopter Textron	206A 205B 206L 206L-1 206L-3 206L-4

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Idler, Tail Rotor Control, Assembly of	ASI-9011-711- 001	209-011-711-1	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> ASI-9011-711 <u>Rev:</u> I/R <u>Date:</u> 04/06/1993, or later FAA approved revisions	Bell Helicopter Textron	205A 205A-1 212 412
Plate, Retaining, Thrust Bearing	E23059551	23059551	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23059551 <u>Rev:</u> A <u>Date:</u> 08/07/2008, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R(ST), C20R/1, C20R/1(RS), C20R/2, C20R/2(SP), C20R4
Gearshaft, Spur-Gas Producer Train Idler	E23063413	23063413	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23063413 <u>Rev:</u> A <u>Date:</u> 05/27/2008, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20R, C20R(ST), C20R/1, C20R/1(RS), C20R/2, C20R/2(SP), C20S, C20W
Shroud Segment	E3020484CL1	3020484CL1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020484CL2	3020484CL2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B

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Shroud Segment	E3020484CL3	3020484CL3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020484CL4	3020484CL4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020484CL5	3020484CL5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020484CL6	3020484CL6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020484CL7	3020484CL7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020484CL8	3020484CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B

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Shroud Segment	E3020484CL9	3020484CL9	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020484CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL1	3020485CL1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL2	3020485CL2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL3	3020485CL3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL4	3020485CL4	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL5	3020485CL5	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B

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Shroud Segment	E3020485CL6	3020485CL6	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL7	3020485CL7	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL8	3020485CL8	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment	E3020485CL9	3020485CL9	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3020485CL <u>Rev:</u> N/C <u>Date:</u> 08/05/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 20, 20A, 20B
Shroud Segment, Compressor Turbine	E3043003CL12	3043003CL12	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL13	3043003CL13	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61

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Shroud Segment, Compressor Turbine	E3043003CL14	3043003CL14	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL15	3043003CL15	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL16	3043003CL16	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL17	3043003CL17	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL18	3043003CL18	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL19	3043003CL19	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61

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Shroud Segment, Compressor Turbine	E3043003CL20	3043003CL20	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL21	3043003CL21	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL22	3043003CL22	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Shroud Segment, Compressor Turbine	E3043003CL23	3043003CL23	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3043003CL <u>Rev:</u> N/C <u>Date:</u> 09/08/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 60AG, 61
Kit, Bearing, First Stage. Reduction	E3035020	3035020	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3035020 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-27, 28, 34, 34AG, 34B, 36, 66, 66B, 110, 112, 114, 114A, 121, 135, 135A
Kit, Bearing, First Stage Reduction	E3035029	3035029	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3035029 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-45A, 45B, 45R, 60A, 60AG, 65B, 65R, 65AR, 65AG, 67, 67R, 67A, 67T

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Gear, Ring, First Stage Reduction	E3037303	3037303	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037303 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-15AG, 110, 112, 121, 27, 28, 34, 34AG, 34B, 36, 38, 41, 42, 42A, 64, 66, 66B, 114, 114A, 135, 135A
Gear, Sun, First Stage Reduction	E3024765	3024765	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3024765 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-15AG, 27, 28, 34, 34AG, 34B,36
Gear Set, Sun, First Stage Reduction	E3028456	3028456	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3028456 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A, 61, 64, 66, 66B
Gear, Sun, First Stage Reduction	E3037304	3037304	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3037304 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-110, 112, 121, 114, 114A, 135, 135A
Gear Set, Planet, First Stage Reduction	E3101455-02	3101455-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3101455-02 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-15AG, 110, 112, 121, 27, 28, 34, 34AG, 34B, 36, 64, 66, 66B, 114, 114A, 135, 135A
Gear Set, Planet, First Stage Reduction	E3101525-02	3101525-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3101525-02 <u>Rev:</u> N/C <u>Date:</u> 11/16/2007, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-15AG, 27, 28, 34, 34AG, 34B, 38, 41,42, 42A

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Bearing - Roller	1-300-665- 01AL	1-300-665-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-665-01AL <u>Rev:</u> A <u>Date:</u> 06/27/2008, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Ring, Sealing Power Turbine Stator Housing	E3024979	3024979	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3024979 <u>Rev:</u> N/C <u>Date:</u> 10/24/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 27, 28, 110, 112, 121
Bearing, Ball - Duplex	1-300-658- 02AL	1-300-658-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-658-02AL <u>Rev:</u> A <u>Date:</u> 06/19/2008, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Crosshead Controls	TAI-2010-775-1	212-010-775-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2010-775 <u>Rev:</u> I/R <u>Date:</u> 01/10/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212, 412
Nut, Mast, Main Rotor	TAI-4010-481- 001	204-010-481-001	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-481 <u>Rev:</u> A <u>Date:</u> 06/20/2008, or later FAA approved revisions	Bell Helicopter Textron	212
Barrel Drag Brace Assembly	TAI-4011-143- 005	204-011-143-005	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4011-143 <u>Rev:</u> I/R <u>Date:</u> 01/09/2006, or later FAA approved revisions	Bell Helicopter Textron	212
Nut, Yoke M/R	TAI-4012-106-1	204-012-106-1	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4012-106 <u>Rev:</u> I/R <u>Date:</u> 02/26/2006, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A, 212

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Hanger-Drive	TAI-4040-617-101	204-040-617-101	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-617 <u>Rev:</u> I/R <u>Date:</u> 01/10/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212
Clamp, Curvic T/R	TAI-4040-809-3	204-040-809-3	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-809 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212
Nut	TAI-5011-710-101	205-011-710-101	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5011-710 <u>Rev:</u> I/R <u>Date:</u> 01/23/2006, or later FAA approved revisions	Bell Helicopter Textron	204B
Control Tube T/R	TAI-9011-710-5	209-011-710-5	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> TAI-9011-710 <u>Rev:</u> I/R <u>Date:</u> 04/11/2006, or later FAA approved revisions	Bell Helicopter Textron	205A, 212, 412
Bearing-Roller	1-300-012-01AL	1-300-012-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-012-01AL <u>Rev:</u> B <u>Date:</u> 08/07/2008, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Nozzle Assembly, Turbine, Second Stage	E23055438	23055428	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23055438 <u>Rev:</u> N/C <u>Date:</u> 04/09/1998, or later FAA approved revisions	Rolls-Royce	250-C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G
Liner Assembly, Combustor	E23060429	23060429	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23060429 <u>Rev:</u> N/C <u>Date:</u> 04/09/1998, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20B, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4

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Liner Assembly, Partial Combustor, Aft Section	E23061212	23061212	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23061212 <u>Rev:</u> N/C <u>Date:</u> 04/13/1998, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20B, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Cone, First Section, Combustion Liner	E23061213	23061213	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23061213 <u>Rev:</u> N/C <u>Date:</u> 04/13/1998, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20B, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Dome, Combustion Sub Assembly	E23061216	23061216	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23061216 <u>Rev:</u> N/C <u>Date:</u> 04/13/1998., or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20B, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4
Nozzle Assembly, Turbine, First Stage - IN 738	E23062028-738	23062028	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062028-738 <u>Rev:</u> N/C <u>Date:</u> 04/09/1998, or later FAA approved revisions	Rolls-Royce	250- C30G/2
Nozzle Assembly, Turbine, Second Stage	E23062029	23062029	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062029 <u>Rev:</u> N/C <u>Date:</u> 04/09/1998, or later FAA approved revisions	Rolls-Royce	250-C30G/2
Nozzle Assembly, Turbine, First Stage	E23062753	23062753	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062753 <u>Rev:</u> N/C <u>Date:</u> 02/25/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W

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Nozzle Assembly, Turbine, First Stage	E23062754	23062754	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23062754 <u>Rev:</u> N/C <u>Date:</u> 02/25/1998, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4
Ring, Retaining – Internal, 1.765 x 0.042 thick	E6859073	6859073	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E6859073 <u>Rev:</u> N/C <u>Date:</u> 11/24/1997, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C, C30, C30L, C30M, C30P, C30S, C30R, C30U, C30G, C30G/2, C30R/1, C30R/3, C40B, C47B, C47M
Bearing, Ball	1-300-663- 01AL	1-300-663-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-663-01AL <u>Rev:</u> A <u>Date:</u> 07/21/2008, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Liner Assembly Combustion	23066675TA	23066675	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23066675TA <u>Rev:</u> IR <u>Date:</u> 01/28/2009, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30S
Liner Assembly Combustion	E23066675	23066675	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E23066675 <u>Rev:</u> G <u>Date:</u> 03/16/2009, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30S

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Bearing Set, Ball	204-011-409- 1AL	204-011-409-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 204-011-409-1AL <u>Rev:</u> B <u>Date:</u> 12/11/2008, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1
Bearing Set, Ball	540-011-449- 3AL	540-011-449-3	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 540-011-449-3AL <u>Rev:</u> B <u>Date:</u> 05/13/2008, or later FAA approved revisions	Bell Helicopter Textron	212 412
Bearing Cone Upper Hub M/R	369D21254- 3TA	369D21254-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 369D21254-3TA <u>Rev:</u> I/R <u>Date:</u> 05/12/2009, or later FAA approved revisions	MD Helicopters,	369D, 369E, 369FF, 500N
Bearing Cup	369D21255- 3TA	369D21255-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 369D21255-3TA <u>Rev:</u> I/R <u>Date:</u> 05/13/2009, or later FAA approved revisions	MD Helicopters,	369D, 369E, 369FF, 500N
Bearing Cone Upper Hub M/R	369D21256- 3TA	369D21256-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 369D21256-3TA <u>Rev:</u> I/R <u>Date:</u> 05/26/2009, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF, 500N
Bearing Cup	369D21257- 3TA	369D21257-3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 369D21257-3TA <u>Rev:</u> I/R <u>Date:</u> 05/26/2009, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF, 500N
Blade, Compressor	E3012801	3012801	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3012801 <u>Rev:</u> B <u>Date:</u> 02/12/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-11, 11AG, 15AG, 21, 25, 25A, 27, 28, 34, 34B, 34AG, 36, 110, 112, 114, 121, 135

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Bearing, Roller, Flanged	3032208TA	3032208	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3032208TA <u>Rev:</u> I/R <u>Date:</u> 04/13/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 38, 40, 41, 41AG, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 64, 65B, 65R, 65AR, 65AG, 66, 66A, 66B, 66D, 67, 67A, 67AF, 67AG, 67B, 67D, 67R, 67T, 110, 112, 114, 114A, 121, 135, 135A; PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6
Bearing, Roller, Output Shaft	369D25148TA	369D25148	Test and Computations per 14 CFR §§21.303 <u>Dwg No.:</u> 369D25148TA <u>Rev:</u> I/R <u>Date:</u> 02/27/2009, or later FAA approved revisions	MDHI	369D, 369E, 369FF, 500N, 600N
Tie Rod, Compressor Rotor	E3018397	3018397	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3018397 <u>Rev:</u> A <u>Date:</u> 11/11/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 45A, 45B, 45R, 50

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Tie Rod, Compressor Rotor	E3019371	3019371	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3019371 <u>Rev:</u> A <u>Date:</u> 11/11/2008, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 35, 36, 110, 112, 114, 114A, 121, 135, 135A; PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6
Tie Rod, Compressor Rotor	E3107549-01	3107549-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3107549-01 <u>Rev:</u> A <u>Date:</u> 08/10/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 65B, 65R, 65AR, 65AG, 66, 66B, 67, 67A, 67AF, 67AG, 67R, 67T, 67B, 67D
Bearing, Roller, Flanged	3040997TA	3040997	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3040997TA <u>Rev:</u> I/R <u>Date:</u> 04/07/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6
Bearing, Ball, Annular, Duplex	3112987-01TA	3112987-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3112987-01TA <u>Rev:</u> I/R <u>Date:</u> 05/14/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6
Bearing, Roller	1-300-664- 01AL	1-300-664-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-664-01AL <u>Rev:</u> A <u>Date:</u> 07/30/2008, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1 T5317B

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Bearing, Roller, Flanged	3032648TA	3032648	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3032648TA <u>Rev:</u> I/R <u>Date:</u> 07/02/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-45A, 45B, 45R, 60A, 60AG, 65B, 65R, 65AR, 65AG, 67, 67R, 67A, 67AF, 67AG, 67B, 67D, 67T
Grip Assembly - Steel, Tail Rotor	TAI-5011-711- 101	205-011-711-101	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-5011-711 <u>Rev:</u> I/R <u>Date:</u> 08/31/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A
Slider -Tail Rotor Assembly	TAI-4010-720- 003	204-010-720-003	Identity per 14 CFR §21.303 <u>Dwg No.:</u> AI-4010-720 <u>Rev:</u> I/R <u>Date:</u> 06/13/2008, or later FAA approved revisions	Bell Helicopter Textron	204B 205A
Slider -Tail Rotor Assembly	TAI-4010-720- 005	204-010-720-005	Identity per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-720 <u>Rev:</u> I/R <u>Date:</u> 06/13/2008, or later FAA approved revisions	Bell Helicopter Textron	204B 205A
Bearing and Sleeve Assembly	23058131TA	23058131	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23058131TA <u>Rev:</u> I/R <u>Date:</u> 04/17/2009, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30S, C30R, C30R/1, C30U, C40B, C47B, C47M
Bearing, Ball	3103585AL-1	3103585-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3103585AL <u>Rev:</u> A <u>Date:</u> 08/22/2008, or later FAA approved revisions	Honeywell	TPE331-5A, 5AB, 10GP, 10GT, 10UG, 10UGR, 11U, 12UA, 12UAR, 12UHR
Shroud, Power Turbine	3120298-01TA	3120298-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3120298-01TA <u>Rev:</u> I/R <u>Date:</u> 07/07/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-41, 42, 42A, 60A, 60AG, 61, 65B, 65R, 65AG, 65AR

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Shaft, Stub, Compressor Rotor	3120663-01TA	3120663-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3120663-01TA <u>Rev:</u> A <u>Date:</u> 07/22/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-64, 65AG, 65AR, 65B, 65R, 66, 66A, 66B, 66D, 67, 67A, 67AF, 67AG, 67B, 67D, 67R, 67T
Adapter, Spline-Turbine 2 nd Stage	23072099TA	23072099	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23072099TA <u>Rev:</u> I/R <u>Date:</u> 01/07/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, C20F/1, C17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Bearing, Roller 40mm I.D. x 68mm O.D. x 15mm Width	23001949AL	23001949	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23001949AL <u>Rev:</u> I/R <u>Date:</u> 03/6/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Cover Assembly, Oil Sump - Power Support	23001952AL	23001952	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23001952AL <u>Rev:</u> I/R <u>Date:</u> 02/18/2000, or later FAA approved revisions	Rolls-Royce	250-C28, C28B, C28C
Seal Power Turbine Support	23004203AL	23004203	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004203AL <u>Rev:</u> I/R <u>Date:</u> 10/15/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, C20, C20B, C20C, C20F, C20R2, C28B, C28C, C30

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Bearing, Ball 40mm x 68mm x 15mm	23004553AL	23004553	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23004553AL <u>Rev:</u> I/R <u>Date:</u> 07/30/2001, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, B15G, C17, C17B, C17C, C17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Bearing Roller 30mm ID x 62mm OD x16mm Wide	23004554AL	23004554	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004554AL <u>Rev:</u> I/R <u>Date:</u> 03/06/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W
Race, Bearing Inner	23005267AL	23005267	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23005267AL <u>Rev:</u> I/R <u>Date:</u> 07/15/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, 250-C30, C30P, C30M, C30S, C30G, C30G/2
Bearing, Ball 12 x 37 x 12mm	23005746AL	23005746	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23005746AL <u>Rev:</u> I/R <u>Date:</u> 09/24/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Bearing, Ball – 15 x 42 x 13mm	23005747AL	23005747	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23005747AL <u>Rev:</u> I/R <u>Date:</u> 11/6/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M

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Bearing, Ball - Annular 10mm X 26mm X 8mm	23005749AL	23005749	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23005749AL <u>Rev:</u> I/R <u>Date:</u> 03/2/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Bearing, Ball - Annular 25mm X 42mm X 9mm	23007151AL	23007151	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23007151AL <u>Rev:</u> I/R <u>Date:</u> 03/2/2001, or later FAA approved revisions	Rolls-Royce	250-C10, B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Bearing, Ball 20mm I.D. X 42mm O.D. X 12mm Width	23007152AL	23007152	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23007152AL <u>Rev:</u> I/R <u>Date:</u> 02/21/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20W, C28B, C28C, C30, C30G, C30G2 C30L, C30M, C30P, C30R, C30R/1, C30S, C30U

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Bearing, Ball	23007157AL	23007157	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23007157AL <u>Rev:</u> I/R <u>Date:</u> 01/20/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30S
Bearing, Outer Race and Roller Assembly	23007202AL	23007202	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23007202AL <u>Rev:</u> I/R <u>Date:</u> 02/21/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Diffuser Assembly, Compressor	23007208AL	23007208	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23007208AL <u>Rev:</u> J <u>Date:</u> 08/08/2008, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Bearing, Ball - Annular 17mm x 30mm x 7mm	23008065AL	23008065	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23008065AL <u>Rev:</u> I/R <u>Date:</u> 07/16/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Bearing, Ball - Annular 12 mm x 28 mm x 8 mm	23008469AL	23008469	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23008469AL <u>Rev:</u> I/R <u>Date:</u> 08/21/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Bearing, Roller 17mm x 30mm x 9mm	23009609AL	23009609	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23009609AL <u>Rev:</u> I/R <u>Date:</u> 10/15/2001, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4
Bearing, Ball - Annular 20mm x 42mm x 15mm	23031478AL	23031478	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23031478AL <u>Rev:</u> I/R <u>Date:</u> 07/17/2001, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30L, C30M, C30P, C30R, C30S, C30U, C30G2, C30R/1, C30R/3, C40B, C47B, C47M

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Tube, Oil Delivery	23034102AL	23034102	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23034102AL <u>Rev:</u> I/R <u>Date:</u> 06/04/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Race, Bearing Inner, 20 mm ID	23034786AL-1; -2	23034786-1; -2	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23034786AL <u>Rev:</u> I/R <u>Date:</u> 06/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/4, C20S, C20W
Bearing, Outer Race and Roller Assembly – 32mm OD X 7mm Wide	23034787AL	23034787	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23034787AL <u>Rev:</u> I/R <u>Date:</u> 02/21/2001, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2 C20, C20B, C20C, C20F, C20J, C20R, C20R/4, C20S, C20W
Ring, Energy, Absorbing Internal	23035175AL	23035175	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23035175AL <u>Rev:</u> I/R <u>Date:</u> 12/10/2003, or later FAA approved revisions	Rolls-Royce	250-B17B, B17C, B17D, B17E, B17F, C20F/1, C20F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Diffuser Assembly, Compressor	23051119AL	23051119	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23051119AL <u>Rev:</u> J <u>Date:</u> 07/03/2008, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R (T703- AD-700), C30S, C30U

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Bearing, Roller Cylinder Assembly 40mm ID X 68mm OD X 14 mm Wide	23053297AL	23053297	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23053297AL <u>Rev:</u> I/R <u>Date:</u> 02/21/2001, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2 C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Nozzle Assembly, Turbine - 4 th Stage	23055939AL	23055939	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23055939AL <u>Rev:</u> I/R <u>Date:</u> 04/22/2003, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20F, C20J, C20W
Sleeve, No. 8 Bearing	23058128AL	23058128	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23058128AL <u>Rev:</u> I/R <u>Date:</u> 08/21/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R1, C30R3, C40B, C47B, C47M
Bearing, Roller, Front Compressor	23066716AL	23066716	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23066716AL <u>Rev:</u> I/R <u>Date:</u> 11/06/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G2, C30M, C30P, C30S, C40B, C47B, C47M
3 rd Stage Nozzle	23069744AL	23069744	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23069744AL <u>Rev:</u> I/R <u>Date:</u> 04/22/2003, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20F, C20J, C20W

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Adapter, Splined- Turbine - 2 nd Stage	23072099AL	23072099	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23072099AL <u>Rev:</u> I/R <u>Date:</u> 03/30/2004, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, C20F/1, C17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Nozzle Assembly; Turbine - 2 nd Stage	23072654AL	23031938	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23072654AL <u>Rev:</u> I/R <u>Date:</u> 05/07/2002, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Gearshaft Assembly Spur Adapter	23073520AL	23073520	Test & Computation Per 14 CFR §21.303 <u>Dwg No.:</u> 23073520AL <u>Rev:</u> I/R <u>Date:</u> 07/15/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30P, C30M, C30S, C30G, C30G/2
Gearshaft, Spur Adapter	23073521AL	23073521	Test & Computation Per 14 CFR §21.303 <u>Dwg No.:</u> 23073521AL <u>Rev:</u> I/R <u>Date:</u> 07/29/2005, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, 250-C30, C30P, C30M, C30S, C30G, C30G/2
Shield Assembly Turbine - 1 st Stage Nozzle	23073566AL	23073566	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23073566AL <u>Rev:</u> I/R <u>Date:</u> 05/25/2004, or later FAA approved revisions	Rolls-Royce	250-C30, C30S, C30G, C30G/2, C30P, C30M, C30R, C30U

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Gearshaft, Helical Pinion	23077527AL	23077527	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23077527AL <u>Rev:</u> I/R <u>Date:</u> 01/10/2005, or later FAA approved revisions	Rolls-Royce	250-C30, C30P, C30S, C30R, C30R/1, C30U, C30M, C40B, C47B, C47M
Gear, Helical, Power Train Pinion	23084247AL	23084247	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23084247AL <u>Rev:</u> I/R <u>Date:</u> 01/12/2008, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R (T703- AD-700), C30R/1 (T703-AD-700B), C30R/3, C30R/3M, C30S, C30U, C47B, C47M
Gearshaft Assembly, Helical, Torquemeter	23084248AL	23084248	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23084248AL <u>Rev:</u> I/R <u>Date:</u> 01/16/2008, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R (T703- AD-700), C30R/1 (T703-AD-700B), C30R/3, C30R/3M, C30S, C30U, C47B, C47M
Gearshaft Assembly, Helical – PTO	23084249AL	23084249	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23084249AL <u>Rev:</u> I/R <u>Date:</u> 01/15/2008, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R (T703- AD-700), C30R/1 (T703-AD-700B), C30R/3, C30R/3M, C30S, C30U, C47B, C47M
Gear, Spur, Gas Producer – Train Idler	6824977AL	6824977	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6824977AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W

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Cage, Bearing - Plain 1.654 I.D. x .467 Wide (and oversizes)	6843358AL-1 thru -5	6843358-1 thru -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843358AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28, C28B, C28C, C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M, C10
Guide Assembly, Poppet - Anti- Icing Valve	6852176AL	6852176	Identity per 14 CFR §21.303 <u>Dwg. No.:</u> 6852176AL <u>Rev:</u> I/R <u>Date:</u> 05/15/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Tube Assembly, Scavenge Oil Gasifier Turbine Support to Accessory Housing	6853463AL	6853463	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6853463AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S
Tube Assembly, External Sump Accessory Housing	6853464AL	6853464	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6853464AL <u>Rev:</u> I/R <u>Date:</u> 10/25/1999, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Body, Valve- Anti-Icing	6856984AL	6856984	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6856984AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Poppet & Seat-Assembly - Anti-Icing Valve	6856986AL	6856986	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6856986AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Poppet Assembly, Anti-Icing Valve	6856987AL	6856987	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6856987AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Tube Assembly, Fuel Control to Pump	6859172AL	6859172	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6859172AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20C, C20B, C20F, C20J, C20R, C20 R/1, C20R/2, C20R/4, C20S, C20W

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Bearing, Ball Annular, 25mm ID X 52mm OD X 15mm Wide	6859433AL	6859433	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 6859433AL <u>Rev:</u> I/R <u>Date:</u> 02/28/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Tube Assembly, Press Oil, Access. Housing to Front Support	6859956AL	6859956	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6859956 <u>Rev:</u> I/R <u>Date:</u> 10/25/1999, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J
Race, Bearing - Inner 25mm I.D. x 11mm Width	6870031AL	6870031	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 6870031AL <u>Rev:</u> I/R <u>Date:</u> 07/16/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30R/2, C30S, C30U, C40B, C47B, C47M
Diffuser Assembly, Compressor	6870813AL	6870813	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6870813AL <u>Rev:</u> D <u>Date:</u> 09/09/2008, or later FAA approved revisions.	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C (T63-A-720), C20F, C20J, C20S, C20W

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Bearing, Ball 40mm ID X 62mm OD X 12mm Wide	6871505AL	6871505	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 6871505AL <u>Rev:</u> I/R <u>Date:</u> 02/28/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20W, C28B, C28C, C30, C30G, C30G2 C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Diaphragm, Compressor Bleed Valve	6874725AL	6874725	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6874725AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F,-C20J, C20R/1, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Bearing, Roller, 25mm x 62mm x 17mm	6875035AL	6875035	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 6875035AL <u>Rev:</u> I/R <u>Date:</u> 08/21/2001, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Bearing, Ball - 30mm I.D. x 55mm O.D. x 13mm Width	6889093AL	6889093	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6889093AL <u>Rev:</u> I/R <u>Date:</u> 03/06/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W

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Bearing, Ball 40mm x 80mm x 18mm	6889336AL	6889336	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 6889336AL <u>Rev:</u> I/R <u>Date:</u> 07/02/2002, or later FAA approved revisions	Rolls-Royce	250-28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R1, C30S, C30U, C40B, C47B, C47M
Bearing, Ball - Annular 10mm x 26mm	6898607AL	6898607	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 6898607AL <u>Rev:</u> I/R <u>Date:</u> 03/02/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C10, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, C28B, C28C, C30, C30L, C30M, C30P, C30R, C30S, C40B, C47B, C47M
Gearshaft Assembly Helical PTO	6899402AL	6899402	Identity per 14 CFR §21.303 <u>Dwg. No.:</u> 6899402AL <u>Rev:</u> I/R <u>Date:</u> 03/27/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20R, C20R/1,-C20R/2, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Plate, Support - Nozzle Assembly, Turbine 1 st Stage	23001806AL	23001806	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23001806AL <u>Rev:</u> I/R <u>Date:</u> 03/03/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Key, Nozzle - Turbine 1 st Stage	23001816AL	23001816	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23001816AL <u>Rev:</u> I/R <u>Date:</u> 03/03/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30

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Nozzle Assembly, Pressure Oil- Power Turbine Support	23001819AL	23001819	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23001819AL <u>Rev:</u> I/R <u>Date:</u> 03/03/1999, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C28B, C28C
Small Corrugated Space 1 st Section	23003108AL	23003108	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23003108AL <u>Rev:</u> I/R <u>Date:</u> 03/10/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30
Primary Combustion Chamber 2 nd Section	23003109AL	23003109	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23003109AL <u>Rev:</u> I/R <u>Date:</u> 03/27/1998, or later FAA approved revisions	Rolls-Royce	250-C28, C28B, C28C, C30, C30G, C30G2, C30M, C30P, C30S
Large Corrugated Space 2 nd Section	23003110AL	23003110	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23003110AL <u>Rev:</u> I/R <u>Date:</u> 03/10/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30
Spacer Clip – Combustor	23003111AL	23003111	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23003111AL <u>Rev:</u> I/R <u>Date:</u> 03/10/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30
Cover, Power Turbine Support	23004204AL	23004204	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004204AL <u>Rev:</u> I/R <u>Date:</u> 11/25/1997, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, C20, C20B, C20F, C28B, C28C, C20R, C20R1, C20R2

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Ring, Keyed - Nozzle Assembly Turbine, 1 st Stage	23004206AL	23004206	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004206AL <u>Rev:</u> I/R <u>Date:</u> 03/03/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30
Diaphragm - Nozzle Assembly Turbine, 1 st Stage	23004207AL	23004207	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004207AL <u>Rev:</u> I/R <u>Date:</u> 03/03/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Gearshaft Assembly – Helical P.T.O.	23004505AL	23004505	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004505AL <u>Rev:</u> I/R <u>Date:</u> 11/14/2000, or later FAA approved revisions	Rolls-Royce	250-C20B, C20J, C20R2, C20R/4
Dome, Combustion Liner - Sub Assembly	23004529AL	23004529	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23004529AL <u>Rev:</u> I/R <u>Date:</u> 10/10/2000, or later FAA approved revisions	Rolls-Royce	B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Body Assembly, Compressor Bleed Intermediate	23005368AL	23005368	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23005368AL <u>Rev:</u> I/R <u>Date:</u> 12/09/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, G30G/2, C30M, C30P, C30R/1, C30S, C40B, C47B, C47M
Diffuser Assembly, Compressor	23007210AL	23007210	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23007210AL <u>Rev:</u> I/R <u>Date:</u> 12/09/1999, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, G30G/2, C30L, C30M, C30P, C30R, C30S, C30U

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Nozzle Assembly Turbine - 1 st Stage	23008593AL	23008593AL	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23008593AL <u>Rev:</u> I/R <u>Date:</u> 03/03/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Shaft, Power Turbine – Inner	23030955AL	23030955	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23030955AL <u>Rev:</u> I/R <u>Date:</u> 11/16/2000, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Shaft, Power Turbine – Outer	23030960AL	23030960	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23030960AL <u>Rev:</u> I/R <u>Date:</u> 11/04/2000, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Adapter, Splined – Turbine, 2 nd Stage	23030974AL	23030974	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 2303974AL <u>Rev:</u> I/R <u>Date:</u> 05/01/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30L, C30M, C30P, C30R, C30S, C30G2, C30R/1, C30R/3, C40B, C47B, C47M
Seal, Serrated Balance Piston	23030978AL-1; 23030978AL-2	23030978-1; 23030978-2	Identicality per 14 CFR §21.303 <u>Dwg. No.:</u> 23030978AL <u>Rev:</u> I/R <u>Date:</u> 03/10/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, C20, C20B, C20C, C20F, C20J, C20R/1, C20R/2, C20S, C28B, C28C, C30, C30G, C30L, C30M, C30P, C30R, C30S, C30G/2, T63-A-720

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Gear, Spur-Oil Pump	23030989AL	23030989	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23030989AL <u>Rev:</u> I/R <u>Date:</u> 09/28/2000, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, G30G/2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Gear Shaft Assembly, Spur Oil Pump and Gasifier Tach	23030990AL	23030990	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23030990AL <u>Rev:</u> I/R <u>Date:</u> 05/10/2000, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C28L, C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Body Assembly, Pump- Scavenge, Oil	23030999AL	23030999	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23030999AL <u>Rev:</u> I/R <u>Date:</u> 09/15/1999, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Dome, Shield Assembly Turbine Nozzle -1st Stage	23031415AL	23031415	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031415AL <u>Rev:</u> I/R <u>Date:</u> 06/04/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30L, C30M, C30P, C30R, C30S, C30G/2

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Plate, Support Nozzle Assembly Turbine 1 st Stage	23031419AL	23031419	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031419AL <u>Rev:</u> I/R <u>Date:</u> 10/29/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20S
Gearshaft, Spur Adapter	23031921AL	23031921	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031921AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Gearshaft Assembly Spur Adapter	23031922AL	23031922	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031922AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Diaphragm, 2 nd Stage Nozzle Assembly	23031934AL	23031934	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031934AL <u>Rev:</u> C <u>Date:</u> 11/13/2002, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30S
Nozzle Assembly, Turbine 2 nd Stage	23031938AL	23031938	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23031938AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W

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Cage, Pinion Bearing - Power Turbine	23032283AL-1 thru -3	23032283-1 thru - 3	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23032283AL <u>Rev:</u> I/R <u>Date:</u> 05/06/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Vane and Band Assembly – 1 st Stage	23032631AL	23032631	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23032631AL <u>Rev:</u> I/R <u>Date:</u> 07/04/2000, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4
Vane and Band Assembly – 2nd Stage	23032632AL	23032632	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23032632AL <u>Rev:</u> I/R <u>Date:</u> 09/05/2000, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2
Vane and Band Assembly – 3rd ¹ Stage	23032633AL	23032633	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23032633AL <u>Rev:</u> I/R <u>Date:</u> 09/05/2000, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4
Vane and Band Assembly 4 th Stage	23032634AL	23032634	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23032634AL <u>Rev:</u> I/R <u>Date:</u> 09/05/2000, or later FAA approved revisions	Rolls-Royce	250-B17F, B17F/1, B17F/2, C20R, C20R/1, C20R/2, C20R/4

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Gearshaft Assembly Helical- Torquemeter	23035299AL	23035299	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23035299AL <u>Rev:</u> I/R <u>Date:</u> 06/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20R, C20R/1, C20R/2, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Shaft, Power Turbine-Inner	23037403AL	23037403	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23037403AL <u>Rev:</u> I/R <u>Date:</u> 06/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Shaft, Power Turbine – Outer	23037413AL	23037413	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23037413AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Seal, Labyrinth, Power Turbine – Rotating	23037444AL	23037444	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23037444AL <u>Rev:</u> I/R <u>Date:</u> 08/24/2001, or later FAA approved revisions	Rolls-Royce	C18, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Shaft, Power Turbine – Outer	23038136AL	23038136	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23038136AL <u>Rev:</u> I/R <u>Date:</u> 05/01/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30L, C30M, C30P, C30R, C30S, C30U, C30G2, C30R/1, C30R/3, C40B, C47B, C47M
Gearshaft Assembly – Helical Torquemeter	23038191AL	23038191	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23038191AL <u>Rev:</u> I/R <u>Date:</u> 11/14/2000, or later FAA approved revisions	Rolls-Royce	250-C20J, C20R2, C20R4
Nozzle Assembly Turbine 1st Stage	23038218AL	23038218	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23038218AL <u>Rev:</u> I/R <u>Date:</u> 06/02/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Coupling, Adapter Compressor	23039791AL -1 thru -3	23039791-1 thru -3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23039791AL <u>Rev:</u> I/R <u>Date:</u> 09/24/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W
Bushing	23039909AL-1; -2	23039909-1; -2	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23039909AL <u>Rev:</u> I/R <u>Date:</u> 09/18/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17F, B17F/1, B17F/2, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Cover, Diffuser Manifold	23051099AL	23051099	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23051099AL <u>Rev:</u> I/R <u>Date:</u> 09/03/1999, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, G30G/2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U

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Ring, Diaphragm Nozzle Assembly Turbine 1 st Stage	23051629AL	23051629	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23051629AL <u>Rev:</u> I/R <u>Date:</u> 09/17/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20S, C20W
Key, Nozzle Turbine 1 st Stage	23051630AL	23051630	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23051630AL <u>Rev:</u> I/R <u>Date:</u> 09/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20S
Diaphragm – Nozzle Assembly Turbine 1 st Stage	23051631AL	23051631	Identicality per 14 CFR §21.303 <u>Dwg No.</u> 23051631AL <u>Rev:</u> I/R <u>Date:</u> 09/17/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20S, C20W
Lever, Actuating Anti-Icing Valve	23053191AL	23053191	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23053191AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Valve Assembly Anti-Icing	23053192AL	23053192	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 23053192AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720

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Sump, Assembly- Scav Oil- External	23053320AL	23053320	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23053320AL <u>Rev:</u> IR <u>Date:</u> 11/25/1997, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W, C28B, C28C
Tube Assembly PC Air-Scroll to PC Filter	23054628AL	23054628	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23054628AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20F, C20J, C20W
Key, Nozzle – Turbine 2 nd Stage	23055441AL-1; -2	23055441-1; -2	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23055441AL <u>Rev:</u> A <u>Date:</u> 10/16/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30L, C30M, C30, C30R, C30S, C30U
Tube Governor PY to Fuel Control	23057137AL	23057137	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23057137AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W
Plate, Thrust – Gas Producer Bearing	23058137AL	23058137	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23058137AL <u>Rev:</u> I/R <u>Date:</u> 08/22/2000, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30L, C30M, C30P, C30R, C30S, C30U, C30G2, C30R/1, C40B, C47B, C47M
Plate Retaining - Gas Producer Thrust Bearing	23059551AL	23059551	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23059551AL <u>Rev:</u> I/R <u>Date:</u> 02/18/2000, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C28L, C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U

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Liner Assembly Combustion	23060429AL	23060429	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23060429AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Cone, 1 st Section - Combustion Liner Assembly	23061213AL	23061213	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23061213AL <u>Rev:</u> C <u>Date:</u> 03/13/2001, or later FAA approved revisions	Rolls-Royce	B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Nozzle Assembly Turbine - 4 th Stage	23061922AL	23061922	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23061922AL <u>Rev:</u> I/R <u>Date:</u> 08/22/2000, or later FAA approved revisions	Rolls-Royce	250-C20B, C20F, C20J, C20S, C20W
Shield Assembly Turbine 1 st Stage Nozzle	23062752AL	23062752	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23062752AL <u>Rev:</u> I/R <u>Date:</u> 06/06/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Nozzle Assembly Turbine - 1 st Stage	23062753AL 23062753AL-2; -3; -4; -5	23062753 23062753-2; -3; - 4; -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23062753AL <u>Rev:</u> A <u>Date:</u> 04/06/1999, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S, C20W
Nozzle Assembly 1 st Stage - Turbine	23062754AL	23062754	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23062754AL <u>Rev:</u> I/R <u>Date:</u> 10/26/1999, or later FAA approved revisions	Rolls-Royce	250-C20R, C20R/1, C20R/2, C20R/4

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Shield Assembly Turbine - 1 st Stage Nozzle	23062765AL	23062765	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23062765AL <u>Rev:</u> I/R <u>Date:</u> 06/06/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C
Gearshaft, Spur – Gasifier Train, Inner	23063413AL	23063413	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23063413AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S
Shaft, Power Turbine-Inner	23071309AL	23071309	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23071309AL <u>Rev:</u> I/R <u>Date:</u> 02/28/2001, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Shaft, Power, Turbine-Inner	23071313AL	23071313	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 23071313AL <u>Rev:</u> I/R <u>Date:</u> 05/01/2001, or later FAA approved revisions	Rolls-Royce	250-C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R1, C30R3, C40B, C30S, C30U, C47B, C47M
Shield Assembly Turbine - Nozzle 1 st Stage	23073533AL	23073533	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23073533AL <u>Rev:</u> I/R <u>Date:</u> 05/25/2004, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Shield Assembly Turbine - Nozzle 1 st Stage	23073565AL	23073565	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 23073565AL <u>Rev:</u> I/R <u>Date:</u> 05/25/2004, or later FAA approved revisions	Rolls-Royce	250-C28, C28B, C28C
Bushing - Sleeve .625 X .815 X .437	6810448AL	6810448	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6810448AL <u>Rev:</u> I/R <u>Date:</u> 12/03/1999, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28, C28C, C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M
Adapter, Splined	6842095AL	6842095	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6842095AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Cage, Bearing - Plain 1.103 I.D. x .312 Wide (and oversizes)	6843356AL-1 thru -5	6843356-1 thru -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843356AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28, C28B, C28C, C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M, C10
Cage, Bearing - Plain 2.678 I.D. x .590 Wide (and Oversizes)	6843357AL-1 thru -5	6843357-1 thru -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843357AL <u>Rev:</u> I/R <u>Date:</u> 05/15/1998, or later FAA revisions	Rolls-Royce	250-B15B, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W
Cage, Bearing - Cover Gear Box (and oversizes)	6843359AL-1 thru -3	6843359-1 thru -3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843359AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C20
Cage, Ball Bearing - 1.182 I.D. x .271 Wide (and oversizes)	6843360AL-1 thru -5	6843360-1 thru -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843360AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Cage, Bearing - Plain 1.024 I.D. x .312 Wide and oversizes)	6843361AL-1 thru -5	6843361-1 thru -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843361AL <u>Rev:</u> I/R <u>Date:</u> 05/10/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Cage Bearing - Plain, 1.654 I.D. x 1.389 Wide (and oversizes)	6843362AL-1 thru -5	6843362-1 thru -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843362AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Cage, Bearing - Plain 1.654 I.D. x .849 Wide (and oversizes)	6843363AL-1 thru -5	6843363-1 thru -5	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6843363AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Tube Assembly Fuel Control to Fire Shield	6844169AL	6844169	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6844169AL <u>Rev:</u> I/R <u>Date:</u> 10/25/1999, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Diaphragm, Nozzle - Turbine, 2 nd Stage	6851499AL	6851499	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6851499AL <u>Rev:</u> D <u>Date:</u> 05/07/2003, or later FAA approved revisions	Rolls-Royce	B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Seal, Labyrinth - Turbine 4th Stage	6851530AL	6851530	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6851530AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20R/1, C20R/2, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Insert, Screw Thread Flanged, .4375-20 x .281 I.D.	6854936AL-1 thru -3	6854936-1 thru -3	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6854936AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C10
Seat, Poppet - Anti-Icing Valve	6856988AL	6856988	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6856988AL <u>Rev:</u> I/R <u>Date:</u> 05/15/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W
Tube Assembly Fuel Pump to Control	6859177AL	6859177	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6859177AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Tube Assembly Air- Power Turbine Governor to Control	6870035AL	6870035	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6870035AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Ring, Piston, Split Seal - 2.230 O.D.	6870736AL	6870736	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6870736AL <u>Rev:</u> I/R <u>Date:</u> 05/29/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Ring, Piston, Split Seal - 2.970 O.D.	6870737AL	6870737	Identity per 14 CFR §21.303 <u>Dwg. No.:</u> 6870737AL <u>Rev:</u> I/R <u>Date:</u> 05/15/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Tube Assembly Air- Fuel Control to Regulator Pressure	6871461AL	6871461	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6871461AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W
Tube Assembly Anti-Ice Right Hand	6871473AL	6871473	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6871473AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20F, C20J

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Tube Assembly Oil- Access, HSG, to Check Valve	6871937AL	6871937	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6871937AL <u>Rev:</u> I/R <u>Date:</u> 10/25/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S, C30U
Valve, Compressor Bleed	6874816AL	6874816	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6874816AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W
Nozzle, Compressor- Bleed Valve	6874818AL-1 thru -12	6874818	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6874818AL <u>Rev:</u> I/R <u>Date:</u> 05/15/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C20, C20B, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W
Insert, Screw Thread. - Flanged, .250- 28 X .500-16 X .490 Long	6875011AL-1 thru -3	6875011-1 thru -3	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6875011AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Insert, Scroll Discharge	6875248AL	6875248	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6875248AL <u>Rev:</u> I/R <u>Date:</u> 04/30/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Seal, Mating Ring – Rotating	6875491AL	6875491	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6875491AL <u>Rev:</u> I/R <u>Date:</u> 09/18/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Liner Assembly Partial – Combustion, AFT Section	6875628AL	6875628	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6875628AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Ring, Retaining – External	6875736AL	6875736	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6875736AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C10

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Stud, Shouldered, Steep - .250- 28 X .500-16 X 1.374 Long	6876030AL-1 thru -5	6876030-1 thru - 5	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6876030AL <u>Rev:</u> I/R <u>Date:</u> 03/03/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30L, C30M, C30P, C30R, C20S
Bolt, Machine - Slab HD, .190-32 UNJF X .500 Long, Close Tolerance, Body BD, AMS 5731	6876097AL	6876097	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6876097AL <u>Rev:</u> I/R <u>Date:</u> 05/15/1998, or later FAA approved revisions	Rolls-Royce	250-B17B, B17C, B17D, B17E, C20B, C20C, C20F, C20J, C20R/1, C20R/2, C20S, C20W
Tube Assembly Air Bleed - Valve	6876294AL	6876294	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6876294AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J
Tube Assembly Anti-Ice Left Hand	6876333AL	6876333	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6876333AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20F, C20J
Gearshaft Assembly - Fuel Control - Oil Pump	6876529AL	6876529	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6876529AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W

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Tube Assembly Air PC Filter to Gov. Tee	6876542AL	6876542	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6876542AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W
Nut Spanner – 1.125-32 Spl. Lock, L.H.	6876878AL	6876878	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6876878AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20S
Tube Assembly Fuel Control to Accumulator	6877277AL	6877277	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6877277AL <u>Rev:</u> I/R <u>Date:</u> 12/17/1998, or later FAA approved revisions	Rolls-Royce	250- C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4
Seal, Labyrinth Stator-Gasifier Turbine Bearing	6877736AL-1; - 2	6877736-1; -2	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6877736AL <u>Rev:</u> I/R <u>Date:</u> 10/29/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W
Nozzle Assembly Turbine – 2 nd Stage, IN-738	6878426AL	6878426	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6878426AL <u>Rev:</u> I/R <u>Date:</u> 05/01/2001, or later FAA approved revisions	Rolls-Royce	250-B15G, C18, C20

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Gearshaft Starter Generator	6888529AL	6888529	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6888529AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Cover Assembly Oil Sump - Power Turbine Support	6888547AL	6888547	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6888547AL <u>Rev:</u> I/R <u>Date:</u> 09/09/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R1, C20R2, C20S, C20W
Gearshaft, Spur – Gasifier Train Idler	6889094AL	6889094	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6889094AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20S
Ring, Piston- Split Seal, 2.8000 O.D	6889096AL	6889096	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6889096AL <u>Rev:</u> I/R <u>Date:</u> 07/06/2000, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30L, C30M, C30P, C30R, C30R/1, C30R1/3, C30S, C30U, C40B, C47B, C47M
Gear, Helical- Power Train Drive	6889700AL	6889700	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6889700AL <u>Rev:</u> I/R <u>Date:</u> 07/28/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Diaphragm - 4th Stage Nozzle	6890034AL	6890034	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6890034AL <u>Rev:</u> I/R <u>Date:</u> 06/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17, C20, C20B, C20C
Ring – 4 th Stage Nozzle	6890035AL	6890035	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6890035AL <u>Rev:</u> I/R <u>Date:</u> 09/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17, C20, C20B
Shield Assembly Turbine - 1 st Stage Nozzle	6890040AL	6890040	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 68890040AL <u>Rev:</u> I/R <u>Date:</u> 06/04/1999, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Body Assembly Pump-Pressure Oil	6892071AL	6892071	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6892071AL <u>Rev:</u> I/R <u>Date:</u> 02/18/2000, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Ring - Nozzle, Turbine - 2 nd Stage	6892823AL	6892823	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6892823AL <u>Rev:</u> I/R <u>Date:</u> 04/23/1999, or later FAA approved revisions	Rolls-Royce	B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Shield Assembly Turbine - 1 st Stage Nozzle	6892828AL	6892828	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6892828AL <u>Rev:</u> I/R <u>Date:</u> 06/06/2001, or later FAA approved revisions	Rolls-Royce	250-30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30M
Dampener- Bellows Compressor Bleed Valve	6893488AL	6893488	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6893488AL <u>Rev:</u> I/R <u>Date:</u> 12/3/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, G30G/2, C30M, C30P, C30R/1, C30S, C40B, C47B, C47M
Deflector, Turbine Nozzle – 1 st Stage	6893676AL	6893676	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6893676AL <u>Rev:</u> I/R <u>Date:</u> 09/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W, C28B, C28C, C30, C30G, C30L, C30M, C30P, C30R, C30S, C30U
Guide, Valve Bushing	6895180AL	6895180	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6895180AL <u>Rev:</u> I/R <u>Date:</u> 06/20/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R/1, C30S, C30M, C40B, C47B, C47M
Shield Assembly Turbine - 1 st Stage Nozzle	6895826AL	6895826	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6895826AL <u>Rev:</u> I/R <u>Date:</u> 05/25/2001, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C

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Nut, Spline – Lock	6896408AL	6896408	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6896408AL <u>Rev:</u> I/R <u>Date:</u> 08/22/2000, or later FAA approved revisions	Rolls-Royce	250-28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30R, C30R/1, C30S, C30U
Gearshaft, Spur – Fuel Control and Oil Pump	6896437AL	6896437	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6896437AL <u>Rev:</u> I/R <u>Date:</u> 09/02/1998, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Adapter, Splined- Turbine 2 nd Stage	6898784AL	6898784	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6898784AL <u>Rev:</u> I/R <u>Date:</u> 03/10/1998, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Bushing-Body Compressor Bleed Valve	6899049AL	6899049	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6899049AL <u>Rev:</u> I/R <u>Date:</u> 09/17/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30M, C30P, C30S, C30G/2, C30R/1, C40B, C47B, C47M
Valve, Poppet CPSRS Bleed	6899050AL	6899050	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 6899050AL <u>Rev:</u> I/R <u>Date:</u> 01/30/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G2, C30R/1

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Valve, Servo Compressor Bleed	6899051AL	6899051	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899051AL <u>Rev:</u> I/R <u>Date:</u> 01/30/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30M, C30P, C30S, C30G, C30G2, C30R/1
Diaphragm, Compressor – Bleed Valve	6899114AL	6899114	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899114AL <u>Rev:</u> I/R <u>Date:</u> 09/02/1998, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G2, C30M, C30P, C30R/1, C30S
Orifice, Compressor Bleed Valve	6899282AL -1 thru -39	6899282 -1 thru -39	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899282AL <u>Rev:</u> I/R <u>Date:</u> 12/06/1999, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, G30G/2, C30M, C30P, C30R/1, C30S, C40B, C47B, C47M
Vane and Band Assembly, 1 st Stage	6899391AL	6899391	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899391AL <u>Rev:</u> I/R <u>Date:</u> 08/23/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W
Vane and Band Assembly, 2 nd Stage	6899392AL	6899392	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899392AL <u>Rev:</u> I/R <u>Date:</u> 08/23/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W
Vane and Band Assembly, 3 rd Stage	6899393AL	6899393	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899393AL <u>Rev:</u> I/R <u>Date:</u> 08/21/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W

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Vane and Band Assembly, 4 th Stage	6899394AL	6899394	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899394AL <u>Rev:</u> I/R <u>Date:</u> 08/21/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W
Vane and Band Assembly, 5 th Stage	6899395AL	6899395	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899395AL <u>Rev:</u> I/R <u>Date:</u> 08/21/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W
Vane and Band Assembly, 6 th Stage	6899396AL	6899396	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 6899396AL <u>Rev:</u> I/R <u>Date:</u> 08/21/1999, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20F, C20J, C20S, C20W
Nut, Bearing, Retaining, Power Output Gear	1-030-023- 02AL	1-030-023-02	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-023-02AL <u>Rev:</u> B <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Plate, Face, Output Gear Shaft	1-030-067- 02AL	1-030-067-02	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-067-02AL <u>Rev:</u> B <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Carrier, Rear Reduction Gear	1-030-181- 01AL	1-030-181-01	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-181-01AL <u>Rev:</u> C <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B

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Carrier, Front Reduction Gear	1-030-182- 03AL	1-030-182-03	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-182-03AL <u>Rev:</u> C <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Plate, Rear Torquemeter	1-030-183- 03AL	1-030-183-03	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-183-03AL <u>Rev:</u> C <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Liner, Bearing Support, Output Shaft	1-030-187- 03AL	1-030-187-03	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-187-03AL <u>Rev:</u> B <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Gearshaft, Helical, Output	1-030-191- 14AL	1-030-191-14	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-191-14AL <u>Rev:</u> D <u>Date:</u> 05/12/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Gearshaft, Helical, Planet	1-030-193- 05AL	1-030-193-05	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-193-05AL <u>Rev:</u> C <u>Date:</u> 05/13/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Nut, Plane, Round-LH, Reduction Gear	1-030-194- 02AL	1-030-194-02	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-194-02AL <u>Rev:</u> B <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B

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Cup, Locking, Gearshaft Bearing	1-030-195- 01AL	1-030-195-01	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-195-01AL <u>Rev:</u> B <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Retainer, Oil Seal-Output Shaft	1-030-229- 03AL	1-030-229-03	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-229-03AL <u>Rev:</u> C <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Plate Assembly, Torquemeter	1-030-240- 02AL	1-030-240-02	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-240-02AL <u>Rev:</u> A <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Plug, Output Gear	1-030-265- 01AL	1-030-265-01	Identicality per 14 CFR § 1.303 <u>Dwg No.:</u> 1-030-265-01AL <u>Rev:</u> A <u>Date:</u> 05/16/2005, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Deflector, Oil- Assembly of	1-030-450- 02AL	1-030-450-02	Identicality per 14 CFR §21.303 <u>Dwg No.:</u> 1-030-450-02AL <u>Rev:</u> I/R <u>Date:</u> 11/11/2004, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Deflector, Combustion Chamber	1-110-500- 03AL	1-110-500-03	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 1-110-500-03AL <u>Rev:</u> A <u>Date:</u> 10/07/04, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1

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Nozzle, 1 st Stage Turbine – Machining, (Vane Holes)	1-110-520- 19AL	1-110-520-19	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 1-110-520-19AL <u>Rev:</u> I/R <u>Date:</u> 08/22/2000, or later FAA approved revisions	Honeywell	T5313B
Nozzle, 1 st Stage Turbine – Machining, (Vane Holes) Overbridge	1-110-710- 06AL	1-110-710-06	Identity per 14 CFR § 1.303 <u>Dwg No.:</u> 1-110-710-06AL <u>Rev:</u> I/R <u>Date:</u> 09/16/2002, or later FAA approved revisions	Honeywell	T5317A, T5317-1, T5317B
Bearing, Roller	1-300-666- 01AL	1-300-666-01	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-666-01AL <u>Rev:</u> I/R <u>Date:</u> 04/08/2003, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Bearing, Roller	1-300-667- 01AL	1-300-667-01	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-667-01AL <u>Rev:</u> I/R <u>Date:</u> 04/28/2003, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Bearing, Ball	1-300-669- 02AL	1-300-669-02	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-669-02AL <u>Rev:</u> I/R <u>Date:</u> 05/05/2003, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B
Bearing, Roller	1-300-682- 01AL	1-300-682-01	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 1-300-682-01AL <u>Rev:</u> I/R <u>Date:</u> 04/16/2003, or later FAA approved revisions	Honeywell	T5313B, T5317A, T5317A-1, T5317B

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Bearing, Roller	169660AL-2	169660-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 169660AL-1 <u>Rev:</u> I/R <u>Date:</u> 05/09/2006, or later FAA approved revisions		Hamilton Sundstrand Power Systems T-62T-40C7E2 when installed on:
				Dassault Aviation	Mystere-Falcon 20(Series)
					Hamilton Sundstrand Power Systems T-62T-40C2 when installed on:
				Lockheed Martin	1329-23A 1329-23D
					Hamilton Sundstrand Power Systems T-62T-40C3 when installed on:
				Dassault Aviation	Mystere-Falcon 200
					Hamilton Sundstrand Power Systems T-62T-40C3A T-62T-40C3A1 when installed on:
				Dassault Aviation	Mystere-Falcon 50
				Cessna Aircraft Company	650 (Citation VI & VII)

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Bearing, Roller (continued)	169660AL-2	169660-2			Hamilton Sundstrand Power Systems T-62T-40C7A when installed on:
				Cessna Aircraft Company	650 (Citation III)
					Hamilton Sundstrand Power Systems T-62T-40C7D when installed on:
				Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series
					Hamilton Sundstrand Power Systems T-62T-40C7E1 when installed on:
				Embraer S.A.	EMB-120

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Bearing, Roller	171146AL-1	171146-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 171146AL-1 <u>Rev:</u> I/R <u>Date:</u> 12/02/2005, or later FAA approved revisions		Hamilton Sundstrand Power Systems T-62T-39 T-62T-40C7E2 when installed on:
				Dassault Aviation	Mystere-Falcon 20(Series)
					Hamilton Sundstrand Power Systems T-62T-39 when installed on:
				Lockheed Martin	1329-23A 1329-23D
				Sabreliner Corporation	NA-265-60
				Beechcraft Corporation	HS.125 (Series)
					Hamilton Sundstrand Power Systems T-62T-40C2 when installed on:
				Lockheed Martin	1329-23A 1329-23D
					Hamilton Sundstrand Power Systems T-62T-40C3 when installed on:
				Dassault Aviation	Mystere- Falcon 200

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Bearing, Roller (continued)	171146AL-1	171146-1			Hamilton Sundstrand Power Systems T-62T-40C3A T-62T-40C3A1 when installed on: Dassault Aviation Hamilton Sundstrand Power Systems T-62T-40C3A T-62T-40C3A1 T-62T-40C7A when installed on: Cessna Aircraft Company 650 (Citation III) Hamilton Sundstrand Power Systems T-62T-40C7D when installed on: Bombardier DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series Hamilton Sundstrand Power Systems T-62T-40C7E1 when installed on: Embraer S.A. EMB-120
Sleeve - Stator Vane	2000T46AL	2000T46	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 2000T46AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2

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Ring, Mating Seal	3002T15AL	3002T15	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 3002T15AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Bearing, Roller, Flanged	3037278TA	3037278	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3037278TA <u>Rev:</u> I/R <u>Date:</u> 06/11/2009, or later FAA- approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 38, 40, 41, 41AG, 42, 42A, 45A, 45B, 45R, 50, 60A, 60AG, 61, 64, 65AG, 65AR, 65B, 65R, 66, 66A, 66B, 66D, 67, 67A, 67AF, 67AG, 67B, 67D, 67R, 67T, 110, 112, 114, 114A, 121, 135, 135A; PT6T-3, 3B, 3D, 3DE, 3DF, 6, 6B
Washer, Key	37B200102AL	37B200102	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 37B200102AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Ring, Front- Shroud Stator - 1 st Stage	37C300011AL	37C300011	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 37C300011AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140
Fastener, Blade - Turbine Rotor	37C300263AL	37C300263	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 37C300263AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2

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Oil Cooler	37D400479P101 AL	37D400479P101	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 37D400479P101AL <u>Rev:</u> I/R <u>Date:</u> 08/12/2002, or later FAA approved revisions	General Electric	CT58-100-2, 110-1, 110-2, 140-1, 140-2
Shroud, Inlet Guide Vane	4002T34AL	4002T34	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 4002T34AL <u>Rev:</u> I/R <u>Date:</u> 05/10/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Lever, Actuating - Inlet Guide Vane	4004T70AL	4004T70	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 4004T70AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Ring, Shroud- Rear Stator 1 st Stage	4016T64AL	4016T64	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 4016T64AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Lever, Vane Actuating-1 st Stage	573D298AL	573D298	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 573D298AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Lever, Vane Actuating-2 nd and 3 rd Stage	573D299AL	573D299	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 573D299AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2

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Ring, Half- Vane Support, 1 st Stage	573D346AL	573D346	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 573D346 <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Ring, Half Vane Support, 2 nd Stage	573D347AL	573D347	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 573D347AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Ring, Half- Vane Support, 3 rd Stage	573D348AL	573D348	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 573D348AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Ring, Half- Support Inlet Guide Vane	573D349AL	573D349	Identity per 14 CFR §21.303 <u>Dwg No.:</u> 573D349AL <u>Rev:</u> I/R <u>Date:</u> 09/14/2000, or later FAA approved revisions	General Electric	CT58-140-1, CT58-140-2
Drive Shaft	753385AL-2	Hamilton Sundstrand 753385-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 753385AL-2 <u>Rev:</u> B <u>Date:</u> 12/17/2004, or later FAA approved revisions	Bombardier Fokker Services B.V.	DHC-8-100 series DHC-8-300 series CL-600-2B19 (Regional Jet Series 100 & 440) F27 Mark 050

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Fan Spacer	753392AL-1	Hamilton Sundstrand 753392-1	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 753392AL-1 <u>Rev:</u> A <u>Date:</u> 12/09/2004, or later FAA approved revisions	Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series CL-600-2B19 (Regional Jet Series 100 & 440)
				Dassault Aviation	Fan Jet Falcon Fan Jet Falcon Series C Fan Jet Falcon Series D Fan Jet Falcon Series E Fan Jet Falcon Series F
				Fokker Services B.V.	F27 Mark 050
Cooling Fan	762049AL-1	Hamilton Sundstrand 762049-1	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 762049AL <u>Rev:</u> D <u>Date:</u> 10/14/2004, or later FAA approved revisions	Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series CL-600-2B19 (Regional Jet Series 100 & 440)
				Dassault Aviation	Fan Jet Falcon Fan Jet Falcon Series C Fan Jet Falcon Series D Fan Jet Falcon Series E Fan Jet Falcon Series F
				Fokker Services B.V.	F27 Mark 050

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Compressor Rotor	782101AL-1	Hamilton Sundstrand 782101-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 782101AL <u>Rev:</u> C <u>Date:</u> 04/27/2004, or later FAA approved revisions	Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series
				Bombardier	CL-600-2B19 (Regional Jet Series 100 & 440)
				Dassault Aviation	Fan Jet Falcon Fan Jet Falcon Series C Fan Jet Falcon Series D Fan Jet Falcon Series E Fan Jet Falcon Series F
Compressor Rotor	782101AL-2	Hamilton Sundstrand 782101-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 782101AL <u>Rev:</u> C <u>Date:</u> 04/27/2004, or later FAA approved revisions	Fokker Services B.V.	F27 Mark 050
				Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series
Ball Bearing	M202HX67AL	Hamilton Sundstrand M202HX67	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> M202HX67AL <u>Rev:</u> A <u>Date:</u> 12/03/2004, or later FAA approved revisions	Fokker Services B.V.	F27 Mark 050
				Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series
				Bombardier	CL-600-2B19

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Turbine Rotor	785733AL-2	Hamilton Sundstrand 785733-2	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 785733AL <u>Rev:</u> D <u>Date:</u> 10/20/2005, or later FAA approved revisions	Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series
				Bombardier	CL-600-2B19 (Regional Jet Series 100 & 440)
				Dassault Aviation	Fan Jet Falcon Fan Jet Falcon Series C Fan Jet Falcon Series D Fan Jet Falcon Series E Fan Jet Falcon Series F
				Fokker Services B.V.	F27 Mark 050
Turbine Rotor	785733AL-5	Hamilton Sundstrand 785733-5	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 785733AL <u>Rev:</u> C <u>Date:</u> 10/07/2004, or later FAA approved revisions	Bombardier	DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series

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Bearing, Ball, Annular	3502689-1TA	3502689-1	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3502689-1TA <u>Rev:</u> A <u>Date:</u> 07/02/2009, or later FAA- approved revisions	Boeing	737-300, 400, 500, 600, 700, 800, 900 series
					747SP, 747-100B, 200B, 200C, 200F, 300 series
					DC-10-10, 10F, 15, 30, 30F, 40, 40F series
				Airbus	A319-100 series A320-100, 200 series A321-100, 200 series A340-200, 300 series
				Bombardier	CL600-1A11, 2A12, 2B16, 2B19 (CRJ100/CRJ200), 2C10 (CRJ700), 2D24 (CRJ900) series
				CFM International	CFM56-5A, 5B, 5C series CFM56-7B series
				Rolls-Royce PLC	RB211-524B, 524C, 524D series
Turbine Rotor	785733AL-5	Hamilton Sundstrand 785733-5	Test and Computation per 14 CFR §21.303 <u>Dwg No.:</u> 785733AL <u>Rev:</u> C <u>Date:</u> 10/07/04, or later FAA approved revisions	Bombardier	BR700-710A, 710C series BR700-715A, 715B, 715C series
					DHC-8-100 Series DHC-8-200 Series DHC-8-300 Series

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Parts Kit, Bearing First Stg. Carrier	3029277TA	3029277	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3029277TA <u>Rev:</u> I/R <u>Date:</u> 05/28/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 35, 36, 38, 41, 42, 42A, 61, 112, 114, 114A, 121, 135, 135A
Bearing Option	3030046TA	3030046	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3030046TA <u>Rev:</u> I/R <u>Date:</u> 05/28/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-45A, 45B, 45R, 50, 60A, 65AR, 65B, 65R, 65AG, 67, 67AF, 67R, 67A, 67B, 67D
Parts Kit, First- Stage Planet Gear Bearing	E3038856	3038856	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3038856 <u>Rev:</u> A <u>Date:</u> 04/22/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-67R, 67AF, 67AG, 67T, 67D
Diffuser Assembly, Compressor	6870813TA	6870813	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6870813TA <u>Rev:</u> I/R <u>Date:</u> 01/15/2010, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C (T63-A-720), C20F, C20J, C20S, C20W
Gearshaft, Assembly Helical, Torquemeter	23084248TA	23084248	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23084248TA <u>Rev:</u> I/R <u>Date:</u> 01/13/2010, or later FAA approved revisions	Rolls-Royce	250- C30, C30G, C30G/2, C30M, C30P, C30R (T703- AD-700), C30R/1 (T-703-AD-700B), C30R/3, C30R/3M, C30S, C30U, C47B, C47M

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Gearshaft Assembly Helical-PTO	23084249TA	23084249	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23084249TA <u>Rev:</u> I/R <u>Date:</u> 01/13/2010, or later FAA approved revisions	Rolls-Royce	250- C30, C30G, C30G/2, C30M, C30P, C30R (T703- AD-700), C30R/1 (T-703-AD-700B), C30R/3, C30R/3M, C30S, C30U, C47B, C47M
Gearshaft Assembly- Fuel Control- Oil Pump	6876529TA	6876529	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6876529TA <u>Rev:</u> I/R <u>Date:</u> 01/09/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F1, B17F2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R1, C20R2, C20R4, C20S, C20W
Tube Assembly External Sump Accessory Housing	6853464TA	6853464	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6853464TA <u>Rev:</u> I/R <u>Date:</u> 01/06/2010, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Tube, Oil Delivery	23034102TA	23034102	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23034102TA <u>Rev:</u> I/R <u>Date:</u> 01/05/2010, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Bearing, Ball	23009670TA	23009670	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23009670TA <u>Rev:</u> I/R <u>Date:</u> 08/12/2009, or later FAA approved revisions	Rolls-Royce	250-C30 250-C30P 250-C30M 250-C30S 250-C30G 250-C30G/2

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Gearshaft Assembly Helical PTO	6899402TA	6899402	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6899402TA <u>Rev:</u> I/R <u>Date:</u> 01/27/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20R, C20R/1, C20R/2, C20S, C20W, T63-A-5A, T63-A-700, T63-A-720
Bearing, Roller, 3032615TA Flanged	3032615TA	3032615	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> 3032615TA <u>Rev:</u> I/R <u>Date:</u> 10/15/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 38, 41, 42, 42A, 61, 64, 66, 66B, 110, 112, 114, 114A, 121, 135, 135A
Nozzle, Assembly, Turbine, 4th Stage	23055939TA	23055939	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23055939TA <u>Rev:</u> I/R <u>Date:</u> 02/11/2010, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20F, C20J, C20W
Nozzle, Turbine, 3rd Stage	23069744TA	23069744	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23069744TA <u>Rev:</u> I/R <u>Date:</u> 02/11/2010, or later FAA approved revisions	Rolls-Royce	250-C20, C20B, C20F, C20J, C20W
Gear Set, Planet, First Stg. Reduction	E3101527-01	3101527-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3101527-01 <u>Rev:</u> A <u>Date:</u> 04/29/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-50, 45A, 45B, 45R, 60A, 60AG, 65B, 65R, 65AR, 65AG, 67, 67R, 67A

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Gear, Sun, First Stg. Reduction	E3100743-01	3100743-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3100743-01 <u>Rev:</u> A <u>Date:</u> 04/29/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-50
Gear, Sun, First Stg. Reduction	E3100767-01	3100767-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> E3100767-01 <u>Rev:</u> A <u>Date:</u> 04/29/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-45A, 45B, 45R, 60A, 60AG, 65B, 65R, 65AR, 65AG, 67, 67R, 67A
Bearing, Roller, Flanged	3032647TA	3032647	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3032647TA <u>Rev:</u> I/R <u>Date:</u> 09/21/2009, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-45A, 45B, 45R, 60A, 60AG, 65B, 65R, 65AG, 65AR, 67, 67A, 67B, 67D, 67R, 67T, 67AF, 67AG
Housing, Compressor Turbine Shroud	3114464-01TA	3114464-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3114464-01TA <u>Rev:</u> I/R <u>Date:</u> 03/17/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT-6A, 60A, 61, 65B, 65R, 65AG, 65AR
Bearing, Ball 40 X 68 X 15 mm	23004553TA	23004553	Test and Computations per 14 CFR § 21.303 <u>Dwg No.:</u> 23004553TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-C18, C18A, C18B, C18C, B15G, C17, C17B, C17C, C17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Bearing, Ball 12 X 37 X 12 mm	23005746TA	23005746	Test and Computations per 14 CFR § 21.303 <u>Dwg No.:</u> 23005746TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M

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Bearing, Ball	23007157TA	23007157	Test and Computations per 14 CFR § 21.303 <u>Dwg No.:</u> 23007157TA <u>Rev:</u> I/R <u>Date:</u> 05/18/2010, or later FAA approved revisions	Rolls-Royce	250- C28B, C28C, C30, C30G, C30G/2, C30M, C30P, C30S,
Bearing, Ball- Annular 20mm x 42mm x 15mm	23031478TA	23031478	Test and Computations per 14 CFR § 21.303 <u>Dwg No.:</u> 23031478TA <u>Rev:</u> I/R <u>Date:</u> 05/18/2010, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30L, C30M, C30P, C30R, C30S, C30U, C30G2, C30R/1, C30R/3, C40B, C47B, C47M
Bearing, Ball 40mm ID X 62mm OD X 12mm W	6871505TA	6871505	Test and Computations per 14 CFR § 21.303 <u>Dwg No.:</u> 6871505TA <u>Rev:</u> I/R <u>Date:</u> 06/15/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20W, C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M

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Bearing, Ball 30mm ID X 55mm OD X 13mm W	6889093TA	6889093	Test and Computations per 14 CFR § 21.303 <u>Dwg No.:</u> 6889093TA <u>Rev:</u> I/R <u>Date:</u> 06/15/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17E, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20S, C20W
Bearing, Ball- Annular 10mm X 26mm	6898607TA	6898607	Test and Computations per 14 CFR § 21.303 <u>Dwg No.:</u> 6898607TA <u>Rev:</u> I/R <u>Date:</u> 06/15/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C10, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20S, C20W, C28B, C28C, C30, C30L, C30M, C30P, C30R, C30S, C40B, C47B, C47M
Bearing, Ball 15mm ID X 42mm OD X 13mm	23005747TA	23005747	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23005747TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-C28B,C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M

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Bearing, Ball – Annular 10mm X 26mm X 8mm	23005749TA	23005749	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23005749TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Bearing, Ball – Annular 25 X 42 X 9 mm	23007151TA	23007151	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23007151TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-C10, B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M

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Bearing, Ball, 20mm ID X 42mm OD X 12 mm W	23007152TA	23007152	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23007152TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, 250- C28B,C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U
Bearing, Ball – Annular 17 X 30 X 7 mm	23008065TA	23008065	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23008065TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Bearing, Ball – Annular 12 mm X 28 mm X 8 mm	23008469TA	23008469	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23008469TA <u>Rev:</u> I/R <u>Date:</u> 05/19/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, 250- C28B,C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30S, C30U, C40B, C47B, C47M

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Bearing, Roller, Flanged	7T0101TA	7T0101	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 7T0101TA <u>Rev:</u> I/R <u>Date:</u> 01/11/2010, or later FAA approved revisions	Pratt & Whitney	PW2037, PW2037D, PW2037M, PW2040, PW2040D, PW2043, PW2143, PW2240, PW2337
Bearing, Roller, Flanged	7T0110TA	7T0110	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 7T0110TA <u>Rev:</u> I/R <u>Date:</u> 02/22/2010, or later FAA approved revisions	Pratt & Whitney	PW2037, PW2037D, PW2037M, PW2040, PW2040D, PW2043, PW2143, PW2240, PW2337
Bearing, Ball, Duplex	TAI-2040-143- 103	212-040-143-103	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2040-143-103 <u>Rev:</u> I/R <u>Date:</u> 05/14/2010, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212, 412
Bearing, Ball, Annular, Duplex	3032633TA	3032633	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3032633TA <u>Rev:</u> C <u>Date:</u> 09/29/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B
Bearing, Roller 40mm ID X 68mm OD X 15mm W	23001949TA	23001949	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23001949TA <u>Rev:</u> I/R <u>Date:</u> 06/01/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Bearing, Roller 30mm ID X 62mm OD X 16mm	23004554TA	23004554	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23004554TA <u>Rev:</u> I/R <u>Date:</u> 06/01/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20S, C20W
Bearing, Outer Race and Roller Assembly	23007202TA	23007202	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23007202TA <u>Rev:</u> I/R <u>Date:</u> 06/01/2010, or later FAA approved revisions	Rolls-Royce	250-B15G, B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C18A, C18B, C18C, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W, C28B, C28C, C30, C30G, C30G2, C30L, C30M, C30P, C30R, C30R/1, C30R/3, C30S, C30U, C40B, C47B, C47M
Bearing, Outer Race and Roller Assembly - 32mm O.D. X 7mm W.	23034787TA	23034787	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23034787TA <u>Rev:</u> I/R <u>Date:</u> 07/28/2010, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/4, C20S, C20W
Bearing, Roller Cylinder Assembly - 40mm I.D. X 68mm O.D. X 14mm	23053297TA	23053297	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23053297TA <u>Rev:</u> I/R <u>Date:</u> 07/28/2010, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W

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Bearing, Roller, Front Compressor	23066716TA	23066716	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 23066716TA <u>Rev:</u> I/R <u>Date:</u> 07/28/2010, or later FAA approved revisions	Rolls-Royce	250-C28B, C28C, C30, C30G, C30G2, C30M, C30P, C30S, C40B, C47B, C47M
Bearing, Roller, 25mm X 62mm X 17mm	6875035TA	6875035	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6875035TA <u>Rev:</u> I/R <u>Date:</u> 07/28/2010, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, B17F, B17F/1, B17F/2, C18, C20, C20B, C20C, C20F, C20J, C20R, C20R/1, C20R/2, C20R/4, C20S, C20W
Bearing, Ball	3032623TA	3032623	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3032623TA <u>Rev:</u> I/R <u>Date:</u> 01/05/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 110, 112, 114, 114A, 121, 135, 135A; PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6
Bearing, Ball, Annular	3033481TA	3033481	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3033481TA <u>Rev:</u> A <u>Date:</u> 11/23/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 38, 41, 42, 110, 112, 114, 114A, 121, 135, 135A; PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

EXTEX Engineered Products Inc.
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Gilbert Arizona 85297 USA

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SUPPLEMENT NO.: 1
DATE: May 9, 2016
REISSUED: September 8, 2016

ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Coupling, Male Spherical, Driveshaft	TAI-4040-603- 007	204-040-603-007	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-603 <u>Rev:</u> B <u>Date:</u> 04/26/2010, or later FAA- approved revisions	Bell Helicopter Textron	204B 205A-1
Coupling, Male Spherical, Driveshaft	TAI-4040-603- 009	204-040-603-009	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-603 <u>Rev:</u> B <u>Date:</u> 04/26/2010, or later FAA- approved revisions	Bell Helicopter Textron	204B 205A-1 212 412
Coupling, Male Spherical, Driveshaft	TAI-4040-603- 011	204-040-603-011	Test & Computation per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-603 <u>Rev:</u> B <u>Date:</u> 04/26/2010, or later FAA- approved revisions	Bell Helicopter Textron	205A-1 212 412
Bearing, Ball, Duplex	TAI-6040-438- 101	206-040-438-101	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6040-438-101 <u>Rev:</u> I/R <u>Date:</u> 07/14/2010, or later FAA- approved revisions	Bell Helicopter Textron	206B, 206L, 206L1, 206L3, 206L4
Bearing, Roller, Flanged	3043035TA	3043035	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 3043035TA <u>Rev:</u> I/R <u>Date:</u> 08/02/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B
Bearing, Ball, Annular	3122868-01TA	3122868-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3122868-01TA <u>Rev:</u> I/R <u>Date:</u> 06/28/2010, or later FAA- approved <u>Revisions</u>	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3D, 3DE, 3DF, 6, 6B

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Support Assembly, Compressor – Front	6890530TA	6890530	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 6890530TA <u>Rev:</u> I/R <u>Date:</u> 01/21/2010, or later FAA approved revisions	Rolls-Royce	250-B17, B17B, B17C, B17D, B17E, C20, C20B, C20C, C20F, C20J, C20S, C20W
Bearing	TAI-2040-210- 101	212-040-210-101	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-2040-210-101 <u>Rev:</u> I/R <u>Date:</u> 06/13/2011, or later FAA approved revisions	Bell Helicopter Textron	205A1, 212, 412, 412EP

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Gearshaft, Sector Assembly IGV, Spur	3810120-1TA	3810120-1	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 3810120-1TA <u>Rev:</u> I/R <u>Date:</u> 09/29/2010, or later FAA approved revisions	Airbus	Honeywell GTCP331-250F, GTCP331-250H as installed on: A310 models 204, 221, 222, 203, 322, 324, 304, 325 A300 models B4-600, B4-603, B4-620, B4-622, C4-605R, F4-605R, F4-622R, Honeywell GTCP331-200AC as installed on: Boeing B767-200 Series, Honeywell GTCP331-200ER as installed on: Boeing 757-200 series, 757-200PF series 757-200CB series 757-300 series B767-200 series B767-300 series B767-300F series B767-400ER series
Bearing, Duplex	TAI-4040-424- 105	204-040-424-105	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-4040-424-105 <u>Rev:</u> I/R <u>Date:</u> 05/18/2011, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212, 412, 412EP

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ARTICLE NAME	ARTICLE NUMBER	APPROVED REPLACEMENT FOR ARTICLE NUMBER	APPROVAL BASIS AND APPROVED DESIGN DATA	MAKE/TCH ELIGIBILITY	MODEL/SERIES ELIGIBILITY
Gear, Idler, Intermediate Drive	3015427TA	3015427	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 3015427TA <u>Rev:</u> I/R <u>Date:</u> 08/18/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B
Gear, Spur, Clutch Drive	3122871-01TA	3122871-01	Test and Computations per 14 CFR § 21.303 <u>Dwg No:</u> 3122871-01TA <u>Rev:</u> I/R <u>Date:</u> 12/01/2010, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B
Bearing, Ball, Duplex	TAI-6040-032- 105	206-040-032-105	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-6040-032-105 <u>Rev:</u> I/R <u>Date:</u> 07/30/2010, or later FAA- approved <u>Revisions</u>	Bell Helicopter Textron	206A, 206B, 206L, 206L-1, 206L-3, 206L-4
Bearing, Ball	369D25149TA	369D25149	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 369D25149TA <u>Rev:</u> I/R <u>Date:</u> 04/26/2010, or later FAA approved revisions	MD Helicopters,	369D, 369E, 369FF, 500N, 600N
Bearing, Roller	369D25142TA	369D25142	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 369D25142TA <u>Rev:</u> I/R <u>Date:</u> 06/01/2010 or later FAA approved revisions	MD Helicopters,	369D, 369E, 369FF, 500N, 600N
Bearing	TAI-2040-136- 001	212-040-136-001	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-2040-136-001 <u>Rev:</u> I/R <u>Date:</u> 03/05/2010, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212

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Bushing-Pitch Horn, Main Rotor	TAI-4010-170- 5	204-010-170-005	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-4010-170 <u>Rev:</u> I/R <u>Date:</u> 01/11/2012, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212
Bushing-Pitch Horn, Main Rotor	TAI-4010-170- 7	204-010-170-007	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-4010-170 <u>Rev:</u> I/R <u>Date:</u> 01/11/2012, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212
Bushing-Pitch Horn, Main Rotor	TAI-4010-170- 9	204-010-170-009	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-4010-170 <u>Rev:</u> I/R <u>Date:</u> 01/11/2012, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212
Bushing	TAI-2010-31- 34-24	22-010-31-34-24	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> TAI-2010-31-34-24 <u>Rev:</u> I/R <u>Date:</u> 11/25/2011, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212
Bearing, Input	369D25417TA	369D25417	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 369D25417TA <u>Rev:</u> I/R <u>Date:</u> 07/29/2010, or later FAA approved revisions	MD Helicopters,	369D, 369E, 369FF

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Bearing, Annular Ball	358629-1TA	358629-1	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 358629-1TA <u>Rev:</u> I/R <u>Date:</u> 11/25/2011, or later FAA approved Revisions	Boeing	Honeywell TSCP 700-4B as installed on: DC-10 KC-10 Honeywell TSCP 700-4E as installed on: Boeing MD-11 Honeywell TSCP 700-5 as installed on Airbus A300
Bearing, Roller	369D25147TA	369D25147	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 369D25147TA <u>Rev:</u> I/R <u>Date:</u> 07/29/2010, or later FAA approved revisions	MD Helicopters,	369D, 369E, 369FF, 500N, 600N
Bearing – Roller, Flanged	3032627TA	3032627	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3032627TA <u>Rev:</u> IR <u>Date:</u> 10/19/2010, or later FAA- approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6
Bearing, Option (IC) (No. 15 Bearing)	3032640TA	3032640	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 3032640TA <u>Rev:</u> I/R <u>Date:</u> 12/23/2011, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 9
Bearing, Ball	3121518-01TA	3121518-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3121518-01TA <u>Rev:</u> IR <u>Date:</u> 12/30/2011, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3, 3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6

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Bearing, Ball	369D25140TA	369D25140	Test and Computations per 14 CFR §21.303 <u>Dwg No:</u> 369D25140TA <u>Rev:</u> I/R <u>Date:</u> 06/02/2010, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF, 500N
Bearing, Roller	369D25143TA	369D25143	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 369D25143TA <u>Rev:</u> I/R <u>Date:</u> 06/09/2010, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF, 500N
Bearing, Ball	369D25144TA	369D25144	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 369D25144TA <u>Rev:</u> I/R <u>Date:</u> 06/21/2010, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF, 500N
Bushing	TAI-4010-413- 15	204-010-413-015	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4010-413-15 <u>Rev:</u> I/R <u>Date:</u> 11/30/2005, or later FAA approved revisions	Bell Helicopter Textron	204B, 205A-1, 212
Bearing, Ball	TAI-4040-623- 111	204-040-623-111	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> TAI-4040-623-111 <u>Rev:</u> A <u>Date:</u> 03/26/2012, or later FAA approved revisions	Bell Helicopter Textron	204B 205A-1
Gear, Sun, First Stage Reduction	3024765TA	3024765	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3024765TA <u>Rev:</u> I/R <u>Date:</u> 08/17/2012, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-15AG, 25C, 27, 28, 34, 34AG, 34B, 35, 36

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Gear Set, Sun, First Stage Reduction	3028456TA	3028456	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3028456TA <u>Rev:</u> I/R <u>Date:</u> 08/17/2012, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41,42 42A, 61, 64, 66, 66B
Gear, Sun, First Stage Reduction	3037304TA	3037304	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3037304TA <u>Rev:</u> I/R <u>Date:</u> 08/17/2012, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-110, 112, 121, 114, 114A, 135, 135A
Gear Set, Planet, First Stage Reduction	3101455-02TA	3101455-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3101455-02TA <u>Rev:</u> I/R <u>Date:</u> 08/17/2012, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-15AG, 110, 112, 121, 25C, 27, 28, 34, 34AG, 34B, 35, 36, 61, 64, 66, 66B, 114, 114A, 135, 135A
Gear Set, Planet, First Stage Reduction	3101525-02TA	3101525-02	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3101525-02TA <u>Rev:</u> I/R <u>Date:</u> 08/17/2012, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-38, 41, 42, 42A
Bearing, Input Shaft	369D25418TA	369D25418	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 369D25418TA <u>Rev:</u> I/R <u>Date:</u> 08/16/2010, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF
Bearing, Output Pinion	369D25419TA	369D25419	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 369D25419TA <u>Rev:</u> I/R <u>Date:</u> 08/20/2010, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF
Bearing	369D25420TA	369D25420	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 369D25420TA <u>Rev:</u> I/R <u>Date:</u> 08/23/2010, or later FAA approved revisions	MD Helicopters	369D, 369E, 369FF

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Bearing, Roller, Flanged	3032642TA	3032642	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3032642TA <u>Rev:</u> IR <u>Date:</u> 01/24/2013, or later FAA approved revisions	Pratt & Whitney Canada	PT6T-3,3B, 3BE, 3BF, 3BG, 3D, 3DE, 3DF, 6, 6B
Bearing, Ball	TAI-2040-456- 103	212-040-456-103	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> TAI-2040-456-103 <u>Rev:</u> I/R <u>Date:</u> 03/08/2013, or later FAA approved revisions	Bell Helicopter Textron	205A-1 205B 212 412 412EP
Bearing	TAI-6040-036- 101	206-040-036-101	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> TAI-6040-036-101 <u>Rev:</u> A <u>Date:</u> 09/16/2013, or later FAA approved revisions	Bell Helicopter Textron	206A, 206B
Ring, Blade Retaining	3010607TA	3010607	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3010607TA <u>Rev:</u> A <u>Date:</u> 02/03/2015, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-6, 6A, 6B, 11, 11AG, 15AG, 20, 20A, 20B, 21, 25, 25A, 25C, 27, 28, 34, 34AG, 34B, 36, 110, 112, 114, 114A, 121, 135, 135A
Fitting Grip, Drag Brace Assembly	ASI-4011-180- 003	204-011-180-003	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-180 <u>Rev:</u> C <u>Date:</u> 10/8/2014, or later FAA approved revisions	Bell Helicopter Textron	212

Installation of the PMA replacement
article on eligible products requires
approval of an alternative method of
compliance (AMOC) to Airworthiness
Directive (AD) 90-26-11.

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Clevis, Drag Brace Assembly	TAI-4011-179- 003	204-011-179-003	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> ASI-4011-179 <u>Rev:</u> IR <u>Date:</u> 04/11/2014, or later FAA approved revisions Installation of the PMA replacement article on eligible products requires approval of an alternative method of compliance (AMOC) to Airworthiness Directive (AD) 90-26-11.	Bell Helicopter Textron	212
Ring, Sealing, Power Output Section	3120085-01TA	3120085-01	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 3120085-01TA <u>Rev:</u> A <u>Date:</u> 10/06/2014, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61
Bearing, Ball 45mm X 85mm X 19mm	6889326AL	6889326	Test and Computations per 14 CFR §21.303 <u>Dwg No.:</u> 6889326AL <u>Rev:</u> I/R <u>Date:</u> 07/03/2002, or later FAA approved revisions	Pratt & Whitney Canada	PT6A-60A, 61


-----End of Data-----

GENERAL NOTES:

- 1) Provide minor design changes in a manner as determined by the ACO. Process major design changes to drawings and specifications in the same manner as that for an original FAA-PMA.
- 2) The FAA accepted ICA for the above articles with their designs. These ICA may refer to those of the respective articles from the holders of type certificates. Otherwise, provide supplemental ICA for differences in the replacement articles. Make referral statements, or supplemental ICA readily available per 14 CFR 21.50.

ORIGINAL SIGNATURE ON FILE

Ronald Atmur
Manager, Propulsion Branch
Los Angeles Aircraft Certification Office


John Rigg
Manager, Phoenix
Manufacturing Inspection District Office