



U.S. Department
of Transportation
**Federal Aviation
Administration**

FEDERAL AVIATION ADMINISTRATION – PARTS MANUFACTURER APPROVAL

EXTEX Engineered Products, Inc.
340 E Germann, Suite 113
Gilbert, Arizona 85297-0614

PMA No. PQ2203NM
Supplement No. 91
Date 9/24/2025

Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Nozzle, Turbine, Turbine Engine First Stage, Machining (Vane Holes) Overbridge	1-110-710-06	1-110-710-06	Identity per 14 CFR 21.303, licensing agreement between EXTEX Engineered Products Inc. and Ozark Aeroworks, LLC. Letter of Assist dated 09/11/2025 <u>DWG No:</u> 1-110-710-06 <u>Rev:</u> AC <u>Date:</u> 08/28/2025 or later FAA-approved revisions	Ozark Aeroworks, LLC	T5313B, T5317A, T5317A-1, T5317B, T5317BCV
Flange, Nozzle, First Turbine, Machined	1-110-255-08	1-110-255-08	Identity per 14 CFR 21.303, licensing agreement between EXTEX Engineered Products Inc. and Ozark Aeroworks, LLC. Letter of Assist dated 09/11/2025 <u>DWG No:</u> 1-110-255-08 <u>Rev:</u> D <u>Date:</u> 09/11/2025 or later FAA-approved revisions	Ozark Aeroworks, LLC	T5313B, T5317A, T5317A-1, T5317B, T5317BCV

Article Name	Article Number	Approved Replacement for Article Number	Approval Basis and Approved Design Data	Make/TCH Eligibility	Model/Series Eligibility
Liner, Combustion Chamber, Forward, Skirted, Reduced Outer Channel	1-110-162-06	1-110-162-06	<p>Identity per 14 CFR 21.303, licensing agreement between EXTEX Engineered Products Inc. and Ozark Aeroworks, LLC.</p> <p>Letter of Assist dated 09/11/2025</p> <p><u>DWG No:</u> 1-110-162-06</p> <p><u>Rev:</u> E</p> <p><u>Date:</u> 01/29/2025 or later FAA-approved revisions</p>	Ozark Aeroworks, LLC	T5313B, T5317A, T5317A-1, T5317B, T5317BCV

-----End of Listing-----

Note: The procedures that have been accepted by the type certificate or TSO authorization holder and its cognizant FAA certification branch, for minor changes to original articles used on type-certificated products, are also acceptable for incorporating the same minor changes on identical PMA replacement articles. The PMA holder must be able to show traceability relating to the TC, STC, or TSO authorization holder on all minor changes incorporated by this procedure. When these procedures are no longer applicable because of completion of the production contract, or termination of the licensing agreement or business relationship, all subsequent minor design changes to the PMA articles must be submitted in a manner as determined by the certification branch. Major design changes (reference 14 CFR 21.319 and 21.619) to drawings and specifications are to be handled in the same manner as that for an original PMA.

**GARY G
NICHOLAS**

Digitally signed by GARY G
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Date: 2025.09.24 10:18:32
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for John P. Rigg
Manager, Scottsdale Certificate Management Section
West Certificate Management Branch
System Oversight Division
Aircraft Certification Service